

Wind tunnel tests on plate girder bridges

Prepared for Highways Agency

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Background

Simple criteria for assessing the susceptibility of bridges to aerodynamic effects first appeared in general form in a paper entitled 'Proposed British Design Rules' (Institution of Civil Engineers, 1981). The rules presented equations for determining critical wind speeds at which vortex shedding or aerodynamic instability (galloping, non-oscillatory divergence) would occur. The critical wind speeds could then be compared with design wind speeds to determine whether they were likely to occur. The rules also presented simplified formulae for the calculation of amplitudes of vibration and loading for vortex excitation and gave guidance on response to turbulence and fatigue assessment.

These simplified rules were used primarily to identify bridges that may be prone to aerodynamic effects without carrying out sophisticated dynamic analysis. Should aerodynamic problems be indicated, then the designer has the option of re-designing or testing in a wind tunnel.

The rules applied to a limited range of bridge size and types, based on the knowledge that existed at that time resulting from wind tunnel tests and practical experience. Further studies were subsequently carried out, including a calibration exercise to derive appropriate partial safety factors for vortex shedding loads, to make the rules applicable to a wider range of structures. The rules were modified accordingly and made available in TRL Contractor Report CR36 (Flint and Neill Partnership 1986), which was widely used in the assessment of bridge stability. The rules were developed further in TRL Contractor Report CR256 (Flint and Neill Partnership 1991) and were eventually published in Departmental Standard BD 49/93.

Scope of the project

Because of the lack of wind tunnel test results, certain bridge types were beyond the scope of BD 49/93. One important category of bridges was omitted, ie, plate girder bridges, for which much of the standard was not applicable. It was also apparent from the many queries posed by bridge engineers using BD 49/93 that the geometric constraints imposed by the rules, particularly those relating to leading edge details, eliminated many typical structures from being covered by the standard. To widen the range of bridge types and configuration covered by the standard, a comprehensive wind tunnel testing programme was carried out. This report summarises the results and conclusions derived from these tests.

The wind tunnel tests were carried out by Flint and Neill Partnership in collaboration with BMT Fluid Mechanics Limited. As well as examining the critical wind speeds, vibration amplitudes and stability of plate girder bridges, other topics were investigated. These included the effect on aerodynamic behaviour of varying edge beam dimensions, parapet height and solidity. The effect of presence of solid parapet over short lengths of the span, as is required for bridges over railway-lines, was also investigated.

Summary

Over 60 wind tunnel tests were carried out on section models of bridge decks of various configuration and edge detail. The tests covered a range of wind speeds and a variety of damping values. The models were mounted on elastic supports that enabled them to move in their fundamental bending and torsional modes of vibration. The tests were mainly carried out in smooth flow since this gives conservative estimates of vortex shedding response and critical wind speeds. Some tests in turbulent flow were included in the programme for reference.

The effect of different edge detail on the aerodynamic response was investigated using parapets of varying height and solidity that could be attached to the different models. Two basic fascia beam configurations were tested, as well as various fascia beam depth and overhang values. Other edge details, such as 'thickening' of the slab were also included.

The models and test configuration are described fully in this report. The report contains the data obtained from the wind tunnel tests and the main conclusions relevant to the 1993 version of BD 49. The tests were subsequently used to produce a revised version of BD 49 (published in 2001). The background to these revisions is presented in a separate report (TRL528).

The data obtained from the wind tunnel tests presented in this report can be used in conjunction with BD 49. This will enable designers to use the data directly rather than in the form of simplified rules, which are necessarily conservative. The data for critical wind speeds, amplitudes of vibration, for both bending and torsion, should only be used where deck configuration and edge details are similar. This may avoid the need for re-design or expensive wind tunnel tests.

Conclusion

The wind tunnel testing programme was used to provide simple rules to determine the aerodynamic behaviour of plate girder bridges of various geometric configurations. These rules are applicable to typical situations and can be used to ascertain the stability of the structure or to identify the situations were more sophisticated studies are required, eg, full dynamic analysis or wind tunnel tests. Where appropriate, the wind tunnel data can be used directly for the assessment of aerodynamic behaviour.

There are still many areas which have not been fully investigated and it is important to note that the need for full dynamic analysis or wind tunnel tests has not been eliminated and is still required in certain situations.

1 Introduction

Flint and Neill Partnership (FNP) in association with BMT Fluid Mechanics Limited (BMT) were awarded a commission by the Transport Research Laboratory (now TRL Limited) to undertake studies and wind tunnel tests on plate girder bridges. The purpose of the investigation was to confirm or extend the content of certain clauses in the simplified design rules for bridge aerodynamics (hereinafter referred to as the Rules), as published in BD 49/93 (Highways Agency, 1993).

During the study it was found that the results were particularly sensitive to edge beam depth and overhang. To address these problems a small programme of further tests was undertaken.

This report sets out the results of the study, proposed modifications to the Rules and general conclusions and recommendations regarding the Rules. As a consequence of this study, the equations used to evaluate critical wind speeds and amplitudes of response were revised. Rules were revised and published as BD 49/01 (Highways Agency 2001). Background to other revisions of BD 49 is presented by Smith *et al.* (2002)

2 Background

All the wind tunnel tests, upon which the aerodynamic Rules have been based, were undertaken by BMT (formerly NMI) and a significant part of the codification of the test results was carried out by Flint and Neill Partnership. The original rules published by the Institution of Civil Engineers (1981) were extended as a result of a calibration study undertaken by Flint and Neill Partnership (1986). A further investigation (Flint and Neill Partnership 1991) modified certain clauses in the Rules and these were subsequently published by the Highways Agency as BD 49/93.

It was recognised however that certain aspects of the Rules required further investigation, particularly those associated with plate girder bridges. Accordingly these aspects were identified and formed the basis of the present investigation.

3 Wind tunnel test procedure

3.1 General

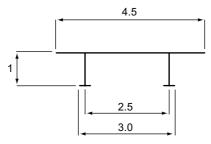
The purpose of the wind tunnel tests was to investigate the aerodynamic behaviour of plate girder bridges in a systematic way so that this category of bridge could be included within the scope of the Rules. The methodology used was to develop a basic matrix of tests covering the parameters to be investigated. The results of the basic matrix were examined as the results became available to assess the sensitivity of changing certain of these parameters. Where it became apparent that small changes had a significant effect on the results the matrix was modified to cover intermediate changes, or to exclude such parameters from the scope of the Rules. The modular form of the models allowed this flexibility. The range of tests

Table 1 Matrix of tests, as undertaken

Fascia		P	arap	oet .	P1	Pa	rape	et P	2	Pa	rape	et P	3	P	araj	pet .	P4
	Angle	а	b	с	d	а	b	с	d	а	b	с	d	а	b	с	G
<i>Model</i> . X	M1 -5 0 +5	~	1		~ ~ ~	k	v		~	ŀ			v				r
Y	-5 0 +5	~	~		222		7		~				~				~
Model . X	M2 -5 0 +5																
Y	-5 0 +5																
Model . X	M3 -5 0 +5					~	✔;	×	~	~	~						
Y	-5 0 +5					~	~		~	~	~		~				~
Model . X	M4 -5 0 +5																
Y	-5 0 +5																
Model . X	M5 -5 0 +5						~			~	~						
Y	-5 0 +5						~			~	~						
Model . X	M6 -5 0 +5																
Y	-5 0 +5																

* Tests marked thus undertaken in smooth and turbulent flow. All other tests in smooth flow only. Model M1 with parapet P1/a was used to assess the effect of solid barriers over short lengths.

within the matrix of configurations is shown in Table 1 and the definitions of the various parameters referred to therein are shown in Figures 1, 2 and 3. The additional tests referred to in Section 1 are shown in Table 2.

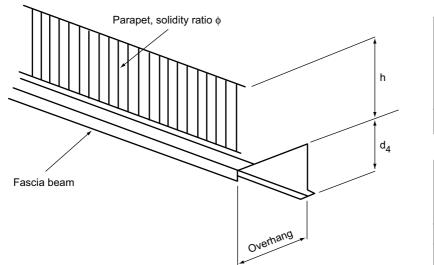


Basic model unit dimensions are relative only

Model number	Configuration	Overhang L/ _{d4}	Number of basic types	b/d4	^{b*} /d4
M1		1.0	1	4.5	3
M2		1.5	1	5.5	3
М3		1.0	2	9	7.5
M4		1.5	2	10	7.5
M5		1.0	3	13.5	12
M6		1.5	3	14.5	12

 $\frac{1}{2}$ Joint in model deck formed by screwing through to stiff backing plate in shadow zone

Figure 1 Model configurations



Parapet type	^h /d ₄
P1	0.3
P2	0.7
P3	1.1
P4	1.5

Suffix	¢
а	0.25
b	0.50
С	0.75
d	1.00

Figure 2 Edge details

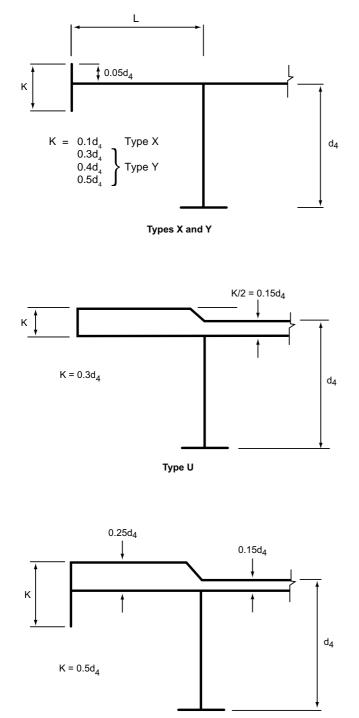


Figure 3 Details of leading edge

Type U¹

Table 2 Matrix of additional tests on model M2 and M3
(All tested at 0° inclination)

			Edg	e detail (k/d_4) or t	ypes	
Over- hang L/d ₄	Parapet type	0.1(X)	0.3(Y)	0.4 (Y)	0.5 (Y)	U	U^{I}
Model M3 1.0	1a 1b						
	2a 2b		v	v	~	~	~
	3a 3b						
0.5	1a 1b						
	2a 2b	<i>v</i> <i>v</i>	r		~ ~	~	v
	3a 3b						
0.25	1a 1b						
	2a 2b		r		~ ~		v
	3a 3b						
Model M1 / 0.5	M2 1a 1b						
	2a 2b			/ †	~		v
	3a 3b				~		v
0.25	1a 1b						
	2a 2b				~		V
	3a 3b			v		~	

All tests undertaken in smooth flow † Test with $L/d_4 = 1.5$

3.2 The models

The programme of wind tunnel tests was performed using a series of section models of representative plate girder bridge decks. The models were manufactured by BMT's workshop facilities and were constructed, on the whole, from carbon fibre. Two basic units were designed and these were assembled to give the family of models shown in Figure 1. This resulted in a series of deck configurations with constant depth, two overhang details and variable overall width.

Previous tests have shown that aerodynamic behaviour of bridge decks is very dependent on the leading edge detail of the structure. For this reason, parapets of various height and solidity were provided and made interchangeable between the various model decks. These are shown in Figure 2. The solidity ϕ is defined as the ratio of solid area to overall area of the parapet.

In addition to parapet details, two basic fascia beam configurations were tested, named X and Y, as shown in Figure 3. The additional tests to investigate fascia beam depth and overhang concentrated on model M3 with overhangs varying from 0.25 to 1.0 times the depth of the beams. In addition, two edge details, types U and U¹, representing thickening of the slab at the edge, were introduced. These details are commonly found in existing bridges, but were beyond the scope of the aerodynamic Rules due to the unavailability of wind tunnel test data. TRL was also concerned with a specific bridge whose configuration was similar to model M2, when mounted with one of the new edge details. Accordingly this was included in the matrix of tests.

The scale of the models was 1:30 based on 1 unit = 1.5m full scale or 1:40 for 1 unit = 2m full scale. A large scale model is essential in providing reliable results, not only because of the need for sufficiently high Reynold's numbers, but also to enable the accurate representation of

small features such as the fascia beam detail at model scale. This is particularly important since slight changes in fascia beam, for example, can have a profound effect on the wind-induced response of a deck section.

However it was still found necessary to use light crossbracing linking the bottom flanges of the beams, to prevent the sections from distorting.

3.3 The wind tunnels

The tests were undertaken in two of BMT's tunnels:

- 1 BMT's No 5 wind tunnel, which is regularly used for section model testing of bridge decks and has a test section of 2.74m wide, 2.14m high and 3.5m long and a top wind speed 65m/sec,
- 2 No 7 environmental wind tunnel, with a test section 4.8m wide, 2.4m high and 15m long, and a top wind speed of 45m/s. Due to the greater width of this wind tunnel, false walls were installed in the test section. These walls spanned the entire height of the test section and were located 2.74m apart. The front end of each wall had a streamlined nose to avoid flow separation at the leading edge. The walls were of suitable length to ensure two-dimensional flow conditions for a sufficient distance downstream of the model. The uniformity of the mean wind speed between the walls was generally better than $\pm 1\%$.

For both wind tunnels the dynamic test rig was mounted on the outside surfaces of the walls. All the components of the test rig were shielded from the flow to avoid any unwanted flow-induced excitation. The set up of the false walls and dynamic rig in wind tunnel No 7 is shown in Figure 4.

Cross-calibration of the two tunnels was undertaken to ensure consistency of the results, by repeat testing some of the configurations and good agreement was obtained.

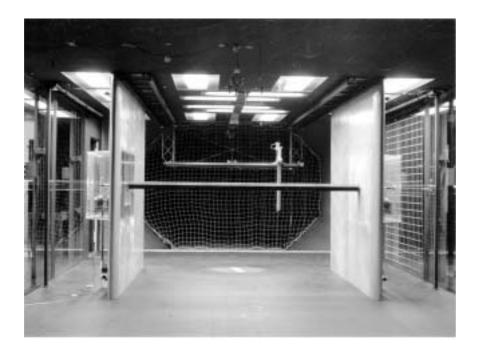


Figure 4 Test set-up in wind tunnel

3.4 Dynamic response measurements

The section models were elastically mounted in the wind tunnel on a dynamic rig incorporating a system of springs to provide two degrees of freedom, ie, vertical bending and torsion. The stiffness of the springs was carefully chosen to provide model scale natural frequencies that result in a convenient wind speed scaling in the wind tunnel. The frequency ratio was not set to a specific value since it was believed that coupled instabilities were unlikely, based on previous experience with plate girder bridge sections. The target values for the mass and inertia were based on full scale data for 22 existing steel plate girder bridges of various widths and spans.

Incorporated in the dynamic rig was an eddy current damping system, which allows the structural damping in both degrees of freedom to be controlled during the test. The dynamic rig also allowed the vertical angle of incidence of the wind relative to the deck to be varied.

In all cases, the test procedure generally consisted of first running through the required wind speed range to provide a general indication of the response. This was followed by measurements of the response amplitudes in bending and torsion at a sufficient number of wind speeds to accurately define any responses due to vortex shedding and the critical wind speeds for the onset of any divergent aeroelastic instabilities such as galloping or flutter. The response was measured using accelerometers mounted on the dynamic rig.

Tests were conducted according to the test matrix defined in Tables 1 and 2. They were generally performed at a wind incidence of 0° , although some measurements were made at $\pm 5^{\circ}$. Tests were performed at up to six levels of structural damping to cover the anticipated range of full scale values.

On the whole, measurements were made in low turbulence flow (nominally smooth flow). Tests in smooth flow are generally recognised to give conservative estimates of the vortex shedding responses and critical wind speeds for divergent instabilities. However, some tests were performed in turbulent flow as defined in the test options to demonstrate the effects of turbulence. Turbulent flow was generated in the wind tunnel using a grid of bars at the entrance to the working section of the wind tunnel.

For the smooth flow tests the longitudinal turbulence intensity (defines as the root mean square of the fluctuating component of the wind divided by the mean wind speed) was less than 1% whereas for the turbulent flow tests the longitudinal turbulence intensity was 7.6%. The longitudinal length scale was of the order of 0.30m.

4 Wind tunnel test results

The results from the wind tunnel tests are presented in full in Appendix A. For each model, with its various fascia beam configurations, the results are tabulated for vortex shedding in bending, vortex shedding in torsion and divergent instability.

For vortex shedding the results are given for various damping values. These generally started at a logarithmic decrement (δ) of 0.01 depending on the set up (eg, minimum δ in torsion for model M3 was 0.0138) and increasing in 0.01 increments until no further response was measured or a maximum logarithmic decrement of 0.06 was

reached. Both the non-dimensional critical wind speed and the maximum root mean square amplitude are tabulated.

For divergent instability the non-dimensional wind speed at the onset of instability is tabulated.

5 Assessment of results

The matrix of tests was limited to three of the six possible models (M1, M3 and M5) to enable general trends in aspect ratio of plate girder bridges to be determined.

The matrix also concentrated initially on solid parapets, which unless they are very low in height were excluded from the simplified formulae given in BD 49/93. It was hoped that the scope for plate girder bridges could be widened, as a result of the tests, to enable such configurations to be assessed using the formulae in the Rules.

The tests pertaining to the effect of changes in the leading edge detail focused on the requirement in the Rules for factoring the predicted amplitudes by 3.0 (Clause 3.1.2 of BD 49/93) for 'bluff' configurations. It was felt that this was simplistic and over-conservative and that a factor dependent on edge detail parameters would be more realistic. These tests were used to develop such an approach.

5.1 Vortex excitation

The general trend of the results for model M1 mounted with solid barriers (ie, parapets with suffix d denoting a solidity of 100%) is shown in Figure 5. In these plots the non-dimensional amplitude for bending is plotted against

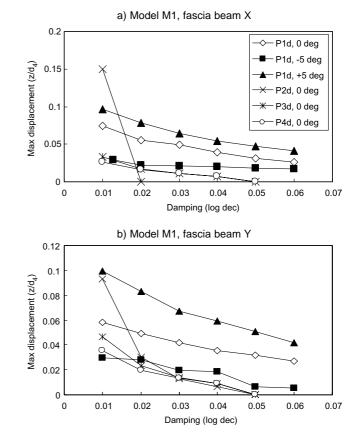


Figure 5 Vortex shedding response in bending with solid parapets

the logarithmic decrement of damping. Figure 5a is for fascia beam X and Figure 5b is for fascia beam Y. As expected the amplitudes decrease with increasing damping. The initially surprising result was that the amplitudes decreased with increasing height of barrier, apart from the single results for P2d at low values of damping. It should be recognised that these results are for the model mounted with solid parapets, a configuration that is outside the scope of the BD 49/93 Rules. The results can probably be explained by the mechanism of vortex shedding which may create strong vortices due to the deep bluff leading face, but a relatively smaller forcing mechanism for higher height to in-wind depth ratios due to a reduction in the effective after-body length.

This trend for reduced amplitudes for higher barriers was reversed for the torsional response, which is plotted in Figure 6a for fascia beam X and in Figure 6b for fascia beam Y.

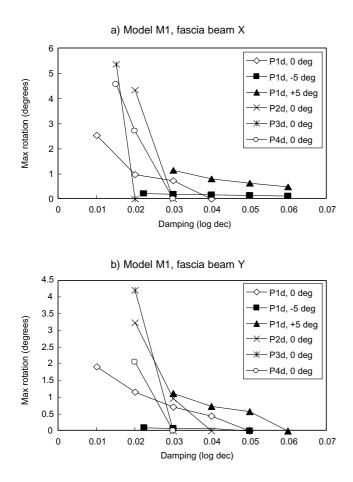
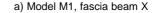


Figure 6 Vortex shedding response in torsion with solid parapets

As may be seen from Figures 5 and 6, amplitudes were higher for positive angles of inclination and lower for negative angles, for an identical configuration (model P1d) for both bending and torsion. Wind inclination can be caused either by topographic effects or by super-elevation of the structure. The design rules have allowed for the effects of up to $\pm 5^{\circ}$ inclination. Super-elevation beyond that angle would be extremely unlikely to occur over a long length of the bridge, particularly for the main span which is of most concern. Wind inclination could occur for sites where an elevated bridge runs normal to the slope of a hill. Such cases are unlikely to occur in practice but bridge designers should be aware of such possibilities.

Results for bending, again for model M1, for open parapets (a and b with solidity 25% and 50% respectively) are given in Figures 7a for fascia beam X and 7b for fascia beam Y which show increased response for increased solidity. Equivalent results for torsion are given in Figures 8a and 8b. It can be seen from Figures 7 and 8 that increasing the height of the 50% porous parapet decreased both the bending and torsional response.



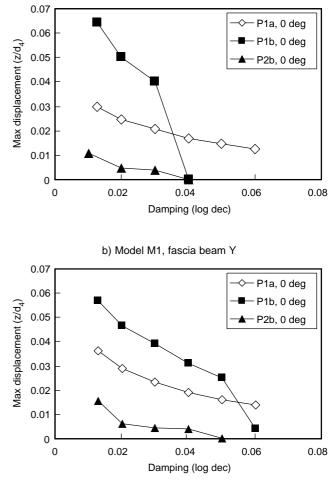


Figure 7 Vortex shedding response in bending with open parapets

Bending results for model M3 with open parapets are summarised in Figure 9a (for fascia beam X) and Figure 9b (for fascia beam Y). For this model it can be seen that:

- a the response increases with increasing height of parapet, although negligibly for the lower solidity parapets;
- b the response increases with increasing solidity of parapet.

Both these trends are as expected, the width of the deck now probably providing an increased forcing mechanism.

Figure 10 shows the torsional results for model M3 with fascia beam Y where the response is higher for the lowest height parapet. However it can be seen that at practical

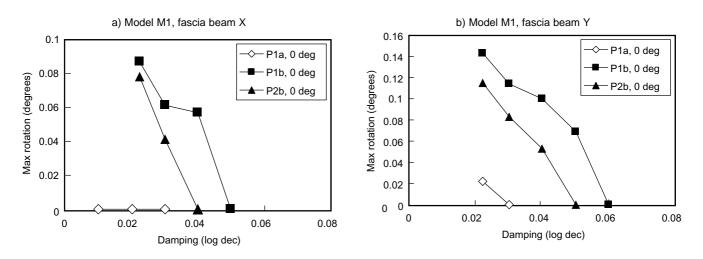


Figure 8 Vortex shedding response in torsion with open parapets

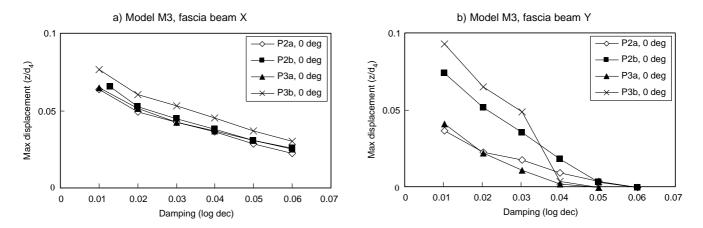


Figure 9 Vortex shedding response in bending with solid parapets

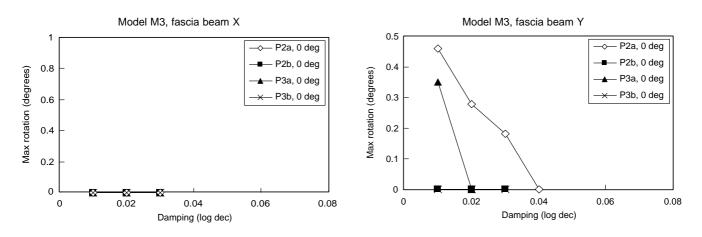


Figure 10 Vortex shedding response in torsion with solid parapets

levels of damping the response is negligible. No torsional response was measured for model M3 with fascia beam X.

For model M3 with solid parapets, large amplitudes were recorded due to vortex shedding in both bending and torsion. For some configurations, the effects of vortex shedding in torsion merged with those of the divergent torsional instability, making the definition of the vortex shedding response almost impossible. Following these initial measurements with model M3, no further configurations were tested with solid parapets. For these reasons the results have not been plotted; the reader is referred to the data given in Appendix A for further details.

Results for the tests with the modified edge details followed the previous pattern in which the amplitudes decrease with increasing damping for both bending and torsion. This may be seen in Figures 11 and 12 for model M3. The amplitudes of vibration for these addition tests, for both bending and torsion, tended to show an increase with the solidity and height of the parapet and with increasing depth of the edge beam. One 'unusual' result was for model M3 with parapet 2b, an overhang of 0.5d and $k/d_4 = 0.1$. For this particular case high amplitudes were obtained from the test results, indeed as high as for the similar configuration with $k/d_4 = 0.5$.

Results for model M5, shown in Figures 13 and 14,

show no consistent trends. Figures 13a and 13b show the results for bending from which it can be seen that for fascia beam X, parapet P2b provides more severe results than P3a whereas for fascia beam Y this trend is reversed. For fascia beam X the lower parapet P2b provides more severe results than parapet P3b. In torsion the logical trends set out in a) and b) above are followed for fascia beam X but for fascia beam Y the parapet with the highest solidity (P3b) produces less onerous results than parapet P3a (see Figure 14.)

From these limited comparisons it can be seen that general and consistent trends could not always be developed, making codification extremely difficult.

Model M1 with open parapets was tested with a solid infill over the middle third of the model's length. Comparisons of critical wind speed and bending amplitudes with the same model without the infill are shown in Figures 15a and 15b. It can be seen that the critical wind speed is virtually unaffected but that the amplitude is increased by up to 30% at very low levels of damping. At realistic damping levels, however, the increase is less than 10%. The model without the infill was stable in torsion, whereas with the solid infill very small torsional amplitudes were recorded (see Appendix B Torsion M1X), some 10% of the current Rule value.

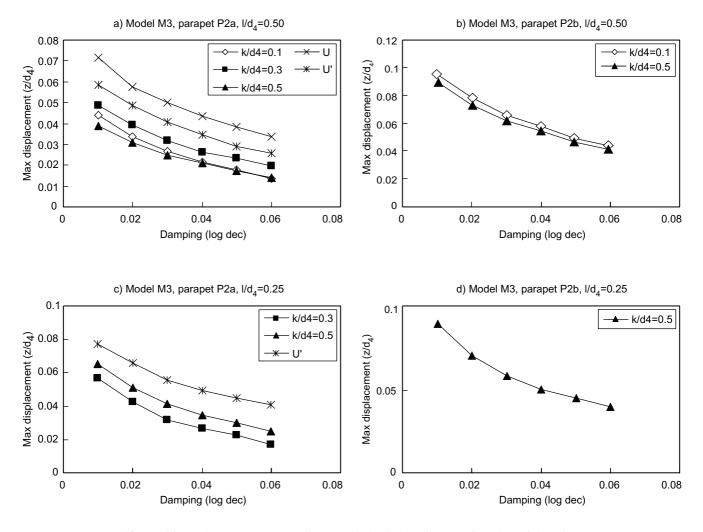


Figure 11 Maximum vortex shedding amplitude in bending as a function of damping

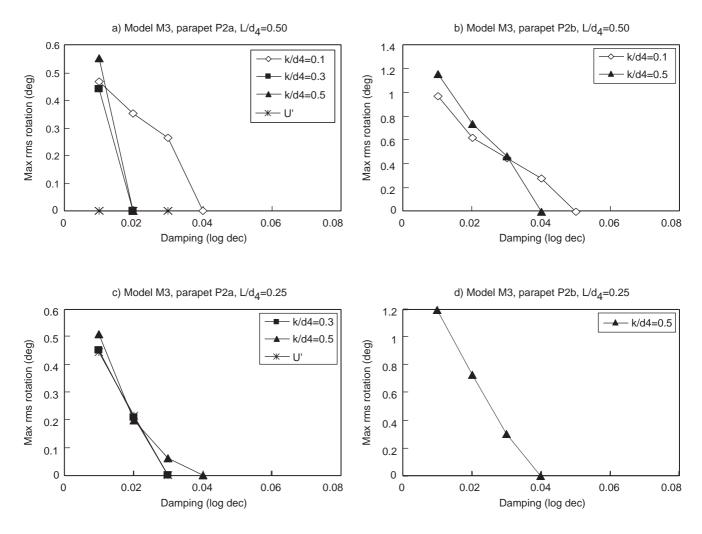


Figure 12 Maximum rotation due to vortex shedding in torsion as a function of damping

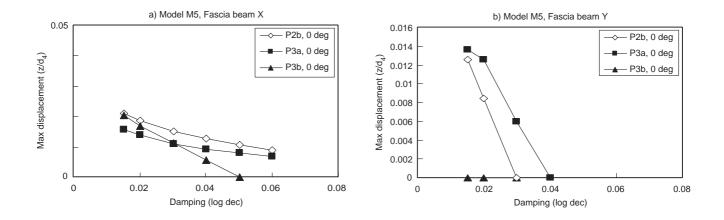


Figure 13 Maximum vortex shedding amplitude in bending with open parapets

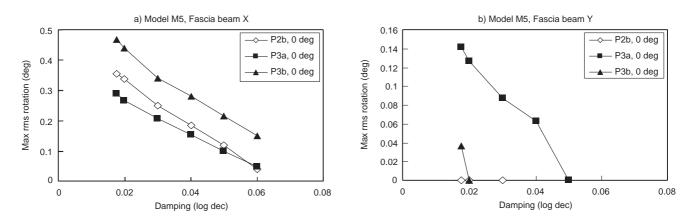


Figure 14 Maximum rotation due to vortex shedding with open parapets

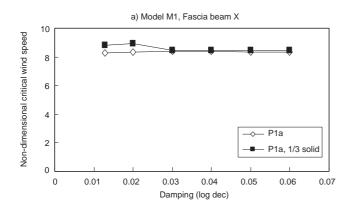


Figure 15a Critical wind speeds for vortex shedding response in bending. Comparison between open parapets and solid infill over 30%

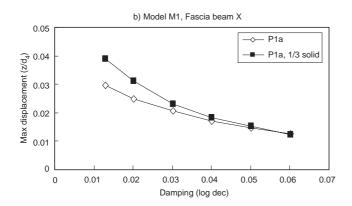


Figure 15b Vortex shedding response in bending with solid parapets. Comparison between open parapets and solid infill over 30%

The effect of turbulent flow was examined on model M1 with fascia beam X and solid parapets P1d and P2d and model M3 with fascia beam X and porous parapet P2b. Results are shown in Figures 16a and 16b for bending and torsion respectively. The bending results show that for parapet P1d the turbulent flow results provide lower amplitudes, as expected, and of the order built into the design rules which allow for turbulent effects. The bending results for P2d only show a single value of response at very low damping, which is suppressed when δ exceeds 0.02.

Whilst the turbulent result figure is marginally higher than the smooth flow result it is not believed that this is a result which can be used for codification purposes. In torsion (see Figure 14b) the turbulence suppresses the response for models with both barriers P1 and P2. Model M3 with fascia beam X and parapet P2b was found to be stable with respect to vortex shedding in both bending and torsion in turbulent flow. In smooth flow, this configuration exhibited a vortex shedding response in bending, but was stable in torsion.

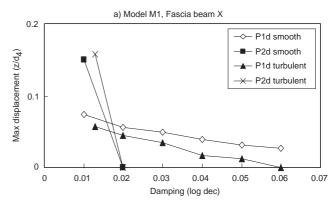
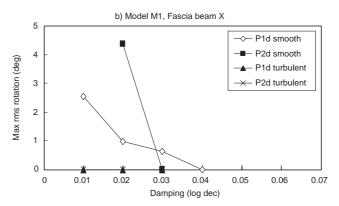
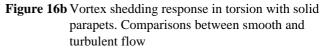


Figure 16a Vortex shedding response in bending with solid parapets. Comparison between smooth and turbulent flow





Model M3 exhibited bending response due to a subharmonic vortex shedding mode at low damping values. However the resulting amplitudes were extremely small some 10% of the main response. The cause of this phenomenon is believed to be associated with vortex formation due to reattachment of the separating shear layers from the leading edge of the deck. Other researchers in this field (Naudascher and Wang 1993) have observed similar responses with, on occasions, several sub-harmonic modes exciting the same natural frequency.

Model M5 similarly exhibited torsion response at a subharmonic with amplitudes less than half of the main response, at a non-dimensional wind speed of about 7.5 to 8.5. The particular configurations displaying this were:

- a fascia beam X with medium parapets (solidity 50%) and medium height (0.7 d₄) (configuration XP2b);
- b fascia beam Y with open parapets (solidity 25%) and height $(1.1 d_{A})$ (configuration YP3a).

Again no common aspect is apparent between these two configurations.

From a general review of these results it was found that the amplitudes, surprisingly in the light of earlier studies, were sensibly independent of the length of overhang. An effective depth could thus be considered taken as:

 $d_4 + k + \phi h$

which tended to reflect the variation of amplitude from the tests, and which could be considered as a useful parameter for codification.

5.2 Divergent response

The onset of large amplitude instabilities were detected and the wind speed at which this occurred was recorded. Such response was detected in bending for model M1 only and in torsion for models M1, M3 and M5.

As expected, no coupled mode response was detected, despite the adoption of relatively low frequency ratios ranging from approximately 1.4 to 1.9.

The reduced wind speeds at which large amplitude instabilities were detected are set out in Tables 3a and 3b, covering all fascia beams and parapet configurations within the range tabulated.

For bending there was very little effect of changes in parapet configuration and the critical speeds only increased marginally with increasing damping, up to a maximum of about 8% over a damping range from $\delta = 0.02$ to $\delta = 0.06$.

The only exception was model M1/M2 with an overhang of one quarter the depth and with parapets P3b. For this case, for both $k/d_4 = 0.3$ and 0.5, the galloping wind speed increased approximately linearly with damping, ie:

δ_{s}	k/d_4	$V/f_B d_4$	$V/f_B d_4 \delta_s$
0.02	0.3	18.8	940
	0.5	18.4	920
0.04	0.3	61.5	1540
	0.5	58.5	1460
0.06	0.3	75.7	1262
	0.5	69.7	1162

For torsion the critical speed increased by up to about 25% over the same damping range.

6 Application to design rules

6.1 General

In order to compare the results from the wind tunnel tests with the predictions from the Rules in the 1993 version of BD 49, all the results were tabulated and compared with these predictions. These comparisons are set out in Appendix B.

These figures were then used to plot the more significant findings and to highlight where the comparisons were satisfactory and/or where the 1993 version of the Rules would need modification.

6.2 Vortex excitation - critical wind speed

As may be seen from Appendix B sheets Vertical Bending/ M1X, M1Y, M3X, M3Y, M5X and M5Y, the critical wind speed for vertical bending was virtually independent of damping. The results are summarised in Figures 17a and 17b, where the non-dimensional wind speed is plotted against b^*/d_4 for fascia beams X and Y respectively. This parameter is used in the Rules and the design curve is also plotted on the figure appropriate to both bending and torsion. The results for each model are plotted over a band of b^*/d_4 for the sake of clarity; in reality the values are:

Model	b*/d ₄
M1	3.0
M3	7.5
M5	12.0

Each set of results, where relevant, contains values for differing damping values.

From this it can be seen that for model M1 the Rules provide a good lower bound to the predicted critical wind speed. For model M3 the Rules can be seen to be conservative but for model M5 they tend to over-estimate the critical wind speed. By comparing Figures 17a and 17b it can be seen that models with the differing fascia beams show very similar behaviour.

Figures 18a and 18b show the corresponding results for torsional response. Ignoring the sub-harmonics, described in Section 5.1, and shown in Figures 18a and 18b, the prediction of critical wind speeds by the Rules can be seen to be reasonable, although in this case the results for model M3 can be seen to straddle the design curve.

The effect of varying the leading edge details on the critical wind speeds are shown in Figures 19 and 20, where the non-dimensional critical wind speed is plotted against the overhang length for models M1 and M3. They tend to show that the critical wind speed is not sensitive to the leading edge detail, although as seen in Figure 19, the results for model M3 in bending again were consistently higher than those for model M1 (M2). The critical wind speeds for torsion however (see Figure 20), for those models which displayed some excitation, whilst again being constant for each basic model for the various edge details tested, showed a marginal decrease from model M1 to M3.

г ·							Parap	pet				
Fascia beam	Angle	Damping	Pla	P1b	P1d	P2a	P2b	P2d	P3a	P3b	P3d	P4a
	V	1										
Model M	13 (torsion) $\frac{V}{f_1}$	\overline{b}										
x	0°	.02				4.90	2.90		4.55	2.22		
		.04				5.22	3.13		4.72	2.80		
		.06				5.35	3.70		5.01	3.01		
Y	0°	.02				4.71	3.13		5.08	2.85		
		.04				4.92	3.48		5.34	3.37		
		.06				5.10	3.69		5.52	3.89		
Model N	(torsion) _	/										
model M	15 (torsion) $\overline{f_1}$	_r b										
x	0°	.02					3.28		3.82	2.86		
		.04					3.32		3.92	2.95		
		.06					3.42		3.96	3.03		
Y	0°	.02					3.54		4.30	3.18		
		.04					3.61		4.38	3.27		
		.06					3.68		4.43	3.43		
			V									
Model N	11 (galloping (I	B)) $\frac{V}{f_B d_4}$ or (torsid	$pn(T)) \frac{V}{f_T b}$									
Model M	11 (galloping (i	B)) $\frac{V}{f_B d_4}$ or (torsid	$\frac{V}{f_T b}$	P1b	P	1d	P2b	P2	2d	P3a	P3d	P4a
Model M	11 (galloping (i	B)) $\frac{V}{f_B d_4}$ or (torside		P1b	<i>P</i> 	1d T	$\frac{P2b}{T}$		2 <u>d</u>	P3a	P3d	P4a
	11 (galloping (1	$(100) \frac{V}{f_B d_4} \text{ or (torsid)}$	Pla	P1b						P3a	P3d	P4a
		.02 .04	Pla	P1b	В	T 3.15 3.28				P3a	P3d	P4a
		.02	Pla	P1b	B	<i>T</i> 3.15				P3a	P3d	P4a
		.02 .04 .06 .02	<u><i>P1a</i></u> <u>T</u> 3.80	3.42	B - - - 22.5	<i>T</i> 3.15 3.28 3.51 3.49		<i>B</i> 22.1	<i>T</i> 4.06	P3a	21.2	21.0
	-5°	.02 .04 .06 .02 .04	<u>P1a</u> T 3.80 4.04	3.42 3.66	B - - 22.5 23.4	<i>T</i> 3.15 3.28 3.51 3.49 3.79	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
	-5°	.02 .04 .06 .02	<u><i>P1a</i></u> <u>T</u> 3.80	3.42	B - - - 22.5	<i>T</i> 3.15 3.28 3.51 3.49		<i>B</i> 22.1	<i>T</i> 4.06	P3a	21.2	21.0
	-5° 0°	.02 .04 .06 .02 .04 .06	<u>P1a</u> T 3.80 4.04 4.40	3.42 3.66	B - - 22.5 23.4 23.7	<i>T</i> 3.15 3.28 3.51 3.49 3.79	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
	-5°	.02 .04 .06 .02 .04	<u>P1a</u> T 3.80 4.04	3.42 3.66	B - - 22.5 23.4	<i>T</i> 3.15 3.28 3.51 3.49 3.79	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
	-5° 0°	.02 .04 .06 .02 .04 .06 .02	P1a T 3.80 4.04 4.40 3.69*	3.42 3.66	B - - 22.5 23.4 23.7 21.6	<i>T</i> 3.15 3.28 3.51 3.49 3.79	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
X	-5° 0°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06	P1a T 3.80 4.04 4.40 3.69* 3.78*	3.42 3.66	B - - 22.5 23.4 23.7 21.6 22.1	<i>T</i> 3.15 3.28 3.51 3.49 3.79 4.11	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
x	-5° 0° +5°	.02 .04 .06 .02 .04 .06 .02 .04 .06	P1a T 3.80 4.04 4.40 3.69* 3.78*	3.42 3.66	B - - - 22.5 23.4 23.7 21.6 22.1 22.2	<i>T</i> 3.15 3.28 3.51 3.49 3.79 4.11	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
X	-5° 0° +5°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06	P1a T 3.80 4.04 4.40 3.69* 3.78*	3.42 3.66	B - - - 22.5 23.4 23.7 21.6 22.1 22.2 -	<i>T</i> 3.15 3.28 3.51 3.49 3.79 4.11	T 3.39 3.65	B 22.1 22.8	T 4.06 4.44	P3a	21.2 21.8	21.0 21.8
X	-5° 0° +5°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02	P1a T 3.80 4.04 4.04 3.69* 3.78* 4.03* 3.79*	3.42 3.66 3.88 3.30	B - - - 22.5 23.4 23.7 21.6 22.1 22.2 -	T 3.15 3.28 3.51 3.49 3.79 4.11 3.39 3.48	T 3.39 3.65 4.22 4.24	B 22.1 22.8 23.5 25.0	T 4.06 4.44	<i>P3a</i>	21.2 21.8 22.5 21.6	21.0 21.8 22.0 22.0
X	-5° 0° +5° -5°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06	P1a T 3.80 4.04 4.04 3.69* 3.78* 4.03* 3.79* 3.92*	3.42 3.66 3.88 3.30 3.30 3.66	B - - - 22.5 23.4 23.7 21.6 22.1 22.2 - - - -	T 3.15 3.28 3.51 3.49 3.79 4.11 3.39 3.48	T 3.39 3.65 4.22 4.24 4.38	B 22.1 22.8 23.5 25.0 26.0	T 4.06 4.44	P3a	21.2 21.8 22.5 21.6 22.3	21.0 21.8 22.0 22.0 22.4
x	-5° 0° +5° -5°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02	P1a T 3.80 4.04 4.04 3.69* 3.78* 4.03* 3.79*	3.42 3.66 3.88 3.30	B - - - 22.5 23.4 23.7 21.6 22.1 22.2 - - - 20.4	T 3.15 3.28 3.51 3.49 3.79 4.11 3.39 3.48	T 3.39 3.65 4.22 4.24	B 22.1 22.8 23.5 25.0	T 4.06 4.44	P3a	21.2 21.8 22.5 21.6	21.0 21.8 22.0 22.0
X	-5° 0° +5° -5°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06	P1a T 3.80 4.04 4.04 3.69* 3.78* 4.03* 3.79* 3.92*	3.42 3.66 3.88 3.30 3.30 3.66	B - - - 22.5 23.4 23.7 21.6 22.1 22.2 - - 20.4 - 20.4 - 20.4 - 25.2	T 3.15 3.28 3.51 3.49 3.79 4.11 3.39 3.48	T 3.39 3.65 4.22 4.24 4.38	B 22.1 22.8 23.5 25.0 26.0	T 4.06 4.44	P3a	21.2 21.8 22.5 21.6 22.3	21.0 21.8 22.0 22.0 22.4
Model M.	-5° 0° +5° -5°	.02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06 .02 .04 .06	P1a T 3.80 4.04 4.04 3.69* 3.78* 4.03* 3.79* 3.92*	3.42 3.66 3.88 3.30 3.30 3.66	B - - - 22.5 23.4 23.7 21.6 22.1 22.2 - - - - 20.4 - - - - - - - - - - - - -	T 3.15 3.28 3.51 3.49 3.79 4.11 3.39 3.48	T 3.39 3.65 4.22 4.24 4.38	B 22.1 22.8 23.5 25.0 26.0	T 4.06 4.44	P3a	21.2 21.8 22.5 21.6 22.3	21.0 21.8 22.0 22.0 22.4

Table 3a Reduced wind speed for large amplitude instabilities

*values at 0° with middle third solid

0		D .			Edge detail	$\binom{k}{d4}$ or type		
Overhang L/d ₄	Parapet	Damping δ_s	0.1(X)	0.3(Y)	0.4(Y)	0.5(Y)	U	U^{I}
Model M3 (to	$(\frac{V}{f_T b})$							
1.0	2a	0.02		4.5	4.9	4.5	3.7	4.6
		0.04		4.7	5.2	4.8	3.9	4.9
		0.06		5.0	5.3	4.9	4.1	5.2
0.5	2a	0.02	4.0	3.5		3.7	3.8	3.7
		0.04	4.2	3.8		4.0	4.1	4.0
		0.06	4.5	4.0		4.2	4.3	4.2
	2b	0.02	2.9			3.2		
		0.04	3.5			3.6		
		0.06	3.8			3.8		
0.25	2a	0.02		3.7		3.6		3.0
		0.04		4.0		3.8		3.5
		0.06		4.2		4.0		3.8
	2b	0.02				3.2		
		0.04				3.8		
		0.06				4.1		
M1/M2 (galla	$(\frac{V}{f_B d_4})$							
0.5	2a	0.02				22.2		20.1
		0.04				22.5		21.0
		0.06				23.1		22.0
	3a	0.02				23.0		21.9
		0.04				23.5		22.7
		0.06				24.0		24.3
0.25	2a	0.02				17.5		14.9
		0.04				17.6		15.1
		0.06				17.8		15.2
	3b	0.02		18.8		18.4		
		0.04		61.5		58.5		
		0.06		75.7		69.7		

Table 3b Reduced wind speed for large amplitude instabilities (All tested at 0° inclination)

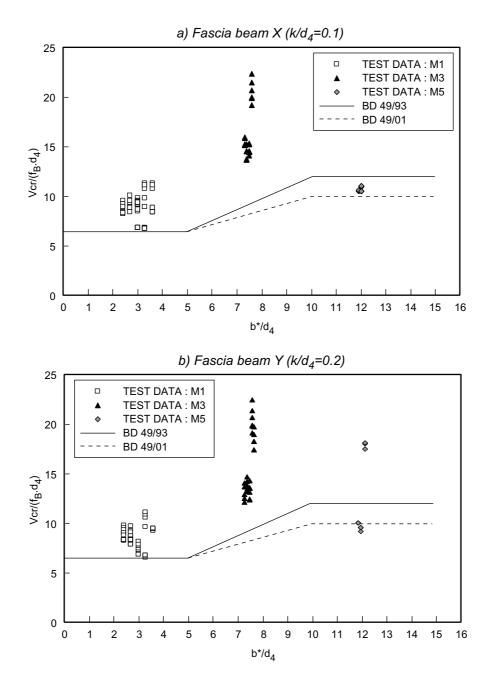


Figure 17 Critical wind speeds for vortex excitation (bending)

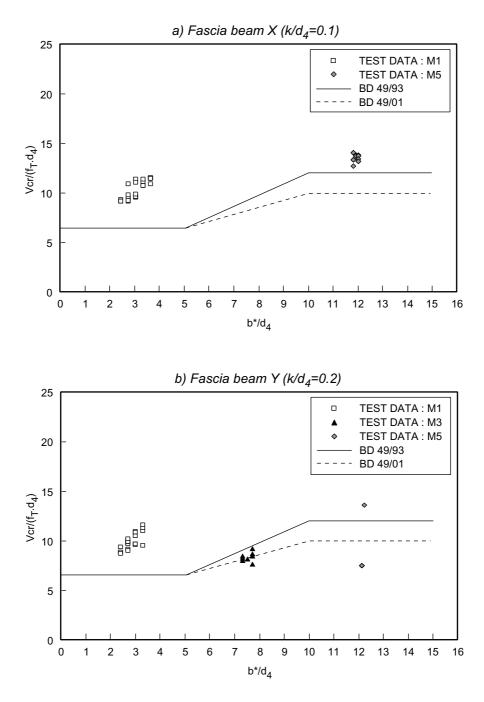
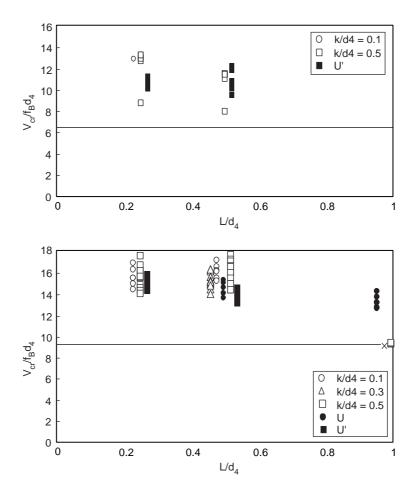
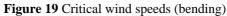


Figure 18 Critical wind speeds for vortex excitation (torsion)





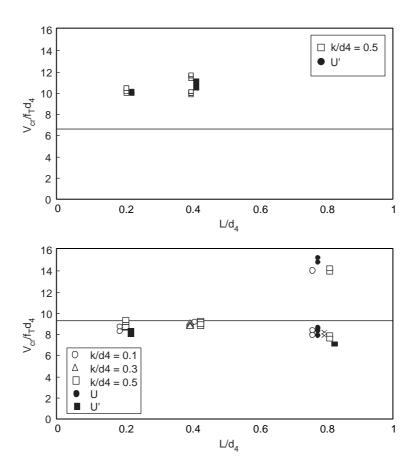


Figure 20 Critical wind speeds (torsion)

These results suggest therefore that it may be prudent to decrease the prediction of critical wind speed, V_{cr} , in bending and torsion for higher b*/d₄ ratios for plate girders using:

$$\begin{array}{rcl} V_{\rm cr} &=& 6.5 {\rm fd}_4 \mbox{ for } b^*/d_4 < 5 \\ V_{\rm cr} &=& {\rm fd}_4 \mbox{ (} 0.7 \mbox{ } b^*/d_4 + 3.0 \mbox{) for } 5 \leq b^*/d_4 < 10 \\ V_{\rm cr} &=& 10 {\rm fd}_4 \mbox{ for } b^*/d_4 \geq 10 \end{array}$$

These revised values are shown on Figures 19 and 20.

6.3 Vortex excitation - amplitudes

6.3.1 Bending

The vertical bending amplitudes for each model are shown in Figure 21a for fascia beam X and Figure 21b for fascia beam Y, plotted against the logarithmic decrement of damping. The relevant curves from the design rules, appropriate to each model are also plotted.

From these figures it can be seen that the amplitudes from many tests exceeded the predicted amplitudes, particularly at the higher levels of damping. However these plots show the results for all tests whereas many of the configurations do not comply with the restraints set out in the current Rules for edge geometry, and only within which the simplified formulae for amplitudes in the Rules apply.

By omitting all those configurations that do not conform to these restraints, the Rules can be seen to apply more satisfactorily, as shown in Figures 22a and 22b. Only the model with parapet heights of 0.7 times the depth and low solidity exceed the design curve at damping values greater than 0.03.

Figure 23 shows the ratio of measured amplitudes to predicted amplitudes for models M1 and M3 normalised to 1.0 at $\delta = 0.03$. This indicates the trend of the test results to increase more rapidly than the accepted linear factor against damping, as incorporated in the Rules.

Comparisons have been made with the earlier tests carried out at BMT (formerly NMI) on box girder bridges which formed the basis of the Rules. These showed that the Rules generally predicted amplitudes between about 100% and 300% of the measured values for those configurations meeting the geometric criteria of the Rules. Typical results, in the form provided for the plate girder results in Appendix B, are given for three models tested in Appendix C.

6.3.2 Torsion

Torsional amplitudes for all test configurations are shown in Figure 24a for fascia beam X and Figure 24b for fascia beam Y. Once again several results are above the curve from the design rules. However when those configurations which do not meet the criteria for edge geometry to the Rules are omitted, as shown in Figures 25a and 25b, then it can be seen that the Rules provide very conservative values of torsional amplitudes. Note that only models M1 and M3 complied with the restraints, and for fascia beam X all the results showed stable behaviour.

6.4 Large amplitude instabilities

The Rules stipulate that plate girder bridges need not be considered for vertical galloping, the instability only being considered appropriate for box girder sections of relatively narrow aspect ratio ($b^*/d < 4$). However the tests showed that vertical bending instability did occur for all configurations of model M1 with solid parapets as shown in Figure 26a for fascia beam X and Figure 26b for fascia beam Y.

The Rules, when applied, assume that the critical wind speed increases with structural damping, but this was not reflected in the test results, as can be seen from these figures. However these models, with solid barriers, would not conform with the geometric criteria of the Rules, so from the point of view of design this is irrelevant. Nevertheless this needs consideration and possibly further study to explain this apparently unexpected result.

Torsional motion was detected in all three models, the results for which are shown in Figures 27a and 27b, for fascia beams X and Y respectively. It can be seen that the Rules overestimate the critical wind speed, which would be non-conservative. However again the edge geometry for many of the test configurations, including that for all of model M5 sections, does not comply with the criteria in the Rules for which the formulation for critical wind speed applies. Figures 28a and 28b give the results for those configurations which do comply from which it can be seen that whereas the results for model M3 are close to the Rules, those for model M1 are significantly below the Rule.

The Rules provide more onerous criteria for box girder sections - particularly for relatively shallow configurations. For model M1, if treated as a box girder, this would produce a non-dimensional critical wind speed Vg/ f_T b of 2.6, thereby providing a lower bound to the measured wind speeds. However if applied to models M3 and M5, the values would be 1.3 and 0.9 thus providing very pessimistic estimates of critical wind speed.

In practice the Rules require the designer to undertake a calculation to predict the critical wind speed for classical flutter for all bridge types (Clause 2.1.3.3).

The lower bound non-dimensional wind speed from this calculation is 2.5, so it is possible that the use of the nonconservative clause for the single degree of freedom instability in torsion would be prevented by the value from Clause 2.1.3.3. However, Clause 2.1.3.2(b) should be altered to allow for plate girder bridges, with the value of non-dimensional wind speed obtained from Clause 2.1.3.3 only being used if it is the lower of the two.

7 Amendments to the rules

7.1 Vortex excitation - critical wind speeds

For the plate girder bridge models tested, the critical wind speeds did not follow the trends of the Rules, although for model M1 the Rules provided an excellent lower bound to the results for bending and a reasonable lower bound for torsion. For model M3 the Rules were conservative for bending but marginally non-conservative for torsion. For model M5 they provided over-estimates (that is nonconservative values), compared with the tests, for certain configurations in bending.

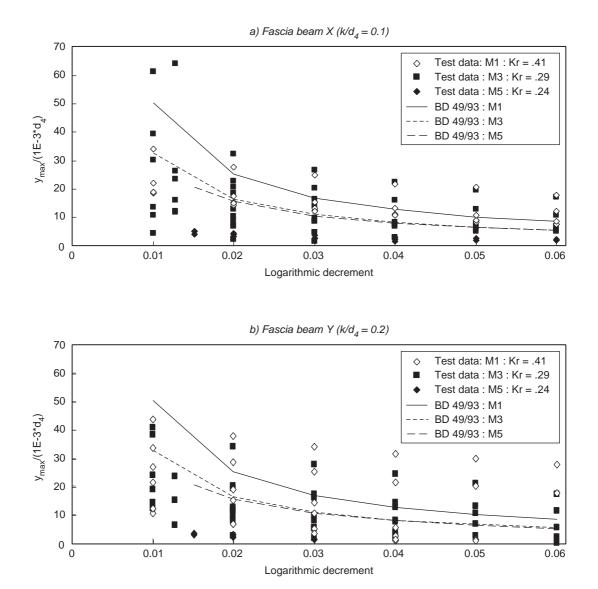


Figure 21 Amplitudes for vortex excitation (bending)



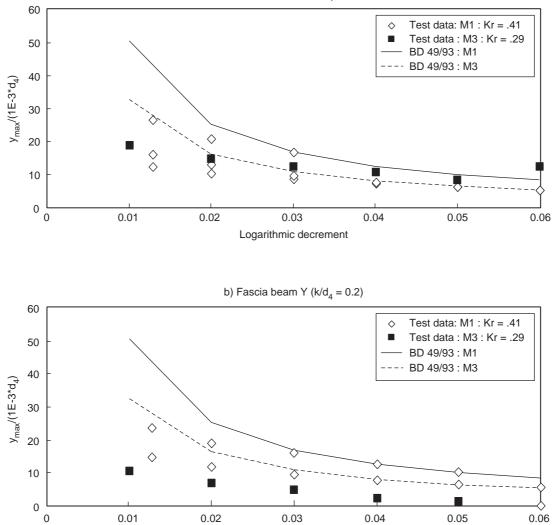


Figure 22 Amplitudes for vortex excitation (bending) – selected results

Logarithmic decrement

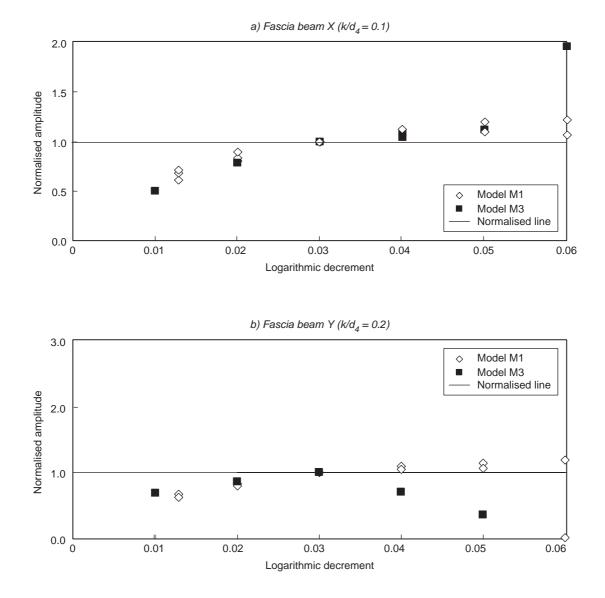


Figure 23 Ratio of test model to code amplitudes for vortex excitation (bending)

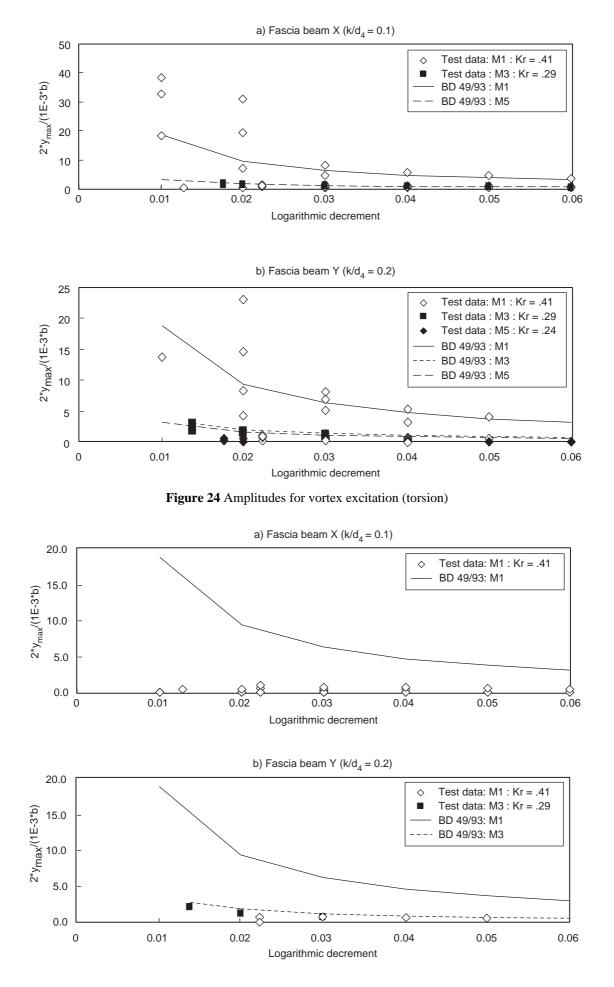


Figure 25 Amplitudes for vortex excitation (torsion)

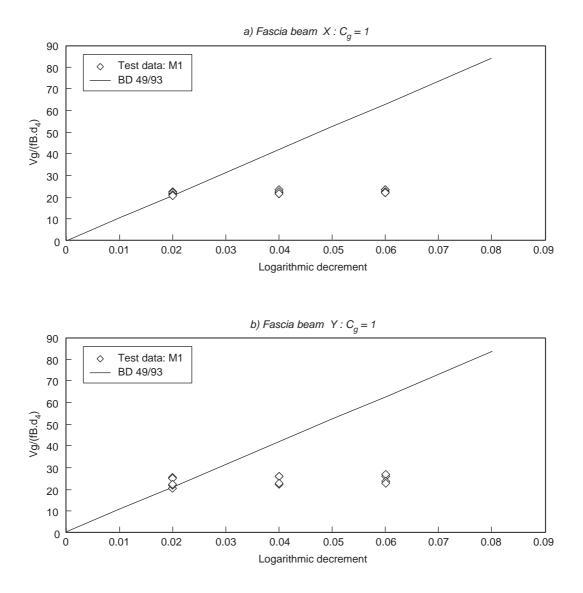


Figure 26 Critical wind speeds for divergent amplitude (bending)

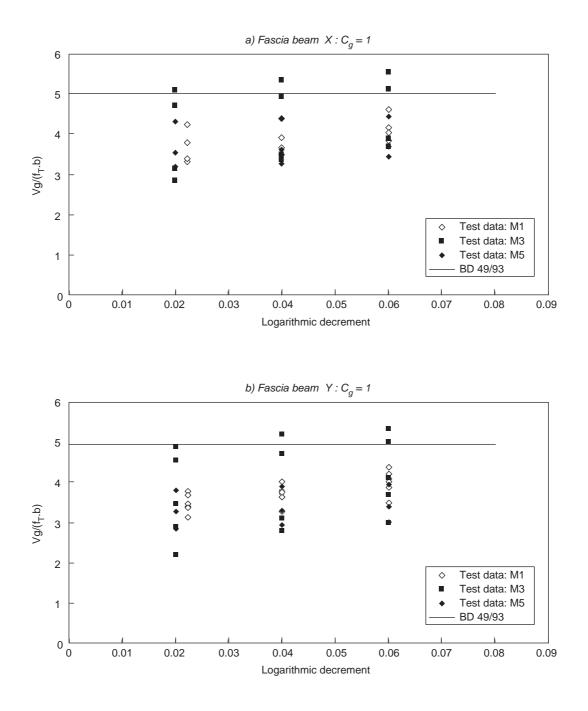


Figure 27 Critical wind speeds for divergent amplitude (torsion)

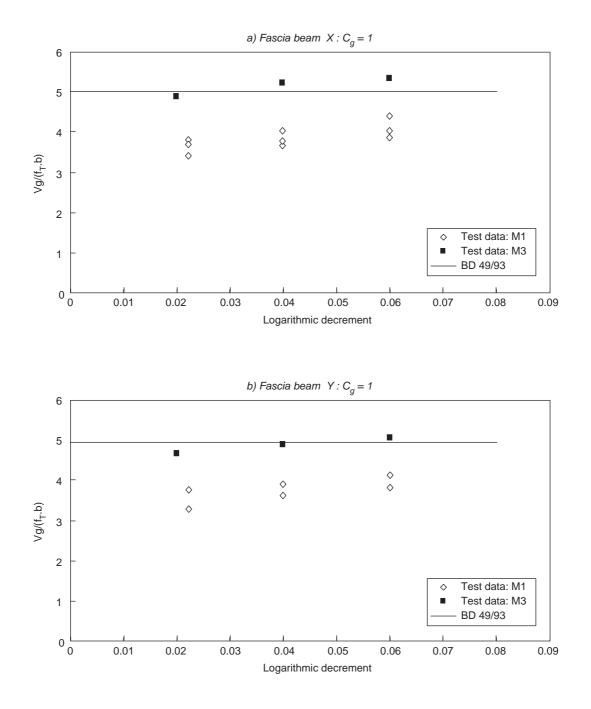


Figure 28 Critical wind speeds for divergent amplitude (torsion)

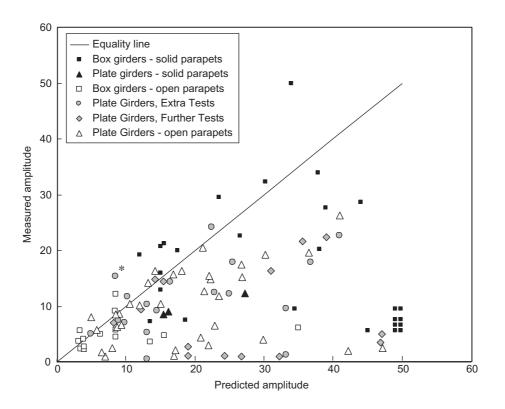


Figure 29 Use of c factor for bending

For torsion one configuration of model M5 provided results below the design Rule prediction, but this appeared to be associated with a sub-harmonic vortex shedding mechanism. The amplitudes associated with this were very small (below the primary results obtained) and it is considered that, for codification purposes, this particular result could be ignored.

Accordingly, for plate girder bridges, the Rule for critical wind speeds is modified to:

$$V_{cr} = 6.5 \text{ fd}_4 \text{ for } b^*/d_4 < 5$$

$$V_{cr} = \text{ fd}_4 (0.7 b^*/d_4 + 3.0) \text{ for } 5 \le b^*/d_4 < 10$$

$$V_{--} = 10 \text{ fd}_4 \text{ for } b^*/d_4 \ge 10$$

In general terms, the equations for V_{cr} in clause 2.1.1.2 may be written as:

For
$$b^*/d_4 < 5$$
:

$$V_{cr} = 6.5 fd$$

For $5 \le b^*/d_4 < 10$:

$$V_{cr} = fd_4 (1.1 b^*/d_4 + 1.0) \text{ for box girder bridges} (types 1, 1A, 3, 3A, 4, 4A) = fd_4 (0.7 b^*/d_4 + 3.0) \text{ for plate girder bridges} (types 2, 5 and 6)$$

For $10 \le b^*/d_4$: V =

$$V_{\rm cr} = 12 {\rm fd}_4$$
 for box girder bridges
= 10 {\rm fd}_4 for plate girder bridges

7.2 Vortex excitation - amplitudes

The test results showed that the Rules provide upper bound predictions of amplitude for configurations whose edge

deck geometry complies with the restraints of the present Rules. However it was not feasible to relax these restraints as large amplitudes were recorded for several of such configurations, although no consistent pattern emerged.

The only possible relaxation for plate girders was to increase the limit of the product h ϕ in Clause 2.1.3.2(a)(ii) from 0.25d₄ as at present to 0.35d₄. This relaxation would only apply to plate girders, but may be of value, particularly for shallow bridge sections.

Whilst undertaking the study it was pointed out that the restriction of edge beams to be less than $0.2d_4$ deep (Clause 2.1.3.2(a)(i)) was leading to anomalies in design where, for example a 300mm deep slab on a 1m deep section was acceptable provided there was no upstand at the edge. However if the slab was thickened at the edge, to say 320mm, then the design would not meet this criteria, having to be treated as an 'edge beam'. It is proposed that for the purposes of defining such members, edge stiffening of the slab to, say, half the slab depth could be ignored.

The amplitudes in the current Rules need to be factored by 3.0 for beams without overhangs but no guidance is given concerning the minimum overhang which would qualify to escape this threefold multiplication of the response. For galloping behaviour the threshold for more severe behaviour was set when the overhang is less than 0.7d and the results of the tests were used to assess whether this limitation should be adopted for vortex excitation. The results of the tests with varying overhang and fascia beam depths which showed the variation of amplitude with 'effective depth', as noted in Section 6.1 above however, led to the consideration of a modification of the Rules which would incorporate both this parameter, and the above factor of 3.0. The result of this study is that a factor on the predicted amplitudes could be adopted for *all* sections of:

$$c = 3\left[\frac{k}{d_4} + \frac{h_{\phi}}{d_4}\right] but not less than 1.0$$

where k = depth of fascia beam, or edge slab;

- d_4 = reference depth of the bridge;
- h = height of parapet or other edge member above normal deck level;
- ϕ = solidity ratio of parapet;

all as defined in Figures 2 and 3, with h, k and d_4 in consistent units.

This would eliminate the necessity of the factor 3.0 and would allow dense, and even solid, barriers to come within the scope of the Rules. However as the tests did not comprehensively cover wind inclinations of up to $\pm 5^{\circ}$, which on previous studies had been shown to be critical, it is proposed that solid barriers are still excluded for the present.

The validity of this proposal is shown in Figure 29, for bending and Figure 30 for torsion for *all* tests undertaken on both box girder and plate girder bridges; the results are given for the range of damping, considered appropriate for steel or composite structures.

In Figure 29 the measured amplitude is plotted against the predicted amplitude incorporating the factor c. The rule is satisfied for all tests which lie below the 45° line shown on the graph. It can be seen that the rule is thus generally satisfied with outliers being either sections with solid parapets (which are not covered by the Rules) or where the factor c being not less than 1 would ensure the rule was satisfied.

An exception to this is the test on plate girder model M3 noted in 6.1 above and asterisked on the figure. This result was carefully re-examined, but appears to have been correctly tabulated. The high value may possibly be due to coupled motion between bending and torsion. Figure 30 shows the equivalent figure for torsion.

Figure 31 shows the results for both bending and torsion plotted non-dimensionally as the ratio of the maximum measured value to the predicted value by the current rule for the damping range $\delta = 0.02$ to 0.05, against factor c. Here it can be seen that the use of the factor provides a good envelope to the majority of the results, but again highlighting the same test (denoted with an asterisk) as an outlier.

The proposed factor c for the configurations tested is shown in Table 4 from which it can be seen that a factor less than unity would be obtained for five configurations if a lower limit were not used. The lower value was 0.5, which still appeared to provide a satisfactory factor. Without testing values with a lower factor it is proposed to use this as a lower bound. In addition the factor is greater than 3 for several of the configurations in such cases specific tests could provide more beneficial results as the proposed rule becomes more penal.

What was significant was that the torsional amplitudes for configurations which comply with the Rules were almost negligible and it is suggested that torsional response due to vortex shedding for such plate girder bridges could be ignored.

The tabulated results from the series of tests can be used for design profiles that closely match a tested configuration, thereby avoiding the use of over-conservative design rules contained in BD 49.

7.3 General

The wind tunnel test results for bending using solid parapets resulted in critical wind speeds for divergence which were, surprisingly, independent of damping. At the lowest value of damping d = 0.02 the Rules provided a good estimate of the critical speed. However at higher values of damping the Rules predicted higher critical wind speeds (in inverse proportion to the damping). The Rules however, as written, do not require a check for plate girders in vertical bending divergence, and galloping

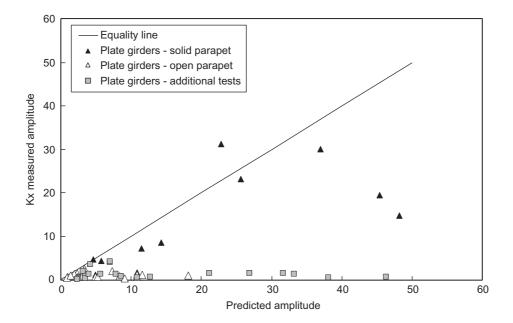


Figure 30 Use of c factor for torsion

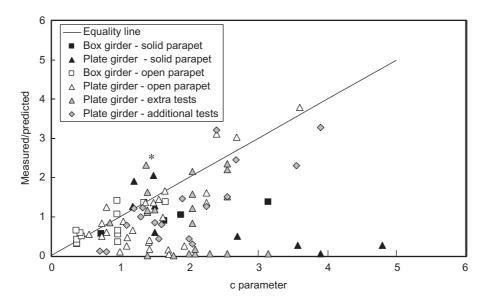


Figure 31 Non-dimensional plot of c factor

Table 4 Proposed fac	ctor c for various	configurations	of edge details	s and barriers

						k/d ₄ [see Figure 3]		
	Barrier solidity	Parapet P-type	$h\phi/d_4$	0.1 or 'X'	0.2 or 'Y'	U - type or 0.3 Y type	0.4 Y type	U' - type or 0.5 Y type
Any	'Open'	1a	0.075	0.525	0.825	1.125	1.425	1.725
overall width of	barriers $\phi \leq \frac{1}{2}$	1b	0.150	0.750	1.050	1.350	1.650	1.950
bridge	$\Psi = r_2$	2a	0.175	0.825	1.125	1.425	1.725	2.025
		2b	0.350	1.350	1.650	1.950	2.250	2.550
		3a	0.275	1.125	1.425	1.725	2.025	2.325
		3b	0.550	1.950	2.250	2.550	2.850	3.150
		4a	0.375	1.425	1.725	2.025	2.325	2.625
		4b	0.750	2.550	2.850	3.150	3.450	3.750
Any	'Dense'	1c	0.225	0.975	1.275	1.575	1.875	2.175
overall width of	barriers $\phi > \frac{1}{2}$	1d	0.300	1.200	1.500	1.800	2.100	2.400
bridge	$\psi > \gamma_2$	2c	0.525	1.875	2.175	2.475	2.775	3.075
		2d	0.700	2.400	2.700	3.000	3.300	3.600
		3c	0.825	2.775	3.075	3.375	3.675	3.975
		3d	1.100	3.600	3.900	4.200	4.500	4.800
		4c	1.125	3.675	3.975	4.275	4.575	4.875
		4d	1.500	4.800	5.100	5.400	5.700	6.000
See Figure	1 S	ee Figure 2				Values of c factor		

 $h\phi/d_4$ may be taken as $h\phi/d_4 + a\phi_b/b_4$ in cases where there are barriers of height a as well as parapets of height h.

Shows where factor is > 3.0, where tests may be more appropriate, and may reap appreciable benefits when the factors are high. Note: the majority of these are with 'dense' barriers, which should be less common in practice.

(particularly for the wider sections) seems unlikely. In any event these results were with solid parapets and are therefore outside the scope of the Rules. It is not believed at this stage that modifications to the Rules can be justified on the basis of these tests, but that further study of the mechanism causing this divergence is recommended.

For torsion the Rules tend to provide a non-conservative prediction of the critical wind speed. Whilst it is likely that this is being partly catered for by the need to calculate the wind speed for classical flutter (as detailed in Section 7.4), Clause 2.3.1.2(b) has been modified as follows:

 $V_g = 3.3 f_T b$, for plate girder bridges

 $V_{g}^{s} = 5.0 f_{T}^{s} b$, for all other bridges

8 Conclusions

The wind tunnel tests described have provided, for the first time, a parametric study of the aerodynamic behaviour of plate girder bridges. Additional tests were undertaken on plate girder bridges with relatively short overhangs and edge details conforming more closely to typical highway bridges and footbridges. The results of the wind tunnel tests, whilst limited in scope from the complete matrix of tests for the models envisaged, have confirmed the general trend of the Rules and provided added confidence in their wider use to plate girder bridges. However there have been unexpected results and inconsistencies, making sound general design guidance extremely difficult.

Whereas it was hoped that some relaxation of the Rules could be produced for plate girder bridges, this has generally not been the case, apart from torsional vibrations due to vortex shedding which it is proposed could be ignored.

As a result of these tests, modified rules have been developed for the prediction of vortex excited amplitudes which provide better agreement with test results. These modified rules have now been incorporated into the revised version of BD 49 (Highways Agency 2001).

It should be noted however that this form of response is very sensitive to small changes in edge details, making codified rules very conservative in many cases. The BD 49 clauses provide reasonable rules which the majority of sections will satisfy. However, it must be recognised that a few sections which satisfy the rules may subsequently show some signs of response in service.

The study has shown the inconsistent patterns that can be obtained due to small changes in leading edge detail. Whilst some 61 different configurations were tested, this covered a wide range of parameters and frequently trends had to be predicted from only two tests on the variation of one of those parameters.

The matrix of tests undertaken as reported here provides a valuable set of data, which can be used in those cases where the design closely matches one of the tested configurations. This permits a much less conservative analysis to be used.

9 References

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Definition of Terms

- B Overall width of model (b in rules)
- $d_4 Depth of girder (including deck slab)$
- M Mass/m of model set-up (m in rules)
- I Mass moment of inertia/m of model set-up (equivalent to mr² in rules)
- U Mean wind speed at model location (V in rules)
- $n_z Natural frequency in bending (f_B in rules)$
- n_{θ} Natural frequency in torsion (f_{T} in rules)
- z Maximum rms deflection due to vortex shedding (y_{max} in rules)
- θ Maximum rms rotation due to vortex shedding (expressed in terms of y_{max} in rules)
- δ Logarithmic decrement of structural damping (δ_s in rules)
- $\rho \quad \quad Density \ of \ air$

Angle 0°

Model M1, B = 225mm,

Parapet Pla, Edge Detail X, d4 = 50mm

Vortex Shedding in Bending

Logdec	MőløB ²	U/nzd4	z/d_
0.0128	0.582	8.28	0,0296
0.02	606'0	8.33	0.0248
0.03	1.363	8.39	0.0206
0.04	1.818	8.39	0.0170
0.05	2.272	8.36	0.0148
0.06	2.726	8.31	0.0126

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

Logdec	18/pB ⁴	U/n _p B
0.0223	0.119	3.80
0.04	0.214	4.04
0.06	0.320	4.40

Angle 0° Parapet P1b, Edge Detail X, $d_4 = 50mm$

Vortex Shedding in Bending

ogdec	M8/pB ²	U/n _z d ₄	*p/z
0.0128	0.582	8.83	0.0644
0.02	606.0	8.87	0.0500
0.03	1.363	8.42	0.0402

Vortex Shedding in Torsion

ogdec	Iå/µB⁴	U/ngd4	θ(°)
0.0223	0.119		0.087
0.03	0.160	9,81	0.061
0.04	0.214		0.057

Divergent Instability

Torsional Instability

Logdec	Ið/µB⁴	U/n ₉ B
0.0223	0.119	3.42
0.04	0.214	3.66
0.06	0.320	3.68

Model M1, B = 225mm,

Parapet P1a (middle third solid) $d_4 = 50mm$

Edge Detail X, B = 225mm,

Model M1, Angle 0° Vortex Shedding in Bending

Logdec	M5/pB2	U/n _z d ₄	zida
0.0128	0.582	8.83	0.0388
0.02	0.909	8.90	0.0310
0.03	1.363	8.43	0.0232
0.04	1.818	8.46	0.0184
0.05	2.272	8.46	0.0152
0.06	2.726	8.45	0.0124

Vortex Shedding in Torsion

ogdec	Iô/pB ⁴	U/ngd4	6()0
0.0223	0.119	10.94	0.140
0.03	0.160	11.52	0.105
0.04	0.214	11.51	0.089
0.05	0.267	11.51	0.070
0.06	0.320	11.49	0.061

Divergent Instability

Torsional Instability

Logdec	Iδ/ρB ⁴	U/n _# B
0.0223	0.119	3.69
0.04	0.214	3.78
0.06	0.320	4.03

Angle 0°

Model M1, Edge Detail X, B = 225mm, $d_4 = 50mm$

Parapet P2b,

Vortex Shedding in Bending

z/d4	0.0106	0.0038
$U/n_z d_4$	10.80	11.14
Mő/pB ²	0.518	1.553
Logdec	0.01	0.03

Vortex Shedding in Torsion

ogdec	16/µB ⁴	U/ngd4	û(°)
0.0223	0.119	11.17	0.078
5U.0	0.100	11.40	140.0

Divergent Instability

gdec	I5/pB ⁴	U/n _p B
0.0223	0.119	3.39
t	0.214	3.65
8	0.320	4.22

Angle -5°

Model M1, B = 225mm,

Parapet P1d, Edge Detail X, d4 = 50mm

Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _z d ₄	z/d4
0.0128	0.582	9.18	0.0288
0.02	606.0	9.18	0.0220
0.03	1.363	9.41	0.0212
0.04	1.818	9.39	0.0198
0.05	2.272	9.42	0.0180
0.06	2.726	9.39	0.0166

Vortex Shedding in Torsion

Logdec	Ič/µB⁴	U/ngd4	(")0
0.0223	0.119	9.62	0.215
0.03	0.160	9.61	0.194
0.04	0.214	9.62	0.165
0.05	0.267	9.63	0.137
0.06	0.320	9,90	0.122

Divergent Instability

Torsional Instability

Logdec	$I_0^{0/\rho}B^4$	U/n _s B
0.0223	0.119	3.15
0.04	0.214	3.28
0.06	0.320	3.51

Parapet Pld.

Edge Detail X, d₄ = 50mm

Angle 0°

Vortex Shedding in Bending

Logdec	$M\delta/\rho B^2$	U/n _z d ₄	z/d4
0.01	0.518	10.08	0.0740
0.02	1.035	9.23	0.0554
0.03	1.553	9.53	0.0494
0.04	2.071	8.98	0.0392
0.05	2.589	8.74	0.0312
0.06	3.107	8.75	0.0264

Vortex Shedding in Torsion

ogdec	I3/pB ⁴	U/ngd4	θ(")
101	0.069	10.92	2.540
0.02	0.137	9.50	0.986
0.03	0.206	9.20	0.637

Divergent Instability

Galloping (Bending)

Logdec	$M\delta/\rho B^2$	U/n ₂ d ₄
0.02	1.035	22.5
0.04	2.071	23.4
0.06	3.107	23.7

Model M1, B = 225mm,

Parapet P1d,

Angle +5^a

Model M1, Edge Detail X, B = 225mm, $d_4 = 50mm$

Parapet P2d,

Vortex Shedding in Bending

Logdec	Mő/pB ²	U/nzd4	z/d4
0.01	0.518	10.6	0.0964
0.02	1.035	9.52	0.0788
0.03	1.553	9.24	0.0646
0.04	2.071	9.23	0.0542
0.05	2.589	8.98	0.0474
0.06	3.107	8.98	0.0416

Vortex Shedding in Torsion

ogdec	13/pB ⁴	U/ngd4	θ(°)
0.03	0.206	9.32	1.147
0.04	0.274	9.20	0.808
0.05	0.343	9.19	0.635
0.06	0.412	9.18	0.499

Divergent Instability

Galloping (Bending)

ogdec	$M\delta/\rho B^2$	U/n _z d ₄
0	1.035	21.6
0.04	2,071	22.1
9	3.107	22.2

Bending	
1	
Shedding	
Vortex	

ogdec	Mő/pB ⁴	U/nzd4	P/2
0.01	0.518	9.79	0.1498

Vortex Shedding in Torsion

Logdec	Ið/pB ⁴	U/ngd4	(_a)g
0.02	0.137	11.42	4.350

Divergent Instability

Galloping (Bending)

Logdec	Mő/pB ²	U/n ² d4
0.02	1.035	22.1
0.04	2.071	22.8
0.06	3.107	23.5

Angle 0^e

Model M1, B = 225mm, 36

Edge Detail X, d4 = 50mm

Parapet P3d,

Angle 0°

Model M1, B = 225mm,

Edge Detall X, d4 = 50mm

Parapet P4d,

Angle 0°

Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n _z d ₄	z/d4
0.01	0.518	6.85	0.0330
0.02	1.035	6.85	0.0174
0.03	1.553	6.83	0.0114
0.04	2.071	6.80	0.0068

Vortex Shedding in Torsion

ogdec	Iő/µB ⁴	U/ngd4	(_)
0.015	0.103	11.09	5.360

Divergent Instability

Galloping (Bending)

ogdec	Mő/µB ²	U/n _x d ₄
0.02	1.035	21.2
0.04	2.071	21.8
0.06	3.107	22.5

M
ogdec
-

Vortex Shedding in Bending

Logdec	Mő/pB ²	U/nzd4	z/d4
0.01	0.518	6.80	0.0262
0.02	1.035	6.78	0.0164
0.03	1.553	6.75	0.0112
0.04	2.071	6.78	0.0072

Vortex Shedding in Torsion

6(°)	4.551 2.709
U/ngd4	10.75
18/pB ⁴	0.103 0.137
Logdec	0.015 0.02

Divergent Instability

Galloping (Bending)

ogdec	Mő/pB ²	U/n,d4
0.02	1.035	21.0
0.04	2.071	21.8
0.06	3.107	22.0

Parapet P2d,

Model MI, Edge Detail X, B = 225mm, $d_4 = 50mm$

Angle 0°

Parapet P1d,

Edge Detail X, $d_4 = 50mm$

Model M1, B = 225mm, In Turbulent Flow

In Turbulent Flow

Vortex Shedding in Bending

ogdec	Mő/pB ²	U/n _z d4	z/d4
0128	0.582	8.99	0.1572

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

Logdec.	$1\delta/\rho B^4$	U/n _g B
0.0223	0.119	4.06
0.04	0.214	4,44
0.06	0.320	5.00

Divergent Instability

Vortex Shedding in Torsion

Stable

Torsional Instability

Vortex Shedding in Bending

Logdec	Mő/pB ²	U/nzd4	\$p/z
0.0128	0.582	9.58	0.0570
0.02	0.909	9.46	0.0452
0.03	1.363	9.13	0.0342
0.04	1.818	8.49	0.0164
0.05	2.272	8.67	0.0122

Angle 0°

Edge Detail Y, $d_4 = 50mm$

Parapet Pla,

Vortex Shedding in Bending

dec	Mő/pB ²	U/nzd4	z/d4
0.0128	0.582	8.28	0.0364
~	0.909	8.33	0.0290
en	1.363	8.23	0.0232
4	1.818	8.33	0.0190
5	2.272	8.35	0.0160
9	2.726	8.35	0.0138

Vortex Shedding in Torsion

10/pB	
0.119	

Divergent Instability

Torsional Instability

_	_	1	_
U/n ₉ B	3.79	3.92	4,15
Iô/pB ⁴	0.119	0.214	0.321
Logdec	0.0223	0.04	0.06

Angle 0°

Model M1, B = 225mm,

Edge Detail Y, $d_4 = 50mm$

Parapet P1b,

Vortex Shedding in Bending

ogdec	$M\delta/\rho B^2$	U/n ₂ d ₄	z/d4
0.0128	0.582	8.37	0.0572
0.02	0.909	8.29	0.0468
0.03	1.363	8.32	0.0392
10.0	1.818	8.36	0.0310
0.05	2.272	7.83	0.0252
0.06	2.726	7.89	0.0004

Vortex Shedding in Torsion

Logdec	$I\delta/\rho B^4$	U/ngd4	θ(")
0.0223	0.119	9.59	0.143
0.03	0.160	9.88	0.114
0.04	0.214	9.87	0.100
0.05	0.267	10.15	0.069

Divergent Instability

Torsional Instability

Logdec	lå/ρB ⁴	U/n _p B
0.0223	0.119	3.30
0.04	0.214	3,66
0.06	0.321	3.84

Model M1, B = 225mm,

Angle 0°

Parapet P1d,

Edge Detail Y, $d_4 = 50mm$ Model M1, B = 225mm,

Parapet P2b,

Angle 0°

Model M1, B = 225mm,

Edge Detail Y, d4 = 50mm

Vortex Shedding in Bending

Logdec	$M\delta/\rho B^2$	U/n _z d ₄	*p/z
0.01	0.518	9.76	0.0580
0.02	1.035	9.24	0.0492
0.03	1.553	9.11	0.0416
0.04	2.071	8.68	0.0352
0.05	2.589	8.70	0.0316
0.06	3.107	8.42	0.0272

0.0154 0.0062 0.0044 0.0038

9.62 10.61 10.93 11.10

0.582 0.909 1.363 1.818

0.0128 0.02 0.03 0.04

z/d_4

U/nzd4

Må/pB²

Logdec

Vortex Shedding in Bending

Vortex Shedding in Torsion

Lordec	15/aB ⁴	U/nad.	θ(*)
		to Acro	
0.01	0.069	10.08	1.915
0.02	0.137	9.67	1.168
0.03	0.206	60'6	0.715
0.04	0.274	8.94	0.451

Divergent Instabilities

Galloping (Bending)

Logdec	Mő/µB ²	U/nzd4
0.02	0.909	26.6
0.04	1.818	26.9
0.06	2.726	27.2

39

Vortex Shedding in Torsion

Logdec	Ið/pB ⁴	U/ngd4	θ(°)
0.0223	0.119	11.00	0.115
0.03	0.160	11.29	0.083
0.04	0.214	11.53	0.053

Divergent Instability

Logdec	Iô/µB ⁴	U/n _p B
0.0223	0.119	4.24
x	0.214	4.38
8	0.321	4.60

Parapet Pld,

Edge Detail Y, d4 = 50mm

Model M1, B = 225mm,

Parapet P1d,

Angle +5°

Model M1, B = 225mm,

Edge Detail Y, d4 = 50mm

Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n ₂ d ₄	z/d ₄
0.01	0.518	9.50	0.0296
0.02	1.035	9.52	0.0278
0.03	1.553	9.23	0,0196
0.04	2.071	9.23	0.0186
0.05	2.589	9.45	0.0064
0.06	3.107	9.49	0.0052

0.0996 0.0832 0.0670 0.0508 0.0508 0.0508

9.85 9.67 9.05 9.04 8.79

0.518 1.035 1.553 2.071 2.589 3.107

0.02 0.03 0.06 0.03 0.06 0.03 0.06

2/d_6

U/n_zd₄

Mő/pB²

Logdec

Vortex Shedding in Bending

Vortex Shedding in Torsion

Logdec	$I\delta/\rho B^4$	U/ngd4	0(°)
0.0223	0.119	9.59	660'0
0.03	0.160	9.58	0.078
0.04	0.214	9.59	0.073

1.120 0.731 0.578

8.80 8.81 8.66

0.206 0.274 0.343

0.03

(..)0

U/ned4

18/pB4

Logdec

Vortex Shedding in Torsion

Divergent Instability

Torsional Instability

Logdec	15/µB ⁴	U/n _e B
0.0223	0.119	3.39
0.04	0.214	3.48
0.06	0.320	4.03

Divergent Instability

Galloping (Bending)

Logdec	Må/pB ²	U/nzd4
0.02	1.035	25.2
0.04	2.071	25.7
0.06	3.107	25.7

40

Angle 0°

Parapet P3d,

Edge Detail Y, $d_4 = 50mm$

Model M1, B = 225mm,

Parapet P2d,

Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _z d ₄	z/d4
0.01	0.518	8.21	0.0932
0.02	1.035	7.85	0.0302
0.03	1.553	7.45	0.0126
0.04	2.071	7.34	0.0064

Vortex Shedding in Torsion

ð(°)	3.222 0.968
U/ngd4	10.86 10.46
16/pB ⁴	0.137 0.206
Logdec	0.02

Divergent Instability

Galloping (Bending)

pB ² U/n _z d ₄	35 25.0	-	
Logdec Mő/pB	0.02 1.035	0.04 2.0	

Model M1, B = 225mm, Angle 0°

Edge Detail Y, $d_4 = 50mm$

Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _z d ₄	z/d4
0.01	0.518	7.12	0.0464
0.02	1.035	16.9	0.0232
0.03	1.553	6.89	0.0138
0.04	2.071	6.87	0.0090

Vortex Shedding in Torsion

-	_
(_)Ø	4.20
U/ngd4	10.82
Iő/pB ⁴	0.137
Logdec	0.02

Divergent Instability

Galloping (Bending)

Logdec	$M\delta/\rho B^2$	U/n _z d ₄
0.02	1.035	21.6
0.04	2.071	22.3
0.06	3.107	23.5

Edge Detail Y, d₄ = 50mm

B = 225mm,

Model M1,

Parapet P4d,

Angle 0°

Model M3, Edge Detail X, Parapet P2a, Angle 0" B = 450mm, d₄ = 50mm

1st Vortex Shedding in Bending

ogdec	Må/pB ²	U/n _z d ₄	z/d4
10	0.283	7.08	0.0194

2nd Vortex Shedding in Bending

0.0352 0.0194 0.0130 0.0092

6.58 6.78 6.78 6.75

0.518 1.035 1.553 2.071

0.01 0.02 0.03 0.03

z/d_a

U/nzd4

Mő/pB²

Logdec

Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n ₂ d ₄	z/d ₄
0.01	0.283	15.94	0.0636
0.02	0.565	15.92	0.0494
0.03	0.848	15.17	0.0424
0.04	1.130	15.19	0.0362
0.05	1.413	15.23	0.0284
0.06	1,696	15.19	0.0226

θ(°) 2.043

9.50

0.137

0.02

Divergent Instability

Galloping (Bending)

U/ngd4

I&/pB⁴

Logdec

Vortex Shedding in Torsion

Vortex Shedding in Torsion

Stable

U/nzd4

Må/øB²

Logdec

22.4 0

1.035 2.071 3.107

0.04 0.04 0.04

Divergent Instability

Torsional Instability

Logdec	lå/ρB ⁴	U/ngE
0.02	0.0569	4.90
0.04	0.1138	5.22
0.06	0.1707	5.35

42

Parapet P2b,	
Edge Detail X,	$d_4 = 50mm$
Model M3,	B = 450mm

Vortex Shedding in Bending

logdec	Mő/pB ²	U/n ₂ d ₄	z/d4
0.01	0.283	15.28	0.0654
0.02	0.565	14.52	0.0526
0.03	0.848	14.52	0.0448
0.04	1.130	14.54	0.0378
0.05	1.413	13.77	0:0308
90.06	1.696	13.72	0.0258

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

ogdec	$1\delta/\rho B^4$	U/n ₉ B
0.02	0.0569	2.90
0.04	0.1138	3.13
0.06	0.1707	3.70

Angle 0°

Model M3, Edge Detail X, Parapet P2b, Angle 0° B = 450mm, $d_4 = 50mm$

In Turbulent Flow

Vortex Shedding in Bending

Stable

Vortex Shedding in Torsion

Stable

Divergent Instability

gdec	Iδ/µB ⁴	U/n _g B
02	0.0569	3.47
0.04	0.1138	3.76
90	0.1707	4.12

Angle 0°

Model M3, B = 450mm,

Angle 0°

Parapet P3a,

Edge Detail X, d₄ = 50mm

Model M3, B = 450mm,

Parapet P3b, Edge Detail X, d4 = 50mm

1st Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n ₂ d4	*p/z
0.01	0.283	7.04	0.0102
0.02	0.565	6.67	0.0052
0.03	0.848	6.67	0.0032

0.0148 z/d4

U/n,d4 6.67

Mő/µB²

Logdec

0.283

0.01

1st Vortex Shedding in Bending

2nd Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _z d ₄	z/d4
10'0	0.283	15.26	0.0764
0.02	0.565	14.47	0.0604
0.03	0.848	14.44	0.0532
0.04	1.130	14.44	0.0452
0.05	1.413	14,44	0.0366
0.06	1.696	14.48	0.0300

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

Logdec	Iô/pB ⁴	U/n ₆ B
0.02	0.0569	2.22
0.04	0.1138	2.80
0.06	0.1707	3.01

	4	is.
0.0569	0.1138	0.1707
0.02	0.04	0.06
	0.0569	0.02 0.0569 4.55 0.04 0.1138 4.72

2nd Vortex Shedding in Bending

00000			
ogdec	Ma/pB-	U/nzd4	*p/z
0.01	0.283	15.31	0.0650
0.02	0.565	15.29	0.0512
0.03	0.848	14.52	0.0422
0.04	1.130	14.52	0.0370
0.05	1.413	14.52	0.0306
0.06	1.696	14.12	0.0252

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

44

Edge Detail X, d4 = 50mm Model M3, B = 450mm,

Angle 0° Parapet P2d,

1st Vortex Shedding in Bending

Logdec	Mő/pB ²	U/nzd4	z/d4
0.01	0.283	8.20	0.0132
0.02	0.565	7.42	0.0082
0.03	0.848	7.42	0.0062
0.04	1.130	7.45	0.0038

2nd Vortex Shedding in Bending

Logdec	$M\delta/\rho B^2$	U/n _z d ₄	₹p/z
0.01	0.283	22.36	0.1172
0.02	0.565	21.45	0960.0
0.03	0.848	20.72	0.0856
0.04	1.130	19.95	0.0750
0.05	1.413	66'61	0.0712
0.06	1.696	19.23	0.0618

Vortex Shedding in Torsion

Stable

Angle 0° Parapet P2a, Edge Detail Y, d₄ = 50mm Model M3, B = 450mm,

1st Vortex Shedding in Bending

z/d_	0.0162
U/n ₂ d ₄	7.11
Mő/pB ²	0.283
Logdec	10.0

2nd Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n ₂ d ₄	z/d ₄
0.01	0.283	14.13	0.0366
0.02	0.565	13.73	0.0228
0.03	0.848	12.92	0.0176
0.04	1.130	12.53	0.0094
0.05	1.413	12.15	0.0038

Vortex Shedding in Torsion

Logdec	$I\delta/\rho B^4$	U/ngd4	(°)
0.0138	0.0393	8.45	0.460
0.02	0.0569	8.18	0.280
0.03	0.0854	7.96	0.183

Divergent Instability

ogdec	Iô/µB ⁴	U/n ₉ B
0.02	0.0569	4.71
0.04	0.1138	4.92
0.06	0.1707	5.10

Edge Detail Y, d4 = 50mm Model M3, B = 450mm,

Parapet P2b,

Angle 0°

1st Vortex Shedding in Bending

z/d_	0.0104
U/n _z d ₄	6.46
Mő/pB ²	0.283
Logdec	0.01

2nd Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n _z d ₄	z/d4
10.0	0.283	14.72	0.0744
0.02	0.565	14.26	0.0520
0.03	0.848	13.89	0.0356
0.04	1.130	13.50	0.0184
0.05	1.413	13.23	0.0034

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

ogdec	Iδ/ρB ⁴	U/n ₆ B
0.02	0.0569	3.13
0.04	0.1138	3,48
0.06	0.1707	3.69

Angle 0°	
Parapet P3a,	
Edge Detail Y,	$d_4 = 50mm$
Model M3.	B = 450 mm,

Ist Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n _x d ₄	z/d4
0.01	0.283	6.98	0.0124
0.02	0.565	6.67	0.0088
0.03	0.848	6,65	0.0062

2nd Vortex Shedding in Bending

ogdec	Mő/pB ²	U/n _z d ₄	2/d4
0.01	0.283	13.66	0.0416
0.02	0.565	13.20	0.0226
0.03	0.848	12.42	0.0112
0.04	1.130	12.46	0.0022

Vortex Shedding in Torsion

$U/n_{\theta}d_{\alpha}$ $\theta(^{\circ})$	8.14 0.353
$I\delta/\rho B^4$	0.0393
Logdec	0.0138

Divergent Instability

SipB* U/n ₉ B		0.1707 5.52
Logdec I5/	0.04 0.1	

Model M3. Edge Detail Y. Parapet P3b, B = 450mm, d₄ = 50mm

Vortex Shedding in Bending

.ogdec	Mő/pB+	U/nzd4	P/Z
0.01	0.283	14.31	0.0932
0.02	0.565	14.31	0.0654
0.03	0.848	13.54	0.0492
0.04	1.130	13.54	0.0040

Vortex Shedding in Torsion

Stable

Divergent Instability

Torsional Instability

Logdec	$I\delta/\rho B^4$	U/n ₉ B
0.02	0.0569	2.85
101	0.1138	3.37
90.0	0.1707	3.89

Angle 0°

Ist Vortex Shedding in Bending

J/n _z d ₄ z/d ₄	8.19 0.0146 7.80 0.0080
Mő/pB ² U	0.283
Logdec	0.01

2nd Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n _x d ₄	z/d4
0.01	0.283	22.54	0.1160
0.02	0.565	21.42	0660.0
0.03	0.848	20.72	0.0868
0.04	1.130	19.90	0.0742
0.05	1.413	19.88	0.0696
0.06	1.696	19.13	0.0602

Vortex Shedding in Torsion

Stable

Angle 0"

1st Vortex Shedding in Bending

Logdec	M5/pB ²	U/n _z d ₄	*p/z
0.01	0.283	7.38	0.0156

2nd Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _x d ₄	z/d_
0.01	0.283	19.80	0.1500
0.02	0.565	18.98	0.1304
0.03	0.848	18.31	0.1170
0.04	1.130	18.30	0.1090
0.05	1.413	17.46	0.1026
0.06	1.696	17.47	0.0958

Vortex Shedding in Torsion

Logdec	Iõ/pB ⁴	U/ngd4	(°)0
0.0138	0.0393	9.15	0.596
0.02	0.0569	8.65	0.357
0.03	0.0854	8.42	0.250

Model M3, Edge Detail Y, Parapet P4d, Angle 0* B = 450mm, d₄ = 50mm

1st Vortex Shedding in Bending

z/d4	0.0156
U/n _z d ₄	7.02 6.99
Mő/µB ²	0.283
Logdec	0.01

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2nd Vortex Shedding in Bending

Logdec	Mő/pB ²	U/nzd4	z/d_
0.04	1.130	18.08	0.1962
0.05	1.413	18.12	0.1756
0.06	1.696	17.49	0.1590

Vortex Shedding in Torsion

Logdec	Iő/pB ⁴	U/n ₆ d ₄	(°)0
0.0138	0.0393	7.64	0.338

Model MD, Edge Detail X, Parapet F2D, Angle V⁻¹ B = 675mm, d₄ = 50mm

Vortex Shedding in Bending

ogdec	Mã/pB ²	U/n ₂ d ₄	z/d4
0.0151	0.2441	10.51	0.0210
0.02	0.3234	10.57	0.0186
0.03	0.4850	10.59	0:0150
0.04	0.6467	10.59	0.0126
0.05	0.8084	10.61	0.0106
0.06	10/6/0	10.60	0600/0

1st Vortex Shedding in Torsion

ogdec	Iő/µB ⁴	U/ngd4	θ(*)
0.0176	0.0153	B.50	0.196
0.02	0.0174	8.51	0.171
0.03	0.0261	8,44	0.115
101	0.0347	8.44	0.078
200	0.0434	8.46	0.047

2nd Vortex Shedding in Torsion

ogdec	Iő/µB ⁴	U/ngd4	θ(")
0.0176	0.0153	14.12	0.354
0.02	0.0174	14.11	0.337
50.0	0.0261	13.38	0.251
10.0	0.0347	13.38	0.184
50.0	0.0434	13.40	0.119
90.0	0.0521	12.76	0.041

Divergent Instability

Torsional Instability

gdec	Iå/pB ⁴	U/n _g B
0.02	0.0174	3.28
5	0.0347	3.32
907	0.0521	3.42

 Edge Detail X, Parapet P3a, Angle 0^e 	$d_4 = 50mm$
Model M5,	B = 675mm,

Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _z d ₄	z/d4
0.0151	0.2441	11.00	0.0158
0.02	0.3234	10.98	0.0140
0.03	0.4850	11.02	0.0110
0.04	0.6467	10.55	0,0092
0.05	0.8084	11.01	0.0080
0.06	1079.0	11.05	0.0068

Vortex Shedding in Torsion

Logdec	$1\delta/\rho B^4$	U/ned4	(_) (
0.0176	0.0153	13.86	0.289
0.02	0.0174	13.86	0.268
0.03	0.0261	13.77	0.208
0.04	0.0347	13.78	0.153
0.05	0.0434	13.78	0.099
0.06	0.0521	13.79	0.049

Divergent Instability

ogdec	$15/\rho B^{4}$	U/n _g B
0.02	0.0174	3.82
0.04	0.0347	3.92
0.06	0.0521	3.96

Model M5, Edge Detail X, Parapet P3b, B = 675mm, $d_4 = 50mm$

Angle 0°

Vortex Shedding in Bending

Logdec	M5/pB ²	U/n _s d ₄	z/d4
0.0151	0.2441	10.51	0.0204
0.02	0.3234	10.51	0.0170
0.03	0.4850	10.51	0.0112
0.04	0.6467	10.49	0.0056

Vortex Shedding in Torsion

ogdec	$I_0/\rho B^4$	U/ngd4	θ(°)
0.0176	0.0153	13.83	0.469
0.02	0.0174	13.77	0.441
0.03	0.0261	13.79	0.340
0.04	0.0347	13.47	0.280
0.05	0.0434	13.18	0.217
0.06	0.0521	13.18	0.151

Divergent Instability

Torstonal Instability

Logdec	lã/µB⁴	U/n _g B
0.02	0.0174	2.86
0.04	0.0347	2.95
0.06	0.0521	3.03

Model M5, Edge Detail Y, Parapet P2b, Angle 0° B = 675mm, $d_4 = 50mm$

Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n ₂ d4	z/d_
0.0151	0.2441	10.03	0.0126
0.02	0.3234	10.03	0.0084

Vortex Shedding in Torsion

Stable

Divergent Instability

ogdec	Iő/µB ⁴	U/n ₆ B
0.02	0.0174	3.54
0.04	0.0347	3.61
0.06	0.0521	3.68

Angle 0°

Vortex Shedding in Bending

Logdec	$M\delta/\rho B^2$	U/nzd4	₽p/z
0.0151	0.2441	9.22	0.0136
0.02	0.3234	9.60	0.0126
0.03	0.4850	9.57	0.0060

Vortex Shedding in Torsion

Logdec	$I\delta/\rho B^4$	U/ngd4	θ(")
0.0176	0.0153	7,44	0.142
0.02	0.0174	7.47	0.127
0.03	0.0261	7.49	0.088
0.04	0.0347	7.44	0.063

Divergent Instability

Torsional Instability

Logdec	Iå/ρB⁴	U/n ₉ B
0.02	0.0174	4.30
0.04	0.0347	4.38
0.06	0.0521	4.43

Angle 0°	
Parapet P3b,	
Edge Detail Y,	$d_4 = 50mm$
Model M5,	B = 675mm,

Vortex Shedding in Bending

Stable

Vortex Shedding in Torsion

θ(°)	0.037
U/n ₀ d ₄	13.56
Ið/pB ⁴	0.0153
Logdec	0.0176

Divergent Instability

Logdec	$I\delta/\rho B^4$	U/n ₉ B
0.02	0.0174	3.18
0.04	0.0347	3.27
0.06	0.0521	3.43

Parapet P3b Overhang $Ud_4 = 0.25$ Edge Detail $k/d_4 = 0.3$ (Y type)

Model MI

Angle 0°

B = 150mm d_{*} = 50mm

1st Vortex Shedding in Bending

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z/d ₄	0.0027
U/n,d.	4,48
M8/pB1	0.957
Logdec	0.01

2nd Vortex Shedding in Bending

Logdec	MővµB ²	U/n,d,	*P/z
0.01	0.957	12.92	0.0028
0.02	1.914	12.92	0.0023
0.03	2.870	12.92	0.0025
0.04	3.827	12.92	0.0022
0.05	4.784	12.92	0.0020
0.06	5.741	12.92	0.0017

Divergent Instability

Galloping

	U/n _s d _s	18.75	61.47	75.74
	MðVµB [‡]	\$16.1	3.827	5.741
1442.312.414	Logdec	0.02	0.04	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Model M1	-00-	Paramat P7s	Anda 0*	
ATTRACTOR ATTAC	2	and a second second	of the second second	
th - 160mm				

B = 150mm $d_4 = 54.6mm$

1st Vortex Shedding in Bending

z/d_	0.0064	0.0022
U/n,d,	4.69	15.4
Må/pB ²	1.169	2.338
Logdet	0.01	0.02

2nd Vortex Shedding in Bending

Logdec	M5V.pB1	Um_d,	zid.
0.01	1.169	10.13	0.0127
0.02	2.338	10.59	0,005
0.03	3,508	10.51	0.0087
0.04	4.677	10.89	0.0061
0.05	5.846	10.89	0,0053
0.06	7.015	11.27	0.0031

mations	taken	1.1.
0.243	10.08	0.06
0.317	18'6	0.059
0.476	9.81	0.024

Divergent Instability

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Angle 0°	
Parapet P2a	
Overhang 1/d _a = 0.25	Edge Detail k/d, = 0.5 (Y-type)
Model M1	

B = 150mm $d_{\star} = 50mm$

1st Vortex Shedding in Bending

z/d,	0.0153	0.0027	
U/n,d.	5.65	4.91	
Mã/pB ²	0.947	1.894	
Logdec	0.01	0.02	

2nd Vortex Shedding in Bending

z/d_	0.0203
U/n,d,	8.77
Mő/pB2	0.947
Logdec	0.01

1

Vortex Shedding in Torsion

_	15/µB*	U/n,d,	(°)8
	0.302	68.6	0.174
	0.403	9.83	0.159
	0.605	10.01	0.088
	0.807	10.01	0.086
	1.008	10.25	0.048
	1.210	10.34	0.037

Divergent Instability

Gallophug

U/n ^s d ₄	17.46	19/21	94.71
M5/pB ²	1,894	3.789	5.683
Lagdec	0.02	0.04	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Parapet P3b	
Overhang 1/d ₄ = 0.25	Edge Detail $k/d_4 = 0.5$ (Y-type)
Model M1	

Angle 0°

B= 150mm d₄ = 50mm

Ist Vortex Shedding in Bending

*p/z	0.0028
U/n,d,	4.50
Mő/pB ¹	0.963
Logdec	10.0

2nd Vortex Shedding in Bending

1

cogdec	Mő/pB ¹	U/n,d.	z/d.
0.01	0.963	12.96	0.0035
0.02	1.926	12.74	0.0030
0.03	2.889	13.04	0.0029
0.04	3.852	12.96	0.0028
0.05	4.815	13,338	0.0023
0.06	5.777	13.338	0.002

Divergent Instability

Mő/pB [‡] U/n _c d _a		3.852 58.45	
Logdec	0.02	0.04	

Model M1 Overhang 1/d₄ = 0.50 Parapet P2a Edge Detail k/d₄ = 0.5 (Y-type)

Angle 0°

B= 175mm d₄ = S0mm Vortex Shedding in Bending

z/d4	0.0138
U/n,d,	10.8
M8/pB ²	0.720
Logdec	0.01

Vortex Shedding in Torsion

	Id/pB*	U/n _s d _a	(a)0
	0.165	39.65	0.221
	0.300	9.82	0,166
	0.449	9.82	0.123
	0.599	9.98	0.078
5	0.749	9.98	0.073
	0.899	9.98	0.047

Divergent Instability

Galloping

U/a,d.	22.16	22.54	23.07
M5/pB2	1.44	2.88	4.32
Logdec	0.02	- 0.04	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Model M1 U° Parapec P2a Angle 0°

B= 175mm d₄ = 54.6mm Ist Vortex Shedding in Bending

z/d4	0.0057
U/n,d,	6.12
Mã/gB²	0.925
Logdec	0.01

2nd Vortex Shedding in Bending

Logdec	Må/pB ²	Um,d.	*DIZ
10.0	0.925	9.49	0.0162
0.02	1.850	10,12	1200.0
0.03	2.774	10.51	0.0047
0.04	3.699	10.83	0.0027
0.05	4.624	10.83	0.0025
0.06	5.549	10.83	0.0021

Vortex Shedding in Torsion

Logdec	15/µB*	U/med4	6(~)
0.0146	0.189	10.33	0.176
0.02	0.259	10.56	0.144
0.03	0.385	10.33	0.048

Divergent Instability

	Mă/pB2 1.850 3.699
0.06	5.549

Model M1 Overhang I/d₄ = 0.50 Parapet P3a Edge Detail k/d₄ = 0.5 (Y-type) B= 175mm

Angle 0°

d, = 50mm

Vortex Shedding in Bending

ordec	MåfæB ²	U/n.d.	z/d.
	- and one of	and the second se	T
	0.725	11.07	0.0070
	1,450	11.45	0.0041
	2,175	11.45	0.0034
	2,900	11.45	0.0030
	3.625	11.53	0.0023
	4.349	11.45	0.0015

Vortex Shedding in Torsion

Logdec	IA/µB*	U/n _p d ₄	6(*)
7610.0	0.227	11.16	0.157
0.03	0.346	11.40	0:090
0.04	0.461	11.40	0.075
0.05	0.576	11.40	0.064
0.06	0.691	11.59	0.036

Galloping

U/n,d,	23.00	23.45	23.98
Må/pB ²	1.450	2.900	4,349
Logdec	0.02	0.04	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Model MI U' Parapet P3a Angle 0"

B = 175mm $d_4 = 54.6mm$ Vortex Shedding in Bending

0.0054	0.0047	0.0031	0.0026	0.0021	0.0020
11.98	86.11	12.29	12.29	11.9	11.9
0.933	1.866	2.799	3.732	4.666	5.599
10.0	0.02	0.03	0.04	0.05	0.03
	0.933 11.98	0.933 11.98 1.866 11.98	0.933 11.98 1.866 11.98 2.799 12.29	0.933 11.98 1.866 11.98 2.799 12.29 3.732 12.29	0.933 1.866 2.799 3.732 4.666

Vortex Shedding in Torsion

Logdec	Iő/pB*	U/n,d.	(_)0
0.0145	0.191	10.96	060'0
0.02	0.263	10.77	0.039

Divergent Instability

U/n,d,	20.03	20.80	22.21
Mő/pB ²	1.866	3.732	5.599
Logdec	0.02	0.04	0.06

Angle 0" Overhang Ud, = 0.25 Parapet P2a Edge detail k/d, = 0.30 (Y-type) Model M3

B= 375mm d, = 50m 1st Vortex Shedding in Bending

2nd Vortex Shedding in Bending

ogdec	M8/pB ²	U/n,d,	z/d4
101	865.0	16.83	0.0566
0.02	0.795	16.25	0.0426
0.03	1193	15.45	0.0319
0.04	1.590	15.45	0.0263
0.05	1.988	14.87	0.0227
0.06	2.385	14.36	0.0171

Vortex Shedding in Torsion

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		_
6(*)	0.451	0.207
Ultada	8.55	8.19
15/pB*	0.0399	0.0799
Logdec	0.01	0.02
_	-	_

Divergent Instability

Torstonal Instability

U/n,B	3.70	3.97	4,22
Iõ/pB*	0.0815	0.1630	0.2445
Logdec	0.02	0.04	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Angle 0° Parapet P2a Overhang $l/d_4 = 0.25$ Edge detail $k/d_4 = 0.50$ (Y-type) B= 375mm d, = 50mm Model M3

1st Vortex Shedding in Bending

Logdec	Mő/pB ²	Um,d.	Th/Z
107	0,398	7.18	0.0192

2nd Vortex Shedding in Bending

Logdec	M5/µB [±]	U/n,d,	z/d.
0.01	0.398	17.52	0.0652
0.02	0.795	16.65	0.0511
0.03	1.193	16.07	0,0413
0.04	1.590	15.79	0.0349
0.05	1.988	15.51	0.0299
0.06	2.385	15.51	0.0249

Logdec	10/01	U/II/da	(.)0
0.01	0.0399	8.55	0.508
0.02	0.0799	8.55	0.198
0.03	0,1198	9.24	0.061

Divergent Instability

Torrional Instability

Logdec	15/pB ⁴	U/n ₉ B
	0.0815	3.56
	0.1630	3.81
	0.2445	4.04

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Angle 0° Parapet Pla ъ Model M3

B= 375mm d₄ = 54.6mm

1st Vortex Shedding in Bending

Logdec	Må/pB ²	U/n,d,	2/da
0.01	0.394	6.18	0.0142
0.02	0.788	6.18	0.008
0.03	1.182	5.71	0.003
0.04	1.576	5.71	0.000

9 P ADD VOLUCX

z/d_	0.0774	0.0657	0.0554	0.0492	0.0450	0.0407
U/n,d,	15.76	15.23	14.71	14.45	14.24	14.24
Mő/µB [±]	0.394	0.788	1.182	1.576	0961	2.363
Logdec	0.01	0.02	0.03	0.04	0,05	0.06

Vortex Shedding in Torsion

Q(_)0	0.442	0.216	
Ulnud.	7.87	8.28	
15/pB*	0.0387	0.0774	
Logdec	0.01	0.02	

Divergent Instability

Torritonal Instability

ogdec	18/µB ⁴	U/n/B
0,02	0.0774	3.04
0.04	0.1547	3.53
907	0.2321	3.78

PLATE GIRDER BRIDGES - FURTHER WORK

Angle 0* Parapet P2b Overhang $I/d_4 = 0.25$ Pa Edge detail $k/d_a = 0.50$ (Y-type) Model M3

B=375mm $d_4=50m$

1st Vortex Shedding in Bending

Logdec	Må/øB ²	U/n,d,	z/d4
	0.398	7.24	0.0340
	0.795	10'1	0.0181
	1.193	6.78	0.0112
	1.590	6.55	0.0073
	1.988	6.26	0.0036

2nd Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n,d.	z/d_
0.01	0.398	16.14	0.0898
0.02	0.795	15,62	0.0706
0.03	1.193	14.76	0.0582
0.04	1.590	14.53	0.0503
0.05	1.988	14.24	0.0451
0.06	2.385	13.96	0.0398

Vortex Shedding in Torsion

Logdec	10/pB*	U/n ₄ d ₄	(.)0
10.0	0.0399	8.78	1.190
0.02	0.0799	8.59	0.730
0.03	0.1198	8.60	0.299

Model M3 Overhang 1/d₄ = 0.25 Parapet P2b Angle 0* Edge detail k/d₄ = 0.50 (Y-type) B= 375mm

 $d_s = 50m$

Divergent Instability

Torstowal Instability

15/µB* U/n _p B		0.1675 3.77	0.2513 4.10
Logdec	0.02	0.04	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Ist Vortex Shedding in Bending

Logdec	Mő/pB ²	U/n,d,	z/d,
0.01	0.342	6.99	0.0141
0.02	0.685	6:39	0.0071
0.03	1.027	6.39	0.0018

2nd Vortex Shedding in Bending

Logdec	Mő/pB ²	U/B/G*	There	
0.01	0.342	16.15	0.0442	
0.02	0.685	15.58	0.0337	
0.03	1.027	15.01	0.0266	
0.04	1.370	15.01	0.0217	
0.05	1.712	14,44	0.0176	
0.06	2.054	13.99	0.0138	

Logdec	15/pB*	U/n,d.	(a)0
0.01	0.029	8.95	0.468
0.02	0.057	8.95	0.353
0.03	0.086	8.69	0.265

Divergent Instability

Logdec	15/µB*	U/n,B
0.02	0.080	4,00
0.04	0.160	4.22
0.06	0.239	4.52

1st Vortex Shedding in Bending

Logdec	Må/pB ²	U/n,d.	z/d_	
0.01	0.342	7.05	0.0133	
				ĺ

2nd Vortex Shedding in Bending

Logdec	Må/pB ²	U/n_d_	z/d_	
0.01	0.342	17.11	0.0488	
0.02	0.685	16.49	0.0395	
0.03	1.027	16.09	0.0320	
0.04	1.370	16.03	0.0262	
0.05	1.712	15.41	0.0236	
0.06	2.054	15.18	0.0195	

Vortex Shedding in Torsion

Logdec	15/µB*	U/n _e d _a	Ø(*)
0.01	0.029	8.95	0.440

Divergent Instability

Torstowal Instability

Logdec	15/µB*	U/n,B	
0.02	0.080	3.49	
0.04	0.160	3.77	
0.06	0.239	3.99	

PLATE GIRDER BRIDGES - FURTHER WORK

Ist Vortex Shedding in Bending

Logdec	Må/pB ²	U/n,d,	z/d4
0.01	0.342	7.16	0.0061

2nd Vortex Shedding in Bending

Logdec	MűVpB [±]	U/n,d.	z/d.
0.01	0.342	17.68	0.0390
0.02	0.685	17.23	0.0310
0.03	1.027	17.06	0.0247
0.04	1.370	16.54	0.0212
0.05	1.712	16.49	0.0172
90'0	2.054	15.97	0.0142

Vortex Shedding in Torsion

15/pB* U//q.d. 6(")	0.029 8.63 0.551
Lopdec 1	0.01 0

ł

Divergent Instability

U/hgB	3.72	3.98	4 22
LBV/pB ⁴	0.080	0.160	0.239
Logdec	0.02	10.04	

Angle 0°

Ist Vortex Shedding in Bending

gdec	Må/pB ³	U/n,d,	z/d_
0.01	0.346	6.27	0.00119
	0.692	5.86	0.0042

2nd Vortex Shedding in Bending

Logdec	Mű/pB ²	U/n,d,	z/d_	
0.01	0.342	15.22	0.0715	
0.02	0.692	15.01	0.0575	
0.03	1.037	14,49	0.0501	
0.04	1.383	14.02	0.0434	
0.05	1.729	14.02	0.0384	
0.06	2.075	13.55	0.0336	

Vortex Shedding in Torsion

STABLE

Divergent Instability

Torsional Instability

U/n,B	3.82 4.11 4.34
15/pB ⁴	0.059 0.118 0.176
Logdec	0.02 0.04 0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Ist Vortex Shedding in Bending

	opdec	Mő/pB ²	U/n,d,	z/d4	
0.33	101	0.346	6.33	0.0120	

2nd Vortex Shedding in Bending

Logdec	Mő/µB ²	DOM: NO	Trace
0.01	0.346	14.49	0.0583
0.02	0.692	14.02	0.0488
0.03	1.037	13.55	0.0409
0.04	1.383	13.55	0.0347
0.05	1.729	13.55	0.0288
0.06	2.075	13.08	0.0256

STABLE

Divergent Instability

Ist Vortex Shedding in Bending

z/d"	0.0259 0.0149 0.0099
U/n,d4	7.56 6.99 7.05
M5/pB ²	0.342 0.685 1.027
Logdec	0.01 0.02 0.03

2nd Vortex Shedding in Bending

Logdec	M5/pB2	U/n,d,	2/d_4	
0.01	0.342	16.32	0.0954	
0.02	0.685	16.32	0.0785	
0.03	1.027	15.58	0.0657	
0.04	1.370	15.23	0.0576	
0.05	1.712	15.17	0.0492	
0.06	2.054	14.78	0.0438	

Vortex Shedding in Torsion

Logdec	LiveB*	U/m,d_a	0(°)
0.01	0.029	9.07	0.963
0.02	0.057	8.84	0.616
0.03	0.086	8.87	0.450
0.04	0.114	8.72	0.279

Divergent Instability

Torsional Instability

Iô(pB*	0.057	0.114	0.171
ogdec	0.02	 40.	0.06

PLATE GIRDER BRIDGES - FURTHER WORK

1st Vortex Shedding in Bending

Logdec	M5/pB ²	U/n,d.	z/d4	
0.01	0,342	7.56	0.0292	
0.02	0.685	6.82	0.0138	
0.03	1.027	6.59	0.0053	

2nd Vortex Shedding in Bending

Logdec	Mő/µB ²	U/n _s d ₄	2/d4
0.01	0.342	15.92	0.0897
0.02	0.685	15,41	0.0731
0.03	1.027	14.89	0.0615
0.04	1.370	14.84	0.0545
0.05	1.712	14.38	0.0461
0.06	2.054	14.33	0.0412

Vortex Shedding in Torsion

Logdec	18/µB ⁴	U/mda	(°.)
0.01	0.029	9.16	1.151
0.02	0.057	9.01	0.737
0.03	0.086	8.87	0.462

Divergent Instability

Torsional Instability

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Logdec	IdvpB4	U/n ₄ B
0.02	0.057	3.16
0.04	0.114	3.57
0.06	0.171	3.78

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Vortex Shedding in Bending

STABLE

1st Vortex Shedding in Torsion

θ.e	0.283 0.066 0.004
U/n _p d _a	7.86 7.86 8.19
16/pB*	0.028 0.054 0.080
Logdec	0.0104 0.02 0.03

2nd Vortex Shedding in Torsion

600	0.019
U/myd4	14.10
16/pB*	0.028
Logdec	0.0104

Divergent Instability

Tarsional Instability

U/n,B	4.46 4.69 4.95
Iô/pB*	0.107 0.167 0.161
Logdec	0.02 0.04 0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Vortex Shedding in Bending

2/d ₄	0.0017	
U/n,d_	80.6	
Mő/pB ²	0.277	
Logdec	0.01	

Vortex Shedding in Torsion

Logdec	Iå/øB⁴	U/n,d,	644	-
0.0104	0.028 0.054	8.14 7.78	0.225 0.022	-

Divergent Instability

1

gdec	15/µB ⁴	U/n ₄ B	
2	0.054	4.90	
0.04	0.107	5.15	
9	0.161	5.32	

Vortex Shedding in Bending

Logdoc	Mő/pB ²	U/n _s d _a	z/d,	
0.01	0.277	9.25	0.0076	
0.02	0.544	9.31	0.0042	
0.03	0.831	9.31	0.0025	
0.04	1.108	9.31	0.0016	
0.05	1.384	9.37	0.0010	

1st Vortex Shedding in Torsion

944	0.007
U/n,d4	7.78 7.45
15/pB ⁴	0.028 0.054
Logdec	0.0104 0.02

2nd Vortex Shedding in Torsion

Logdec	IA/µB*	U/n/d,	B10	
0.0104	0.028	13.81	0.249	
0.02	0.054	14,10	0.014	

Divergent Instability

Torsional Instability

Logdec	16/pB+	U/neB
0.02	0.054	4.52
0.04	0.107	4.77
0.06	0.161	4.94

PLATE GIRDER BRIDGES - FURTHER WORK

Model M3 U Parapet P2a Angle 0" B = 450mm d4 = 54.6mm

Vortex Shedding in Bending

Logdec	M8/pB ²	U/n ₄ d ₄	z/d4	
0.01	0.300	14.13	0.0511	
0.02	0.599	13.59	0.0420	
0.03	0.899	13.09	0.0327	
0.04	1.199	13.09	0.0268	
0.05	1.498	12.66	0.0234	
0.06	1.798	12.60	0.0201	

1st Vortex Shedding in Torsion

Logdec	18/µB*	UInda	940	
0.01	0.023	7.83	0.296	
0.02	0.047	8.12	0.139	
0.03	0.070	8.40	0.037	
0.04	0.094	8.40	0.009	

2nd Vortex Shedding in Torsion

600	0.123 0.033
U/n _p d.	15.01 14.73
15/pB4	0.023 0.047
Logdec	0.01 0.02

Divergent Instability

Torstonal Instability

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optec	IS/pB ⁴	U/n _e B
.02	0.047	3,65
10	0.094	3.89
901	0.141	4.10

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Vortes Shedding in Bending

STABLE

Vortex Shedding in Torsion

Logdec	Ib/pB*	U/m,d,	Beet
0.01	0.023	6.93	0.122

Divergent Instability

Torsional Instability

U/naB	4.55 4,86 5.15
18/øB ⁴	0.047 0.094 0.141
Logdec	0.02 0.04 0.06

PLATE GIRDER BRIDGES - FURTHER WORK

Repeat Tests

1st Vortex Shedding in Bending

z/d ₄	0.0143	
U/n _r d _s	7,41	
M6/pB ²	0.283	
Logdec	0.0102	

2nd Vortex Shedding in Bending

Logdec	Må/pB ²	U/n_d4	z/d ₄	-
0.0102	0.283	14.37	0.0361	-

Divergent Instability

F

Iă/pB ⁴ U 0.057 5	/n ₀ B	28
4	n	5.
	Lö/pB ⁴	0

Model M2	Edge Detail $k = 0.4 d_4$	Angle O ^o
Aire Valley Viaduct		

B = 275mm d₄ = 50mm

Vortex Shedding in Bending

STABLE

Vortex Shedding in Torsion

Logdec	15/pB4	U/m/d4	θ
0.0193	0.048	8.836	0.064
0.03	0.075	8.836	0.028
0.04	0.150	8.876	0.010

Divergent Instability

Torsional Instability

18/pB* U/npB		0.100 5.18	
Logdec	0.02	0.04	0.06

Model M2 Edge Detail k = 0.4 d₄ Angle -5* Vortex Shedding in Bending

STABLE

Vortex Shedding in Torsion

STABLE

Divergent Instability

Logdec	16/pB ⁴	U/n,B
0.02	0.050	10.11
0.04	0.100	10.18
0:00	0.150	10.39

Vortex Shedding in Bending

Logdec	M6/µB ²	U/n,d4	z/d.
0.0135	0.461	14.73	0.0823
0.02	0.683	14.32	0.0706
0.03	1.025	13.91	0.0591
0.04	1.367	13.58	0.0493
0.05	1.708	13.18	0.0421
0.06	2.050	13.09	0.0367

Vortex Shedding in Torsion

STABLE

Divergent Instability

Logdec	L6/pB*	U/n _y B
0.02	0.050	2.96
0.04	0.100	3.44
0.06	0.150	3.74

EP ref. = Edge / Parapet reference										
Edge	K/d4	Parapet	h/d_{α}	suffix	ф					
х	0.05	P1	0.3	a	1/					
	U.N.O	P2	0.7	b	1/2					
Y	0.20	P3	1.1	c	3/4					
	U.N.O	P4	1.5	d	1					

Model	Units	1 da	b^{\ast}/d_{4}	bid4	$(b/d_4)^{1.5}$	$m/\rho b^2$	K _R
мі •	1	۰ <i>\</i> ‡	3	3 1/2	6.5	471 (534U)	0.43
M2	1	1 1/2	3	5 1/2	12.9	0.307	0.37
M3 🕤	2	124	$7^{-1}l_{2}$	9	27.0		0.29
M4 🕤	2	1 1/2	7 1/2	10	31.6		0.27
M5 🕤	3	1	12	13 1/2	49.6		0.24
M6 🕞	3	1 1/2	12	14 1/2	55.2		0.23

Further Tests Results v. Current Rules - Vertical Bending - Table M1F1

NOTE :

GENERAL NOTES	0.87	PARAPETS	MODEL REF.	
reduction for rough flow,	$K_R = \frac{3.07}{\sqrt{3} \frac{V_1}{V_2}} = 0.465 \ (0.486 \ \text{for U'})$	P2a, P3a	Reduced overhang	M1

mode	EP Ref.	Edge d	$\delta_{\rm S}$	2	$\frac{m\delta_s}{\rho h^2}$	$\frac{V}{f_0 d_4}$		easured redicted	$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_R y_{max}}{10^3 d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10}{45} \left(\frac{d_4}{b}\right)^{1.5}$		easured edicted
-		M					rule ¹	proposed ²				rule ³	proposed
	YP		0.01					1000000000					
	YP		0.02										
	YP		0.03										
	YP		0.04			- C							
	YP		0.05										
_	YP		0.06			cb	_					200	
1st	YP 2a		0.01	0	0.720	8.01	1.23		.0138	6.4	53.1	.12	.06
	YP	S	0.02	0									0.022
	YP	= 0.5 (Y)	0.03	0					I I				
	YP .	1,	0.04	0									
	YP	r.	0.05	0		- B							
	YP		0.06	0								S	
1st	0.000	-	0.01	0	0.720	11.07	1,70	1	.0070	3.3	\$3,1	.06	.03
1st	YP 3a	= 0.5 (Y)	0.02	0	1.450	11.45	1.76		.0041	1.9	26.5	.07	.03
1st	YP 3a	5.0	0,03	0	2.175	11.45	1.76		.0034	1.6	17.7	.09	.04
1st	YP 3a	1,	0.04	0	2.900	11.45	1.76		.0030	1.4	13.3	.10	.04
1st	YP 3a	"Pi	0.05	0	3.625	11.53	1.77		.0023	1.1	10.6	.10	.04
1st	YP 3a		0.06	0	4.349	11.45	1.76		.0015	0.7	8.8	.08	.03
	YP		0,01		1000								
	YP		0.02										
	YP		0.03								+		
	YP		0.04						1				
	YP		0.05					S					-
	UP* 2a	U'	0.01	0	0.925	6.12	0.94		.0057	2.8	46,8	.06	.03
Ind	U'P 2a		0.01	0	0.925	9.49	1.46	2	.0162	7.9	46.8	.17	.08
Ind	UP 2a	-	0.02	0	1.850	10.12	1.56		,0071	3.5	23.4	.15	.07
Ind		in a	0.03	0	2.774	10.51	1.62		.0047	2.3	15.6	.15	.07
Ind	U'P 2a	Edge	0.04	0	3.699	10.83	1.67		.0027	1.3	11.7	.11	.05
2nd	U'P 2a	-	0.05	0	4.624	10.83	1.67		.0025	1.2	9.4	.13	.06
Ind	U'P 2a		0.06	0	5.549	10.83	1.67		.0021	1.0	7.8	.13	.06
1st	UPP 3a		0.01	0	0.933	11.98	1.84		.0054	2.6	46.8	.06	.03
1st	U'P 3a	5	0.02	0	1.866	11.98	1.84		.0047	2.3	23,4	.10	.04
1st	U'P 3a	a.	0.03	0	2.799	12.29	1.89		.0031	1.5	15.6	.10	.04
1st	U'P 3a	Edge	0.04	0	3.732	12.29	1.89		.0026	1.3	11.7	.11	.05
1st	UPP 3a		0.05	0	4.666	11.90	1.83		.0021	1.0	9.4	.11	.05
1st	U'P 3a		0.06	0	5,599	11.90	1.83		.0020	1.0	7.8	.12	.05

¹ c.f. 6.5 ² still 6.5 3 without 3.0 factor

4 with c factor

Further Test Results v. Current Rules - Galloping - Table M1F1

T'L	ref. $=$ 1	Edge / P	arape	a refere	nce
Edge	K/d4	Parapet	h/d_{α}	suffix	ġ
Х	0.05	Pl	0.3	a	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3_	1.1	c	3/4
	U.N.O	P4	1.5	d	1

Mo	del	Units	Id,	$b^{\phi} d_{\phi}$	h/d ₄	(b/d) ¹³	(b/d) ^{2.5}
MI		1	1 1/2	3	3 1/1	12 ¹ / ₂	882 (955 U*)
342	0	1	1%	3	5 1/3	12.9	
М3	0	2	1	7 1/2	9	27.0	
M4	0	2	1.1_{2}	7 1/2	10	31.6	
M5	0	3	1	12	13 ¹ / ₂	49.5	
M6	0	3	1 1/2	12	14 1/2	55.2	

NOTE :

GENER	AL N	OTES	Cg -> 1.0 Galloping						3a	MODEL REF. Reduced overhang		M1
EP Ref.	Edge P	$\delta_{\rm S}$	Z	$\frac{m\delta_{5}}{\rho b^{2}}$	$\frac{V_g}{f_b d_4} =$	$S_0 = \left(\frac{m}{c^{b^2}}\right) \delta_s \left(\frac{b}{d_s}\right)^2$	measured predicted			4) C 1	4	ŝ.
YP YP YP YP YP YP		0.01 0.02 0.03 0.04 0.05 0.06										
YP YP 2a YP YP 2a YP 2a	$b(d_i) = 0.5 (V)$	0.01 0.02 0.03 0.04 0.05	0 0 0 0 0 0	1.44 2.88	22.16 22.54	17.64 35.28	1.26 .64					
YP 2a YP YP 3a YP YP 3a	-0.5(Y) I	0.06 0.01 0.02 0.03 0.04	0 0 0	4.32 1.45 2.90	23.06 23.00 23.45	52.92 17.64 35.28	.44 1.30 .66			분원		
YP YP 3a YP	kd.	0.05 0.06 0.01	0	435	23.98	52.92	.45			-		-
ҮР ҮР ҮР ҮР ҮР		0.02 0.03 0.04 0.05 0.06										
P P 2a P P 2a	Edge U'	0.01 0.02 0.03 0.04	0 0 0	1.85	20.06	19.1 38.2	1.05				-	
P P 2a	Ed	0.05 0.06	0 0	5,55	21.95	57.3	.38					
P P 3a P	Edge U'	0.01 0.02 0.03	00000	1.87	21.88	19.1	1.15					
Р 3a Р Р 3a	Ed	0.04 0.05 0.06	0000	3.73 5.60	22.72	38.2 57.3	.59					

Futher Test Results v.	Current Rules - Torsion - Table M1F1

EP r	ef. = E	dge / P	arape	t refere	ence
Edge	K/da	Parapet	h'd ₄	suffix	ф.
Х	0.05	P1	0.3	a	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
	U.N.O	P4	1.5	d	1

Mod	el	Units	I'd4	$b^{\ast} d_{s}$	b/d ₄	$\left(b/d_{s}\right)^{1.5}$	$m/\rho b^2$	$K_{\rm R}$
мі	•	1	η.	3	3 1/2	\$0	1203 922 766 U	0.43
M2	0	1	1^{1}_{2}	3	5 1/2	390		0.37
М3	0	2	24	7 1/2	9	2187		0.29
M4	0	2	1 1/2	7 1/2	10	3162		0.27
M\$	0	3	1	12	$13^{-1}\!/_2$	9040		0.24
M6	0	3	1 4	12	14 1/2	11609		0.23

GENER.	AL N	OTES	0	$=\frac{2y_{max}}{b}$	т К., К _В	as for v	vertical bendi	ing	PARAPETS P2a, P		MODEL REF. Reduced Overhau	g	M1
EP Ref.	Edge d	$\delta_{\rm S}$	4	$\frac{Im\delta_{5}}{\rho b^{4}}$	$\frac{V}{f_{1}d_{4}}$	pi	easured redicted	(θ)	$\frac{2K_By_{max}}{10^3b}$	$\left(\frac{\rho b^4}{Im}\right)$	$\frac{10^3}{4\delta s}\left(\frac{d_4}{b}\right)^{1.5}$	pi	easured redicted
ҮР ҮР ҮР ҮР ҮР		0.01 0.02 0.03 0.04 0.05 0.06				rule ¹	proposed*					rule	proposed
YP 2a YP 2a YP 2a YP 2a YP 2a YP 2a YP 2a	$k/d_a = 0.5 (V)$	0.011 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0	0.165 0.300 0.449 0.599 0.749 0.899	9.65 9.82 9.82 9.98 9.98 9.98 9.98	1.48 1.51 1.51 1.54 1.54 1.54		.221 .166 .123 .078 .073 .047	1.79 1.35 1.00 0.63 0.59 0.38		18.89 10.39 6.92 5.20 4.16 3.46	.09 .13 .14 .12 .14 .14 .12	.04 .06 .07 .06 .07 .05
YP YP 3a YP 3a YP 3a YP 3a YP 3a	$k(d_4 = 0.5 (Y))$	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	0.227 0.346 0.461 0.576 0.691	11.16 11.40 11.40 11.40 11.59	1.72 1.75 1.75 1.75 1.75 1.75		.157 .090 .075 .064 .036	1.29 0.73 0.61 0.52 0.29		13.56 9.04 6.78 5.42 4.52	.09 .08 .09 .10 .06	.04 .03 .04 .04 .03
YP YP YP YP YP YP		0.01 0.02 0.03 0.04 0.05 0.06											
U'P 2a U'P 2a U'P 2a U'P U'P U'P	Edge U'	0.015 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	0.189 0.259 0.388	10.33 10.56 10.33	1.59 1.62 1.59		.176 .144 .048	1.49 1.22 0.41		21.75 16.31 10.87	.07 .07 .04	.03 .03 .02
U'P 3a U'P 3a U'P U'P U'P U'P	Edge U'	0.015 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0 0 0	0.191 0.263	10.96 10.77	1.68 1.66		.090 .039	0.76 0.33		21.76 16.32	.03	.01 .01

¹ c.f. 6.50 ² still 6.50

3 without 3.0 factor 4 with c factor

Test Results v. Current Rules - Vertical Bending - Table M1F2

EP r	ef. = E	dge / Pa	irape	t refere	ence
Edge	K/d4	Parapet	h'd,	suffix	ò
х	0.05	P1	0.3	a	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
	U.N.O	P4	1.5	d	1

Mo	del	Units	1d,	$b^{\mu}d_{4}$	b'd ₄	$(b/d_4)^{1.3}$	$m^\prime\rho b^2$	Ka
МІ	٠	1	124	3	3	5.2	inviction for C1	0.5
342	0	1	$1^{1}/_{2}$	3	5 1/3	12.9		0.37
343	0	2	1_{24}	7^{+}	9	27.0	1.11	0.29
M4	0	2	1.%	7 1/2	10	31.6		0.27
М5	ö	3	1	12	13 1/2	49.6		0.24
M6	0	3	1 1/2	12	14 1/2	55.2		0.23

NOTE :

GENERAL NOTES	$K_p = \frac{0.87}{C} = 0.50 \ (0.525 \ \text{for U'})$	PARAPETS P2a, P3a	MODEL REF. further reduced overhang M	1
reduction for rough flow.	≈ √3	P2n, P3n	further reduced overhang 1V1	

mode	EP Ref.	Edge d	δs	Z	$\frac{m\delta_3}{\rho b^2}$	V 5.d.	_	easured redicted	ymas d.	$\frac{K_R y_{max}}{10^3 d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10^3}{4\delta s} \left(\frac{d_4}{b}\right)^{13}$		easured redicted
E	La tere	Ed	•a		100	1000	rule	proposed ²	C.7			rule ³	proposed
	YP		0.01		1								
	YP		0.02										
	YP		0.03										
	ΥP		0.04										
	YP		0.05										
_	YP	-	0.06					_					
10.00	YP 36	_	0.01	0	0.957	12.92	1.99		.0028	1.4	50.1	.03	.01
	YP 3b	- 0.3 (Y)	0.02	0	1.914	12.92	1.99		.0027	1.4	25.1	.06	.02
2.101	YP 3b	3	0.03	0	2.870	12.92	1.99		,0025	1.3	16.7	.08	.03
72.04	YP 3b	lold	0.04	0	3.827	12.92	1.99		,0022	1.1	12.5	.09	.04
2nd		1	0.05	0	4.784	12.92	1.99		.0020	1.0	10.5	.10	.04
-	YP 3h		0.06	0	5.741	12.92	1.99		.0017	0.9	8.4 50.1	.11	.10
2nd	YP 2a	0	0.01	0	0.947	8.77	1.35		.0203	10.2	50.1	-20	.10
AGE 123	YP Za	= 0.5 (Y)	0.02	0	1 0								1.1
22.0	YP	0	1.00	0									
2nd	YP	Md.	0.04	0	1 3								1000
2nd	YP YP	3	0.05	0		_					1		
2nd 2nd	YP 3b	-	0.00	0	0.963	12.96	1.99		.0035	1.8	50.1	.04	112.1
2nd 2nd	YP 3b	0	0.01	0	1.926	12.74	1.96		.0030	1.5	25.1	.06	
2nd	YP 3b	S(S)	0.02	i i	2.889	13.04	2.01		.0029	1.5	16.7	.09	
2nd 2nd	YP 3b	- 0.5	0.03	0	3.852	12.96	1.99		.0028	1.4	12.5	.11	
	YP 3b	F.14.	0.05	0	4.815	13.34	2.05		.0022	1.1	10.5	.10	
	YP 3b	-	0.01	0	5.777	13.34	2.05		.0022	1.1	8.4	.13	
	YP	-	0.01	Ť								-	
	YP		0.02								Line (a		100
	YP		0.03						1.000				1.1
	YP		0.04										
	YP		0.05										
	YP		0.06										1000
2nd	UPP 2a		0.01	0	1.169	10.13	1.56		.0127	6.7	46.9	.14	.07
2nd	U'P 2a		0.02	0	2.338	10.59	1.63		.0095	5.0	23.4	.21	.10
2nd	UP 2a	Edge U*	0.03	0	3.508	10.51	1.62		.0087	4.6	15.6	.28	.14
2nd	UP 2a	dig.	0.04	0	4.677	10.89	1.67		.0061	3.2	11.7	.27	.13
2nd	U'P 2a	3	0.05	0	5.846	10.89	1.67		.0053	2.8	9.4	_29	.14
2nd	UP 2a		0.66	0	7.015	11.27	1.73		.0031	1.6	7.8	.21	.10

¹ c.f. 6.5 ² still 6.5 3 without 3.0 factor

4 with c factor

Test Results v. Current Rules - Galloping - Table M1F2

				t refere	
Edge	K/d4	Parapet	h/d ₄	suffix	ò
х	0.05	P1	0.3	а	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
	U.N.O	P4	1.5	d	1

Mo	del	Units	$\ /d_{\alpha}$	b^{μ}/d_4	h/d _a	(b/d) ^{1.5}	(b/d)3.3 m/pd3
MI	•	1	1/4	3	3	9	864 (885 U')
M2	0	1	$1^{ij}{}_2$	3	5 1/2	12.9	
М3	0	2	1	7^{1} / ₂	9	27.0	
M4	0	2	$1^{-1}\!/_{2}$	7 1/2	10	31.6	
М5	0	3	1	12	13 1/2	49.6	
M6	0	3	1 1/2	12	14 1/2	55.2	

GENER	AL N	OTES	c	• →	• 1.0	Galloping		PARAPET P2a.		MO furth	DEL REF.	werhang	M1
EP Ref.	Edge P*	δs	2	$\frac{m\delta_5}{\rho b^2}$	$\frac{V_z}{f_0 d_z}$	$C_{G} = \left(\frac{m}{\rho b^{2}}\right) \delta_{5} \left(\frac{b}{d_{3}}\right)^{2}$	measured predicted		-	meas predi	1. The second	3.3	
YP	33	0.01			-				(5.0	ŋ	(N/A)	-	
YP YP YP YP YP		0.02 0.03 0.04 0.05 0.06											
YP		0.01	0		-				-		-		
YP 3b	3	0.02	0	1.914	18.75	17.28	1.09						1.26
YP YP 3b YP	kd 0.3 (Y)	0.03 0.04 0.05	0	3.827	61.47	34.56	1.78						
YP 3b	-	0.06	0	5.741	75.74	51.84	1.46						
YP YP 2a	ŝ	0.01 0.02	0	1.894	17.46	17.28	1.01						
YP YP 2a YP	$b/d_4 = 0.5 (Y)$	0.03 0.04 0.05	0	3.789	17.61	34.56	.51		1				
YP 2a	-	0.06	0	5.683	17.76	57.84	.34						
ҮР ҮР 3Ь ҮР	5 (V)	0.01 0.02 0.03	0 0 0 0	1.926	18.36	17.28	1.06						
ҮР 3Ь ҮР	kd4 = 0.5 (V)	0.04	0	3.852	58.45	34.56	1.69						
YP 3b		0.06	0	5.777	69.69	51.84	1.34		-	_		-	
YP YP YP		0.01 0.02 0.03											
YP YP		0.04											
YP		0.06							-	_		-	
U'P U'P 2a U'P	5	0.01 0.02 0.03	0 0	2,338	14.90	17.7	.84						
UP 2a UP 2a	Edge U'	0.03	0	4.677	15.12	35.4	.43						
U'P 2a		0.06	0	7.015	15.20	53.1	.29						

Test Results v. Current Rules - Torsion - Table M1F2

EP ref. = Edge / Parapet reference											
Edge	K/d ₄	Parapet	h/d4	suffix	ģ						
X	0.05	P1	0.3	a	1/4						
	U.N.O	P2	0.7	b	1/2						
Y	0.20	P3	1.1	c	3/4						
_	U.N.O	P4	1.5	d	1						

Мо	del	Units	Vd4	$b^{a_{ij}}d_{k}$	b/d4	$(b/d_4)^{1.5}$	m/ph ²	Ka	
MI	•	1	1/4	3	3	47	948 (548) U*	0.50	
M2	0	18	$1^{-1}/_{2}$	3	5 1/2	390		0.37	
M3	0	2	1	$7^{-1}/_{2}$	9	2187		0.29	
M4	0	2	$1^{-1}/_{2}$	7^{-1} /2	10	3162		0.27	
M5	0	3	1	12	13 1/2	9040		0.24	
M6	0	3	1 1/2	12	14 1/2	11609		0.23	

-NOTE :

	GENER.	ALN	OTES	θ.	2yms b	; K _R a	is for ve	ertical bendi	ng	PARAPET P2a, 1	· · · · · · · · ·	MODEL Reduced		ng	M1
mode	EP Ref.	Edge Det.	δs	4	$\frac{Im\delta_{S}}{\rho b^{4}}$	$\frac{V}{f_{T}d_{4}}$		easured edicted proposed ²	$\left(\theta^{o}\right)_{s}$	$\frac{2K_{H}y_{max}}{10^{-3}b}$	(pb ⁴)	$\frac{10^3}{4\delta s}\left(\frac{d}{b}\right)$) 11		edicted proposed
and a second sec	ҮР ҮР ҮР ҮР ҮР ҮР	-	0.01 0.02 0.03 0.04 0.05 0.06					proposa							
	YP 36 YP 36 YP 36 YP 36 YP 36 YP 36 YP 36	k/d_4 = 0.3 (Y)	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	Sta	ble			Stable			14.1			11111
A REAL PROPERTY OF A REAL PROPER	YP 2a YP 2a YP 2a YP 2a YP 2a YP 2a YP 2a	$\operatorname{kold}_{4} = 0.5 (V)$.015 0.02 0.03 0.04 0.05 0.06	000000000000000000000000000000000000000	0.302 0.403 0.605 0.807 1.008 1.210	9.83 9.83 10.01 10.01 10.25 10.34	1.51 1.51 1.54 1.54 1.58 1.58		.174 .159 .088 .086 .048 .037	1.52 1.39 0.77 0.75 0.42 0.32		17.58 13.19 8.79 6.59 5.27 4.40		.09 .11 .09 .11 .08 .07	.04 .05 .04 .05 .04 .03
	YP 3b YP 3b YP 3b YP 3b YP 3b YP 3b YP 3b	$k/d_4 = 0.5 (V)$	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	Su	ble			Stable			1			11111
	ҮР ҮР ҮР ҮР ҮР ҮР	Edge U	0.01 0.02 0.03 0.04 0.05 0.06	0000000											
	U'P 2a U'P 2a U'P 2a U'P 2a U'P 2a U'P 2a	Edge Ut	.015 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	0.243 0.317 0.476	10.08 9.81 9.81	1.55 1.51 1.51		.060 .059 .024	0.55 0.54 0.22		29.80 22.80 15.30		.02 .02 .02	.01 .01 .01

¹ c.f. 6.5 ² still 6.5 ³ without 3.0 factor ⁴ with c factor

Test Results v. Current Rules - Divergent TI - M1X & Y (continued)

EP	ref. =)	Edge / P	arapet	t referen	ice
Edge	K/d,	Parapet	h/d4	suffix	φ
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

3100	del.	Units	l'd,	$b^{\ast} d_{4}$	bid,	$(b/d)^2$	m/pb ²
MI	•	1	1	3	4 1/3	20	
MZ	0	1	$1^{-1}/_{2}$	3	5 1/2	30	
М3	0	2	1	$7 y_2$	9	81	
M4	0	2	$1^{-1}/_{2}$	7 1/2	10	100	
M5	0	3	1	12	13 1/2	182	
M6	0	3	1 1/2	12	14 1/2	210	

GENER/	AL NOT	ES	Initia	ally C _t →	1.0			PARAPETS X & YP 1	& 2	MODEL R	EE.	M1
EP Ref.	δs	2	$\frac{Im\delta_S}{\rho b^4}$	$\frac{V_g}{f_T d}$	predicted		or			edicted (2.6)	3.3	
XP 1a	0.023	0	0.119	3.80	5 (or 2.6)	-	3.3		.76	1.46	1.15	
YP 1a	0.023	0	0.119	3.79	5 (or 2.6)		3.3		.76	1.46	1.15	
XP 1a	0.040	0	0.214	4.04	5 (or 2.6)	-	3.3		.81	1.55	1.23	
YP 1a	0.040	0	0.214	3.92	5 (or 2.6)	- 1	3.3		.78	1.51	1.18	
XP Ia	0.060	0	0.320	4.40	5 (or 2.6)		3.3	12	.88	1.69	1.33	
VP 1a	0.060	0	0.320	4.15	5 (or 2.6)		3.3		.83	1.60	1.26	
XP 1b	0.023	0	0.119	3.42	5 (or 2.6)		3.3		.68	1.32	1.03	
YP 1b	0.023	0	0.119	3.30	5 (or 2.6)		3.3		.66	1.27	1.00	
XP 1b	0.040	0	0.214	3.66	5 (or 2.6)		3.3		.73	1.41	1.11	
YP 1b	0.040	0	0.214	3.66	5 (or 2.6)	_	3.3		.73	1.41	1.11	
XP 1b	0.060	0	0.320	3.88	5 (or 2.6)		3.3		.78	1.49	1.18	
YP 1b	0.060	0	0.320	3.84	5 (or 2.6)	1	3.3		.77	1.48	1.17	
XP 1d	0.023	0	0.119	3.49	5 (or 2.6)		3.3		.70	1.34	1.06	
XP 2d	0.023	0	0.119	4.06	5 (or 2.6)		3.3			1.56	1.23	
XP 1d	0.040	0	0.214	3.79	5 (or 2.6)		3.3		.76	1.46	1.15	1.000
XP 2d	0.040	0	0.214	4.44	5 (or 2.6)		3.3			1.71	1.35	
XP 1d	0.060	0	0.320	4.11	5 (or 2.6)		3.3		.82	1.58	1.24	
XP 2d	0.060	0	0.320	5.00	5 (or 2.6)		3.3		1.00	1.92	1.52	
XP 1d	0.023	-5	0.119	3.15	5 (or 2.6)	ž	3.3		.63	1.21	.95	
YP 1d	0.023	-5	0.119	3.39	5 (or 2.6)	Furbalent Flow	3.3		.68	1.30	1.03	
XP 1d	0.040	-5	0.214	3.28	5 (or 2.6)	Ē	3.3		.66	1.26	1.00	
YP 1d	0.040	-5	0.214	3.48	5 (or 2.6)	1	3.3		.70	1.34	1.06	
XP 1d	0.060	-5	0.320	3.51	5 (or 2.6)	121	3.3		.70	1.34	1.06	
YP 1d	0.060	-5	0.320	4.03	5 (or 2.6)	-	3.3		.81	1.55	1.23	
XP 2b	0.023		0.119	3.39	5 (or 2.6)		3.3		.68	1.30	1.03	1.0
YP 2b	0.023		0.119	4.24	5 (or 2.6)		3.3		.85	1.63	1.29	
XP 2b	0.040		0.214	3.65	5 (or 2.6)		3.3		.73	1.40	1.11	
VP 2b	0.040		0.214	4.38	5 (or 2.6)		3.3		.88	1.68	1.33	
XP 2b	0.060		0.320	4.22	5 (or 2.6)		3.3		.85	1.63	1.29	
YP 2b	0.060		0.320	4.60	5 (or 2.6)		3.3		.92	1.77	1.39	-
XP 1a*	0.023	0	0.119	3.69	5 (or 2.6)		3.3		.74	1.42	1.12	
XP	0.023	0		1								
XP 1a* XP	0.040	0	0.214	3.78	5 (or 2.6)		3.3		.76	1.46	1.15	* Middle Third
XP 1a* XP	0.060	0	0.320	4.03	5 (or 2.6)		3.3		.81	1.55	1.23	Solid

Test Results v. Current Rules - Vertical Bending - Table M1X

EP	ref. =	Edge / P	arape	t referei	ice
Edge	K/d4	Parapet	h/d4	suffix	ф
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Mo	del	Units	14.	$b^{\mathbf{s}}/d_{\mathbf{s}}$	b/d4	(b/d ₄) ^{1,2}	$m / \rho b^T$	KR	
M1	•	1	1	3	4 1/2	9.5	492	0.41	
M2	0	1	$1^{1}b$	3	5 1/2	12.9	14		
MB.	0	2	1	7 1/2	9	27.0			
M4	0	2	$1 \eta_{\sharp}$	7 1/2	10	31.6			
M5.	0	3	1	12	13 1/2	49.6			
M6	0	3	$1^{i_{i_2}}$	12	14 1/2	55.2			

GENER/	AL NOT	ES	K _a (re	l ^e for rou	igh flow)	$=\frac{0.87}{\sqrt{4.5}}=0$	1.41	NP1 - 4d 8	-	MODEL R	EF.	M1
EP Ref.	δ	2	$\frac{m\delta_{S}}{\rho b^{2}}$	$\frac{V}{f_k d_4}$		edicted	$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_{it}y_{max}}{10^3d_4}$	$\left(\frac{\rho b^3}{m}\right) \frac{10}{4\delta}$			rasured edicted
					rule ¹	proposed ³					rule ³	proposed
XP 1d	0.01	5	0.518	9.01	1.39		.0964	39.5	50.8		.78	.65 *
XP 1d	0.02	5	1.035	9.52	1.46		.0788	32.3	25.4		1.27	1.06 *
XP 1d	0.03	5	1.553	9.24	1.42		.0646	26.5	16.9		1.56	1.30 *
XP 1d	0.04	5	2.071	9.23	1.42		.0542	22.2	12.7		1.75	1.46 *
XP 1d	0.05	5	2.589	8.98	1.38	1 1	.0474	19.4	10.2		1.90	1.58 *
XP 1d	0.06	5	3.107	8.98	1.38		.0416	17.1	8.5		2.01	1.68 *
XP 1d	0.01	0	0.518	10.08	1,55		.0740	30.3	50.8		0.60	,50
XP 1d	0.02	0	1.035	9.23	1.42		.0554	22.7	25.4		0.89	.74
XP 1d	0.03	0	1.553	9.53	1.46		.0494	20.3	16.9		1.20	1.00
XP 1d	0.04	0	2.071	8.98	1.38		.0392	16.1	12.7		1.27	1.06
XP 1d	0.05	0	2.589	8.74	1.34	1 1	.0312	12.8	10.2		1.25	1.04
XP 1d	0.06	0	3.107	8.75	1.35		.0264	10.8	8.5		1.27	1.06
XP 2d	0.01	0	0.518	9.79	1.51		.1498	61.4	50.8		1.21	,50
XP	0.02	.0	100000	1.2695.47	1,00,000	1 - 1	1	10.51%	1100000			1.000
XP	0.03	0										
XP	0.04	0										
XP	0.05	0										
XP	0.06	0										
XP 3d	0.01	0	0.518	6.85	1.05		.0330	13.5	50.8		0.27	.08
XP 3d	0.02	0	1.035	6.85	1.05		.0174	7.1	25.4		0.28	.08
XP 3d	0.03	0	1.553	6.83	1.05		.0114	4.7	16.9		0.28	.08
XP 3d	0.04	0	2.071	6.80	1.05	1 - 1	.0068	2.8	12.7		0.22	.06
XP	0.05	0										
XP	0.06	0	-							11000		12200
XP 4d	0.01	0	0.518	6.80	1.05		.0262	10.7	50.8		0.21	.04
XP 4d	0.02	0	1.035	6.78	1.04		.0164	6.7	25.4		0.26	.06
XP 4d	0.03	0	1.553	6.75	1.04		.0112	4.6	16.9		0.27	.06
XP 4d	0.04	0	2.071	6.78	1.04		.0072	3.0	12.7		0.23	.05
XP	0.05	0			1.00000			0.000	1.1.1.1.1.1		1.1.200.00	1
XP	0.06	0										
XP 2b	0.01	0	0.518	10.80	1.66		.0106	4.3	50.8		0.09	.07
XP 2b	0.02	0	1.035	11.32	1.74		.0048	2.0	25.4		0.08	.06
XP 2b	0.03	0	1.553	11.14	1.71		.0038	0.6	16.9		0.09	.07
XP	0.04			100000				21100	1000000			1
XP	0.05											
XP	0.06											

¹ c.f. 6.50 ² still 6.50 3 without 3.0 factor

4 with c factor

* Note ∠_ = +5°

Test Results v. Current Rules - Vertical Bending - Table MIX (continued)

EP	ref. =	Edge / P	arape	t refere	nce
Edge	K/d _e	Parapet	h/d_4	suffix	ò
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Mo	del	Units	$1d_4$	b^{α}/d_{α}	b/d ₄	(b:d4)1.5	m/pb ²	Ka
MI.	•	1	1	3	4 1/2	9.5	432	0.41
12	0	1	1^{1}_{2}	з	5 1/2	12.9		
43	0	2	1	7 1/2	9	27.0		
44	0	2	$1 \frac{1}{2}$	7 1/2	10	31,6		
115	0	3	1	12	13 1/2	49.6		
\16	0	3	1 1/2	12	14 1/2	55.2		

GENER/	AL NOT	ES	K _R	= 0.4	1			PARAPETS NPla-d		MODEL RI	<u>SF.</u>	M1
EP Ref.	δ	2	$\frac{m\delta_S}{\rho b^2}$	V 64		easured vdicted	$\left(\frac{y_{max}}{d_4}\right)_{i}$	$\frac{K_R y_{max}}{10^{-3} d_4}$	$\left(\frac{\rho b^2}{m}\right)\frac{10^3}{4\delta s}$	$\left(\frac{d^4}{b}\right)^{1.5}$	-	rasured edicted
	_				rule ¹	proposed ²	1 1	-			rule ³	proposed
XP 1a	.0128	0	0.582	8.28	1.27		.0296	12.1	45.2		.27	.51
XP 1a	0.02	0	0.909	8.33	1.28		.0248	10.2	29.0		.35	.67
XP 1a	0.03	0	1.363	8.39	1.29		.0206	8.5	19.3		.44	.84
VP 1a	0.04	0	1 818	8.30	1.20		0512	7.0	115		10	

	11/10				10.4 40			0.00				
.84	.44		19.3	8.5	.0206		1.29	8.39	1.363	0	0.03	XP 1a
.91	.48		14.5	7.0	.0542		1.29	8.39	1.818	0	0.04	XP 1a
.99	.52		11.6	6.1	.0170		1.29	8.36	2.272	0	0.05	XP 1a
1.01	.53		9.6	5.2	.0126		1.28	8.31	2.726	0	0.06	XP 1a
.78	.58		45.2	26.4	.0644		1.36	8.83	0.582	0	.0128	XP 1b
.95	.71		29.0	20.6	.0502		1.36	8,87	0.909	0	0.02	XP 1b
1.13	.85		19.3	16.5	.0402		1.30	8.42	1.363	0	0.03	XP 1b
	1			- 1997					1		0.04	XP
									6 5		0.05	XP
											0.06	XP
1.05*	1.26	(0.60)	45.2	57.0	,0570	2	1.47	9.58	0.582	0	.0128	XP 1d
1.30*	1.56	(0.89)	29.0	45.2	.0452	Plaw	1.45	9.46	0.909	0	0.02	XP 1d
1.47*	1.77	(1.20)	19.3	34.2	.0342	Ŧ	1.40	9.13	1.363	0	0.03	XP 1d
.94*	1.13	(1.27)	14.5	16.4	.0164	Turbulent	1.31	8.49	1.818	0	0.04	XP 1d
.88*	1.05	(1.25)	11.6	12.2	.0122	Ŧ	1.33	8.67	2.272	0	0.05	XP 1d
		smooth m/p		$K_R \rightarrow 1.0$	→ i.e	-		1000	1	0	0.06	YP
.22*	.26		45.2	11.8	.0288		1.41	9,18	0.582	-5	.0128	XP 1d
.26*	.31		29.0	9.0	.0220		1.41	9.18	0.909	-5	0.02	XP 1d
.38*	.45		19.3	8.7	.0212		1.45	9.41	1.363	-5	0.03	XP 1d
.47*	.56		14.5	8.1	.0198		1.44	9.39	1.818	-5	0.04	XP 1d
.53*	.64		11.6	7.4	,0180		1.45	9.42	2.272	-5	0.05	XP 1d
.59*	.71		9.6	6.8	.0166		1.44	9.39	2.726	-5	0.06	XP 1d
See	.09		50.8	4.3	.0106		1.66	10.80	0.518	0	0.01	XP 2b
P. MIX	.08		25.4	2.0	.0048		1.74	11.32	1.035	0	0.02	XP 2b
	.09		16.9	1.6	,0038		1.71	11.14	1.553	0	0.03	XP 2b
	1.21		50.8	61.4	,1498		1.51	9,79	0.518	0	0.01	XP 2d
1.45*	3.47	(1.21)	45.2	157.20	.1572	Turh	1.38	8.99	0.582	0	.0128	XP 2d
		10.00		$K_B \rightarrow 1.0$	-> i.e	2		100		1	2004 BA	XP
.47"	.35		45.2	15.9	.0388		1.36	8.83	0.582	0	.0128	XP 1a ⁴
.59"	.44		29.0	12.7	.0310		1.37	8.90	0.909	0	0.02	XP 1a"
.65"	.49		19.3	9.5	.0232		1.30	8.43	1.363	0	0.03	XP 1a*
.69*	.52		14.5	7.5	,0184		1.30	8.46	1.818	0	0.04	XP 1a4
.72*	.54		11.6	6.2	.0152		1.30	8.46	2.272	0	0.05	XP 1a*
.71"	.53		9.6	5.1	.0124		1.30	8,45	2.726	0	0.06	XP 1a*

² still 6.50 (proposed rul

without 3.0 factor

(proposed rule averaged 1/3 & 2/3) * Note turbulent flow

4 with c factor

or -5° /

Test Results v. Current Rules - Galloping - M1X

Edge	K/d4	Parapet	h da	saffix	ģ
х	0.05	Pl	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Mo	del	Units	Vd ₄	b^*/d_4	b/d4	$(b/d)^2$	m/pb ²
MI	•	1	1	3	4 1/2	20	1049
M2	0	1	$1 \ ^{1}\mathrm{b}$	3	5.1/2	30	
мв	0	2	1	7 1/2	9	81	
M4	0	2	$1^{1}\!/_{2}$	7 1/3	10	100	
M5	0	3	1	12	13 1/2	182	
M6	0	3	1 1/2	12	14 1/2	210	

GENERA	L NOT	ES	Initia	ally $C_i \rightarrow$	1.0		PARAPETS NP 1-4d	MODEL REF.	M1
-		-	2 2			11 C			
EP Ref.	$\delta_{\rm S}$	2	$\frac{m\delta_{5}}{\rho b^{2}}$	$\frac{V_g}{f_s d_4}$ C	$c_{\rm ci} = \left(\frac{m}{cb^2}\right) \delta_{\rm S} \left(\frac{b}{d_{\rm s}}\right)^2$	<u>ш</u> р			
XP	0.01	5				-			
XP 1d	0.02	5	1.035	21.6	21.0	1.03		-	
XP	0.03	5							
XP 1d XP	0.04	5	2.071	22.1	42.0	.53			
XP 1d	0.05	5	3.107	22.2	62.9	.35			
XP	0.00	0	20107		112.7				
XP 1d	0.02	0	1.035	22.5	21.0	1.07			
XP	0.03	0	101000	122269		1.000			
XP 14	0.04	0	2.071	23.4	42.0	,56			
XP	0.05	0	000000			1.11			
XP 1d	0.06	0	3.107	23.7	62.9	.38			
XP	0.01	0							
XP 2d	0.02	0	1.035	22.1	21.0	1.05			
XP	0.03	0	2.071	22.8	42.0	.54			
XP 2d XP	0.04	0	2.9/1	22.8	42.0	-54			
XP 2d	0.06	0	3.107	23.5	62.9	.37			
XP	0.01	0					1		
XP 3d	0.02	0	1.035	21.2	21.0	1.01			
XP	0.03	0	111111	10000		1000			
XP 3d	0.04	0	2.071	21.8	42.0	.52			
XP	0.05	0				1			
XP 3d	0.06	0	3.107	22.5	62.9	.36			
XP	0.01	0		1.1.1	1.1.1	1000			
XP 4d	0.02	0	1.035	21.0	21.0	1.00			
XP	0.03	0	3.071	71.9	42.0	.52			
XP 4d XP	0.04	0	2.071	21.8	42.0	4			
XP 4d	0.05	0	3.107	22.0	62.9	35			
XP	0.01	-	51107		that y				
XP	0.02								
XP	0.03							-	
XP	0.04								
XP	0.05								
XP	0.06				<u> </u>			10.00	

Test Results v.	Current	Rules -	Torsion -	Table M1X
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EP	ref. =	Edge / P	arape	t refere	nce
Edge	K/d4	Parapet	h/d4	suffix	ф
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
		P4	1.5	d	1

Mode	el	Units	l'd4	b^{a}/d_{4}	b/d _#	(b/d ₄) ¹³	$m / \rho b^2$	$K_{\rm R}$
11	•	1	1	3	4 1/2	193	1326	0.41
12	0	1	1 1/2	3	5 1/2	390		0.37
43	0	2	1	$7^{-1}/_{2}$	9	2187		0.29
44	0	2	1 1/4	7 1/2	10	3162		0.27
15	0	3	1	12	13 1/2	9040		0.24
16	0	3	1 1/2	12	14.1/2	11609	-	0.23

GENER	AL NOT	ES	$\theta = \frac{2}{2}$	y _{mm} ^T b	C _R as for	vertical ben	ling	PARAPETS XP 1-		MODEL R	EF.	M1
EP Ref.	δs	2	$\frac{Im\delta_{5}}{\rho b^{4}}$	$\frac{V}{f_{T}d_{4}}$		easured edicted proposed ²	(θ"),	2K ₈ y _{max} ^T 10 ⁻³ b	$\left(\frac{\rho b^4}{Im}\right)\frac{10^3}{4\delta s}$	$\left(\frac{d_4}{b}\right)^{13}$	pr	easured edicted
XP	0.01	\vdash		-	ruic	proposed					rule	propose
XP	0.02											
XP 1d	0.03	5	0.206	9.32	1.43		1.147	8.21	6.28		1.31	1.09
XP 1d	0.04	5	0.274	9.20	1.42		0.808	5.78	4.71		1.23	1.02
XP 1d	0.05	5	0.343	9.19	1.42	1 1	0.635	4.54	3.77		1.20	1.00
XP 1d	0.06	5	0.412	9.18	1.41		0.499	3.57	3.14		1.14	.95
XP 1d	0.01	0	0.069	10.92	1.68		2.540	18.17	18.85		.96	.80
XP 1d	0.02	0	0.137	9.50	1.46		0.986	7.06	9.43		.75	.63
XP 1d	0.03	0	0.206	9.20	1.42		0.637	4.56	6.28		.73	.61
XP	0.04	0										
XP	0.05	0										
XP	0.06	0				1 1						
XP	0.01	0			-		-		1000		1	
XP 2d	0.02	0	0.137	11.42	1.76		4.350	31.12	9.43		3.30	1.38
XP	0.03	0										
XP	0.04	0										
XP	0.05	0										
XP	0.06	0					_					
XP 3d	0.01	0	0.103	11.09	1.70		5.360	38.36	18.85		2.03	.56
XP	0.02	0						-			-	
XP	0.03	0			1 13							
XP	0.04	0										
XP	0.05	0						and the second s				
XP	0.06	0			í							
XP 4d	0.01	0	0.103	10.75	1.65		4.551	32.57	18.85		1.73	.36
XP 4d	0.02	0	0.137	10.75	1.65		2.709	19.39	9.43		2.06	.43
XP	0.03	0										
XP	0.04	0										
XP	0.05	0										
XP	0.06	0			-							-
XP	0.01											
XP	0.02											
XP	0.03											
XP	0.04											
XP	0.05											
XP	0.06											3.0 facto

² still 6.5

³ without 3.0 facto ⁴ with c factor

Test Results v. Current Rules - Torsion - Table M1X(continued)

Edge	K/d ₄	Parapet	h'd.	suffix	
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
- 1 I		P4	1.5	d	1

Мо	del	Units	1 d.	$p_{n,0}q^{\dagger}$	$\mathbf{h} \cdot \mathbf{d}_{a}$	$\langle h d_i \rangle^{1.9}$	$m/\rho b^{\ddagger}$	Ka
MI		1	1	3	4 %	193	1029	0.41
MZ	0	1	1^{-1} ₇	3	5 1/2	390		0.37
М3	0	2	1	7 1/2	9	2187		0.29
M4	0	2	$1^{ij} z$	7 1/2	10	3162		0.27
M5	0	3	1	12	13 1/2	9040		0.24
M6 :	0	3	1^{1}	12	14 ¹ / ₂	11609		0.23

GENERA	LL NOT	ES	$\theta = \frac{2}{2}$	y _{max} ; k b	_R as for	vertical bend	ling	PARAPETS XP 1a-d		ODEL REF		M1
EP Ref.	δs	Z	$\frac{Im\delta_{s}}{\rho b^{4}}$	$\frac{V}{f_{T}d_{4}}$		-asured edicted	$\left(\theta^{a}\right)_{s}$	2Ka ymax ¹ 10 ³ b	$\left(\frac{\rho b^{i}}{Im}\right) \frac{10^{i}}{4\delta s}$	$\left(\frac{d_4}{b}\right)^{1/2}$	-	asured edicted
					rule ¹	proposed3	s				rale ³	proposed
XP 1a XP XP XP XP XP XP	0.01 0.02 0.03 0.04 0.05 0.06	0	Sta	ble			5	itable				
XP XP 1b XP 1b XP 1b XP XP XP	0.01 .0223 0.03 0.04 0.05 0.06	0 0 0	0.119 0.160 0.214	9.55 9.81 9.28	1.47 1.51 1.43		.0870 .0610 .0570	0.62 0.44 0.41	10.89 8.10 6.07		.06 .05 .07	.08 .07 .09
XP 1d XP XP XP XP XP	0.01 0.02 0.03 0.04 0.05 0.06	0	Sta	ble		Turbalent Flow		stable				Turbulent Flow (see M1X for smooth
XP XP 1d XP 1d XP 1d XP 1d XP 1d XP XP XP 2b XP 2b	0.01 .0223 0.03 0.04 0.05 0.06 0.01 .0223 0.03	* * * * * *	0.119 0.160 0.214 0.267 0.320 0.119 0.160	9.62 9.61 9.62 9.63 9.90 11.17 11.40	1.48 1.48 1.48 1.48 1.52 1.72 1.75		.2150 .1940 .1650 .1370 .1220 .0780 .0410	1.54 1.39 1.18 0.98 0.87 0.56 0.29	10.89 8.10 6.07 4.86 4.05 10.89 8.10		.14 .17 .19 .20 .22 .05 .04	.12 .14 .16 .17 .18 .04 .03
XP XP 2d XP	0.01 0.02 0.03			able		Turb		Stable			7	Turb. Fl. (See M1X for smooth
XP 1a* XP 1a* XP 1a* XP 1a* XP 1a* XP 1a*	0.01 .0223 0.03 0.04 0.05 0.06		0.119 0.160 0.214 0.267 0.320	10.94 11.52 11.51 11.51 11.49	1.68 1.77 1.77 1.77 1.77		.1400 .1050 .0890 .0700 .0610	1.00 0.75 0.64 0.50 0.44	10.89 8.10 6.07 4.86 4.05		.09 .09 .10 .10 .11	0.12* 0.12* 0.13* 0.13* 0.15*

² still 6.5

⁴ without 3.0 factor

* middle third solid (prposed rule averaged 1/3 & 2/3)

Test Results v. Current Rules - Vertical Bending - Table M1Y (continued)

EP	ref. =	Edge / P	arape	t refere	nce
Edge	K/d _s	Parapet	h/d ₄	suffix	- ¢
х	0.05	P1	0,3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
		P4	1.5	d	1

Me	del	Units	1/d _i	$b^{\ast}d_{4}$	b/d ₄	$(h/d_{\rm s})^{1.3}$	$m/\rho b^2$	Ka
М1	٠	1	1	3	4 1/2	09.5	432	0.41
M2	0	1	$1 \stackrel{h_{\pm}}{\to}$	3	5 1/2	12.9		
KI3	0	2	1	7 1/2	9	27.0		
M4	0	2	$1^{-1}/_{2}$	7 1/2	10	31.6		
M5	0	3	1	12	13 1/2	49.6		
M6	0	3	1.7_{1}	12	14 1/2	55.2		

	AL NOT							PARAPETS		MODEL R	EF.	3.54
K _B	= 0.41	(as f	for MIY)	; m/p	b ² is now	45.4, was 57.	8	YP1a, b;	2a, b			M1
			mős	v		easured	Yunes	Kayma	(pb ²) 10 ³	(ª)13	1.1	easured
EP Ref.	$\delta_{\rm S}$	4	bp_2	f _s d ₄		rdicted	(4,)	$10^{-3}d_{4}$	(m) 48s	5		edicted
VID 4	0120		0.000	0.40	rule	proposed ²		8 		85 (S)	rule	proposed
YP 1a	.0128	0	0.582	8.28	1.27		.0364	14.9	45.2		.33	.40
YP 1a	0.02	0	0.909	8.33	1.28	1 1	.0290	11.9	29.0		.41	.50
YP 1a YP 1a	0.03	0	1.363	8.23	1.27	1 1	.0232	9.5	19.3		.49	.59
YP 1a	0.04	0	1.818	8.33	1.28	1 1	.0190	7.8	14.5		.54	.65
YP Ia	0.06	0	2.272	8.35	1.29		.0160	6.6	11.6		-57	.69
YP 1b	.0128	0	2.726	8.35	1.29		.0138	5.7	9,6	_	-59	.72
YP 1b	0.02	0	0.584	8.37	1.29		.0572	23.5	45.2		.52	.50
YP 1b	0.02	0	1.363	8.29 8.32	1.27		.0468	19.2	29.0		.66	.63
YP 1b	0.03	0	1.363	8.36	1.28		.0392	16.1	19.3		.83	.79
YP 1b	0.04	0	2.272	7.83	1.29		.0310	12.7	14.5		.88	.84
YP 1b	0.06	0	2.726	7.89	01000		,0252	10.3	11.6		.89	.85
YP	.0128	-	2.720	/.89	1.21	<u> </u>	,0004	0.2	9.6		.02	.02
YP	0.02											
YP	0.02											
YP	0.04											
YP	0.05											
YP	0.06											
YP	.0128	-								_		
YP	0.02											
YP	0.03	1.1										
YP	0.04								1			
YP	0.05											
YP	0.06											
YP 2b	.0128	0	0.582	9.62	1.48		.0154	6.3	45.2		11	00
YP 2b	0.02	0	0.909	10.61	1.63		.0062	2.5	45.2		.14	.08
YP 2b	0.03	0	1.363	10.93	1.68		.0044	1.8	19.3		10 70 B P	1.120.0
YP 2b	0.04	0	1.818	11.10	1.71		.0038	1.5	19.5		.09	.05
VP	0.05	0	1.010				50030	1.0	14.5		.11	.07
VP	0.06	0										
VP	.0128		-				-	_	-	-	-	
VP	0.02											
YP	0.03											
VP	0.04											
YP	0.05											
YP	0.06											

³ without 3.0 factor ⁴ with c factor

Test Results v.	Current Rules - Ga	alloping - Table M	11Y
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EP ref. = Edge / Parapet reference					
Edge	K/d ₄	Parapet	h d ₄	suffix	, þ
X	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Μο	dei	Units	Vd4	b^{\ast}/d_{4}	b/d ₄	(b/d)2	m/pb ²
MI		1	1	3	4 1/2	20	1049
M2	0	1	$1^{-1}/_{2}$	3	5 1/2	30	
M3	0	2	1	7 1/2	9	81	
M4	0	2	$1 \frac{1}{2}$	7 1/2	10	100	
М5	0	3	1	12	13 1/2	182	1.1
M6	0	3	1 1/2	12	14 1/2	210	

GENERA	L NOT	ES	Initia	dly Cg →	1.0		PARAPE YP	<u>TS</u> 1-4d	MODEL REF.	M1
EP Ref.	$\delta_{\rm S}$	Z	$\frac{m\delta_s}{\rho b^2}$	$\frac{V_{g}}{f_{g}d_{4}}$ C	$_{G} = \left(\frac{m}{c^{b^{2}}} \right) \delta_{S} \left(\frac{b}{d_{4}} \right)^{2}$	<u>m</u> ρ		$\binom{m}{p}$	Torsion	-0
YP	0.01	5					-			-
YP 1d	0.02	5	1.035	25.2	21.0	1.20				
YP	0.03	5	100			125695-1				10.000
YP 1d	0.04	5	2.071	25.7	42.0	.61		1.00	1.110.001	1.00
YP	0.05	5								
YP 14	0.06	5	3.107	25.7	62.9	.41			100	
YP	0.01		Sec. de		100	1.00	100		1.	
YP 1d	0.02		1.035	20.4	21.0	.97				
YP	0.03							1		
YP	0.04							1.1.1		
YP	0.05						1			
YP 1d	0.06	T	0.056	2,95	5 or 2.6	-	+	.59	or 1.13	Torsion
YP	0.01									
YP 2d	0.02		1.035	25	21.0	1.19				
YP	0.03									
YP 2d	0.04		2.071	26.0	42.0	.62				1.125
YP	0.05			24	62.9			1 1		
YP 2d	0.06	-	3.107	27	62.9	.43		_		-
YP	0.01		1.070	21.6	21.0	1.03				
YP 3d YP	0.02		1.035	21.6	21.0	1.05				
YP 3d	0.03		2.071	22.3	42.0	.53				
YP	0.04		2.071	44.3	44.0					
YP 3d	0.06		3.107	23.5	62.9	.37				
YP	0.01	-	2.141	adre	Mary		1	-		
YP 4d	0.02		1.035	22.0	21.0	1.05				
YP	0.03		10,000							
YP 4d	0.04		2.071	22.4	42.0	.53				
YP	0.05				0.00				-	
YP 4d	0.06		3.107	22.4	62.9	.36				and the second
YP	0.01					_				
YP	0.02									
YP	0.03									1.000
YP	0.04									
YP	0.05									100
YP	0.06	T	0.137	03.25	5 or 2.6 -		-	.65	or 1.25	Torsio

Torsion cases shown T

Test Results v. Current Rules - Torsion - Table M1Y

EF	ref. =	Edge / P	arape	t referei	nce
Edge	K/d ₄	Parapet	h/d4	suffix	ó
Х	0.05	P1	0.3	a	-1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Mo	del.	Units	ld4	$b^{\mathbf{z}_i}d_i$	b/d4	$(h \cdot d_{\mathfrak{g}})^{1.2}$	m/pb ²	KR
М1	•	1	1	3	4 1/2	193	1326	0.41
M2	0	1	1 1/2	3	5 1/2	390		0.37
M3	0	2	1	7 $\frac{1}{2}$	9	2187		0.29
M4	0	2	1 1/2	7 1/2	10	3162		0.27
M5	Ö	3	1	12	13 1/2	9040		0.24
M66	0	3	1 1/2	12	14 1/2	11609		0.23

GENER/	AL NOT	ES	$\theta = \frac{2}{2}$	y _{mm} ; K b	R as for	vertical ben	ling	PARAPETS YF 1-		MODEL R	EF.	M1
EP Ref.	$\delta_{\rm s}$	2	$\frac{Im\delta_8}{\rho b^4}$	$\frac{V}{f_{1}d_{4}}$	pr	easured edicted	(0°)	$\frac{2K_Ry_{max}}{10^3b}$	$\left(\frac{\rho b^4}{lm}\right)\frac{10^3}{45s}$	$\left(\frac{d_4}{b}\right)^{13}$		easured redicted
			-		rule ¹	proposed?	5	·			rule ³	proposed
YP	0.01											
YP	0.02	1	1.2.2.2	19222	12222		10 0000 ¹		1.000		1000	1005307
YP 1d	0.03	5	0.206	8.80	1.35		1.120	8.01	6.28		1.27	0.85*
YP 1d	0.04	5	0.274	8.81	1.36		0.731	5.23	4.71		1.11	0.74*
YP 1d	0.05	5	0.343	8.66	1.33		0.578	4.14	3.77		1.10	0.73*
YP	0.06			10.00								-
YP 1d	0.01	0	0.069	10.08	1.55		1.915	13.70	18.85		.73	.49
YP 1d	0.02	0	0.137	9.67	1.49		1.168	8.36	9.43		.89	.59
YP 1d	0.03	0	0.206	9.09	1.40		0,715	5.12	6.28		.82	.55
YP 1d	0.04	0	0.274	8.94	1.38		0.451	3.23	4.71		.69	,46
YP	0.05	0										
YP	0.06	0								_	-	-
YP	0.01	0			1.001							11714
YP 2d	0.02	0	0.137	10.86	1.67		3.222	23.06	9.43		2.45	.91
YP 2d	0.03	0	0.206	10.46	1.61		0.968	6.93	6.28		1.10	.41
YP	0.04	0										100
YP	0.05	0										1.00
YP	0.06	0								-	1.1	1.000
YP	0.01	0			100.44			0.0770.007			1.000	
YP 3d	0.02	0	0.137	10.82	1.66		4.200	30.05	9.43		3.19	.82
YP	0.03	0	100									1.10
YP	0.04	0	-									
YP	0.05	0										
YP	0.06	0										
YP	0.01	0	1.5023	192.94			Sec.	1 2023	3.1		1.03	1 31 -
YP 4d	0.02	0	0.137	9.50	1.46		2.043	14.62	9.43		1.55	.30
YP	0.03	0									1.1	
YP	0.04	0									1.00	
YP	0.05	0										
YP	0.06	0					_				1	1.000
YP	0.01	-5	No Resu	its								
YP	0.02											
YP	0.03											
YP	0.04											
YP	0.05											1000
YP	0.06		_									

² still 6.5

⁵ without 3.0 factor ⁴ with c factor

* Note : ∠ +5°

Test Results v. Current Rules - Torsion - Table M1Y(continued)

Edge	K/d4	Parapet	h/d_4	saffix	¢
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
	1000	P4	1.5	d	1

Mo	del	Units	l'd+	b^{μ}/d_4	b/d _e	(b/d _e) ^{L5}	$m/\rho b^2$	K _R
MEI	•	1	1	3	4 1/2	193	1029	0.41
M2	0	1	$1 \frac{1}{2}$	3	5 1/2	390		0.37
43	0	2	1	7^{1}_{2}	9	2187		0.29
M4	D	2	$1^{-1}/_{2}$	$7 i_2$	10	3162	1	0.27
M15	0	3	1	12	13 1/2	9040	-	0.24
M6	0	3	$1 \frac{1}{2}$	12	14 1/2	11609	-	0.23

GENER/	AL NOT	ES	θ =	<u>2y_{nn}[⊤]</u> ; b	K _R as f	or vertical 1	sending	PARAPETS YP 1a-		MODEL R	EF.	M1
EP Ref.	$\delta_{\mathbf{s}}$	2	$\frac{Im\delta_{S}}{\rho b^{4}}$	$\frac{V}{f_{t}d_{4}}$	pr	easured edicted	(0°)	$\frac{2K_Ry_{max}}{10^3b}$	$\left[\left(\frac{\rho b^4}{Im} \right) \frac{10^3}{4 \tilde{a} s} \right]$	$\left(\frac{d_4}{b}\right)^{1.5}$	pr	edicted
-	0.01				rule	proposed ²					rule ³	proposed
YP YP 1a	.0223	0	0.119	9.35	1.44		.0220	.16	10.89		.01	.01
YP	0.03	~	Gentle.						10000			
YP	0.04		-1.					100				1 J. Com
YP	0.05		_								1.000	1.1
YP	0.06						1					1.11
YP	0.01											
YP 1b	.0223	0	0.119	9.59	1.47		.1430	1.02	10.89		.09	.09
YP 1b	0.03	0	0.160	9.88	1.52		.1140	0.82	8.10		.10	.10
YP 1b	0.04	0	0.214	9.87	1.52		.1000	0.72	6.07		.12	.11
YP 1b	0.05	0	0.267	10.15	1.56		.0690	0.49	4.86		.10	.10
YΡ	0.06	0	10000	13894440			Cherole 17	0.8	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		1.200	1000
YP	0.01			-	8 - 8							
YP	0.02	1.1	-								1.1.1.1	1.1
YP	0.03	1.0	C					1000			1.00	
YP	0.04											
YP	0.05								1			
YP	0.06											
YP	0.01											5 1 1 1 C
YP 1d	.0223	-5	0.119	9.59	1.47		.0990	0.71	10.89		.07	0.05*
YP 1d	0.03		0.160	9.58	1.47		.0780	0.56	8.10		.07	0.05*
YP 1d	0.04		0.214	9.59	1.47		.0730	0.52	6.07		.09	0.06*
YP	0.05											
YP	0.06					-	-			_	-	-
YP	0.01		0.00076	2000	100000		- constant	15.26	1000000		12.20	245
YP 25	.0223	0	0.119	11.00	1.69	1	.1150	0.82	10.89		.08	.05
YP 2b	0.03	0	0.160	11.29	1.74		.0830	0.59	8.10		.07	.04
YP 2b	0.04	0	0.214	11.53	1.77		.0530	0.38	6.07		.06	.04
YP	0.05											
YP	0.06		_							_		-
YР	0.01											
YP	0.02											
YP	0.03											
YP	0.04											
YP	0.05											
YP	0.06			¹ c.f. 6.							-	3.0 facto

³ without 3.0 factor ⁴ with c factor

* Note : 2 -5°

Test Results v. Current Rules - Vertical Bending - Table M3E1

_	-	_	Parapo		-
Edge	K/d ₄	Parapet	h da	suffix	- þ
х	0.05	PI	0.3	a	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
	U.N.O	P4	1.5	d	1

Me	del	Units	lid.	$b^{\ast} \cdot d_{a}$	b/d4	$(b/d_4)^{1,2}$	$m^\prime\rho b^2$	Ka
М1	0	1	1	3	4 1/2	9.5		0.41
M2	0	1	$1^{-1}\!/_{2}$	3	5 1/2	12.9		0.37
M3	•	2	1	7 1/2	9	27.0	748 or (810)	0.29
M4	0	2	1%	$7^{(1)}_{2}$	10	31.6		0.27
M3	0	3	1	12	13 1/2	49.6		0.24
M6	0	3	1^{-1}_{\pm}	12	14 1/2	55.2	_	0.23

	GENERAL reduction for				К, =	$\frac{0.87}{\sqrt{9}} =$	0.290:		PARAPE all P	200000000000000000000000000000000000000		M3
mode	EP Ref.	δs	2	$\frac{m\delta_8}{\rho b^2}$	$\frac{V}{f_4d_4}$		easured redicted proposed ²	$\left(\frac{y_{max}}{d_4}\right)_{s}$	$\frac{K_{0.}y_{max}}{10^{13}d_{4}}$	$\left[\left(\frac{\rho b^2}{m} \right) \frac{10^3}{4\delta s} \left(\frac{d^4}{b} \right)^{13} \right.$		easured redicted
-	XP 2a XP XP XP XP XP XP XP XP	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0									propose
	YP 2a YP F YP F YP I YP I YP I YP YP	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0 0			stable						
lst	YP 2a YP YP YP YP YP YP	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0 0	0.277	9.08	0.98	1.10	.0017	0.5	33.4	.01	.01
st st	YP 2a YP 2a YP 2a YP 2a YP 2a YP 2a YP 2a YP	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0	0.277 0.544 0.831 1.108 1.384	9.25 9.31 9.31 9.31 9.31	1.00 1.01 1.01 1.01 1.01	1.12	.0076 .0042 .0025 .0016 .0010	2.2 1.2 0.7 0.5 0.3	33.4 16.7 11.1 8.4 6.7 5.6	.07 .07 .06 .06 .04	.04 .04 .03 .03 .02
st st st	UP 2a UP 2a UP 2a UP 2a UP 2a UP 2a UP 2a	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0 0	0.300 0.599 0.899 1.199 1.498 1.798	14.13 13.59 13.09 13.09 12.66 12.60	1.53 1.47 1.42 1.42 1.37 1.36	1.72	.0511 .0420 .0327 .2680 .0234 .0201	14.8 12.2 9,5 7.8 6.8 5.8	35.2 17.6 11.7 8.8 7.0 5.9	.42 .69 .81 .89 .97 .98	.29 .48 .57 .62 .68 .69
	U'P 2a U'P 2a U'P 2a U'P 2a U'P 2a U'P 2a U'P 2a	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0 0			stable				0.7	.70	

² c.f. 8.25

⁴ with c factor

Test Results v. Current Rules - Vertical Bending - Table M3E2

EP ref. = Edge / Parapet reference									
Edge	K/d4	Parapet	h'd4	suffix	ģ				
Х	0.05	P1	0.3	a	1/4				
	U.N.O	P2	0.7	b	1/2				
Y	0.20	P3	1.1	c	3/4				
	U.N.O	P4	1.5	d	1				

Mo	del	Units	I'd4	$b^{\bullet} {}^{\bullet} d_{\pm}$	b/d4	$(b/d_1)^{1.5}$	$m^\prime\rho b^2$	Ka
311	0	1	L	3	4 1/2	9.5	· · · · · ·	0.41
342	0	1	$1^{-1}/_{2}$	3	5 1/2	12.9	mer l	0.37
M3	•	2	1/2	7 $^{1}/_{2}$	8	22.6	773 or (780)	0.31
514	0	2	$1^{-1}/_2$	$7\ ^{1}l_{2}$	10	31.6		0.27
M5	0	3	1	12	13 1/2	49.6		0.24
MG	0	3	1 1/2	12	14 1/2	55.2		0.23

- 1	GENERAL reduction for			K _a -	$=\frac{0.87}{\sqrt{8}}=$	0.3076;			PARAPET all P2	222		M3
mode	EP Ref.	δ	2	$\frac{m\delta_{S}}{\rho b^{2}}$	$\frac{V}{f_{y}d_{4}}$	in the second second	rasured edicted	$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_{B}y_{max}}{10^{-3}d_4}$	$\left(\frac{\rho b^2}{m} \right) \frac{10^3}{4\delta s} \left(\frac{d^4}{b} \right)^{1.5}$		casured redicted
-						rule ¹	proposed ²				rule	propose
nd	XP 2a	0.01	0	0.342	16.15	1.75	1.96	.0442	13.6	32.3	.42	.51
Ind	XP _	0.02	0	0.685	15.58	1.68		.0337	10.4	16.2	.64	.78
Ind	XP 7	0.03	0	1.027	15.01	1.62		.0266	8.2	10.8	.76	.92
Ind	XP T	0.04	0	1.370	15.01	1.62		.0217	6.7	8.1	.83	1.01
Ind	XP 📓	0.05	0	1.712	14.44	1.56		.0176	5.4	6.5	.83	1.01
Ind	XP	0.06	0	2.054	13.99	1.51	1.69	.0138	4.2	5.4	.78	.94
Ind	YP 2a	0.01	0	0.342	17.11	1.85	2.07	.0488	15.0	32.3	.46	.32
Ind	YP "	0.02	0	0.685	16.49	1.78		.0395	12.2	16.2	.75	.53
Ind	YP ?	0.03	0	1.027	16.09	1.74		.0320	9.8	10.8	.91	.64
Ind	YP . T	0.04	0	1.370	16.03	1.73		.0262	8.1	8.1	1.00	.70
Ind	YP ⅔	0.05	0	1,712	15.41	1.66		.0236	7.3	6.5	1.12	.79
Ind	YP	0.06	0	2.054	15.18	1.64	1.84	.0195	6.0	5.4	1.11	.78
	YP 2a	0.01	0						19-11-11			
3	YP 🚽	0.02	0									
- 3	YP TO = TP	0.03	0						16 d			
- 8	YP Z	0.04	0									
- 3	YP -	0.05	0									
	YP	0.06	0								-	-
	YP 2a	0.01	0	0.342	17.68	1.91	2.14	.0390	12.0	32.3	.37	.18
	YP 2a vy	0.02	0	0.685	17.23	1.86		.0310	9.5	16.0	.57	_28
2nd	YP 2a ∉	0.03	0	1.027	17.06	1.84		.0247	7,6	10.8	.70	.35
	YP 2a 🚽	0.04	0	1.370	16.54	1.79		.0212	6.5	8.1	.80	.40
	YP 2a	0.05	0	1.712	16.49	1.78		.0172	5.3	6.5	.82	.41
-	YP	0.06	0	2.054	15.97	1.73	1.94	.0142	4.4	5.4	.81	.40
	UP 2a	0.01	0	0.344	15.22	1.65	1.85	.0715	22.0	36,4	.60	.42
	UP 2a	0.02	0	0.692	15.01	1.62		.0575	17.7	18.2	.97	.68
	UP 2a B	0.03	0	1.037	14,49	1.57		.0501	15.4	12.1	1.27	.89
	UP Za S	0.04	0	1.382	14.02	1.51		.0434	13.3	9.1	1.46	1.02
	UP 2a	0.05	0	1.729	14.02	1.51	- States	.0384	11.8	7.3	1.62	1.14
_	UP 2a	0.06	0	2.075	13.55	1.46	1.64	.0336	10.3	6.1	1.69	1.18
	U'P 2a	0.01	0	0.346	14.49	1.57	1.76	.0583	17.9	36.4	.49	.24
	U'P 2a	0.02	0	0.692	14.02	1.51		.0488	15.0	18.2	.82	.40
2nd	UP2a g	0.03	0	1.037	13.55	1.46		.0409	12.6	12.1	1.04	.51
2nd	UP 2a S	0.04	0	1.383	13.55	1.46		.0347	10.7	9.1	1.18	.58
2md	U'P 2a	0.05	0	1.729	13.55	1.46		.0288	8.8	73	1.21	.60
Said	U'P 2a	0.06	0	2.075	13.08	1.41	1.58	.0256	7.9	6.1	1.30	.64

Test Results v. Current Rules - Divergent TI - Table M3E2

EP ref. = Edge / Parapet reference										
Edge	K/d4	Parapet	h/d4	suffix	φ					
х	0.05	PI	0.3	a	1/4					
	U.N.O	P2	0.7	b	1/2					
Y	0.20	P3	1.1	c	3/4					
	U.N.O	P4	1.5	d	1					

Mo	del	Units	i'd4	b^{α}/d_{η}	b/d4	(b/d) ^{1.3}	(b/d) ¹³
М1	0	1	1	3	4 1/2	9.5	193
M2	0	I	1 1/2	3	5 1/2	12.9	390
M3	•	2	1/2	7 1/2	8	22.6	1448
M4	0	2	1 1/2	7 1/2	10	31.6	3162
M5	0	3	1	12	13 1/2	49.6	9040
M6	0	3	1 %	12	14 1/2	55.2	11609

GENERAL	NOTES		vergent To	orsional I	nstability		PARAPETS all P2:	Ĩ	MODEL R Reduced ove		M3
EP Ref.	$\delta_{\rm S}$	2	$\frac{Im\delta_{S}}{\rho b^{4}}$	V _e m f _t b	predicted	or		pr	asured edicted	3.3	
XP	0.01	0		_		-		(5.0)	(1.5)	-	-
XP 2a Z	0.02	0	0.080	4.00	5 (or 1.5) ⁸⁴	3.3		.80	2.67	1.21	
KP 2a Z	0.04 0.05	0 0	0.160	4.22	(NA - not applied here since	3.3		.84	2.81	1.28	
XP 2a	0.06	0	0.239	4.52	$h/d_{a} > 4$)	3.3		.90	3.01	1.37	
YP YP 2a YP 2	0.01 0.02 0.03	0 0 0	0.080	3,49	5 (or 1.5)	3.3		.70	2.33	1.06	
YP 2a T	0.05	0	0.160	3.77	5 (or 1.5)	3.3		.75	2.51	1.14	
YP 2a	0.06	0	0.239	3.99	5 (or 1.5)	3.3	1 1	.80	2.66	1.21	
YP	0.01	0	101000	1							
VP 2a 3	0.02	0									
	0.03	0									
YP 2a ਤੋਂ YP	0.04	0									
YP 2a	0.06	0									
YP	0.01	0	_								-
YP 2a vo	0.02	0	0.080	3.72	5 (or 1.5)	3.3		.74	2.48	1.13	
YP 2a 🚽	0.04 0.05	0	0.160	3.98	5 (or 1.5)	3.3		.80	2.65	1.21	
YP 2a	0.06	0	0.239	4.22	5 (or 1.5)	3.3		.84	2.81	1.28	
UP UP 2a UP P	0.01 0.02	0	0.059	3.82	5 (or 1.5)	3.3		.76	2.54	1.16	
UP 2a 2 UP 2a 2 UP	0.03	0	0.118	4.11	5 (or 1.5)	3.3		.82	2.74	1.25	
UP 2a	0.05	0	0.176	434	5 (or 1.5)	3.3		.87	2.89	1.32	
UP	0.00	0	0.1.0		2 (01 4-0)	d'al		.0.	£-07	1.00	-
U'P 2a	0.02	0	0.059	3.72	5 (or 1.5)	3.3		.74	2.48	1.13	
UPP 2a 3	0.04	0	0.118	3.97	5 (or 1.5)	3.3		.79	2.65	1.20	
U'P 2a	0.06	0	0.176	4.18	5 (or 1.5)	3.3		.84	2.79	1.27	

Test Results v. Current Rules - Torsion - Table M3E2

EP ref. = Edge / Parapet reference										
Edge	K/d4	Parapet	h/d4	suffix	ф					
х	0.05	Pl	0.3	a	1/4					
	U.N.O	P2	0.7	b	1/2					
Y	0.20	P3	1.1	c	3/4					
	U.N.O	P4	1.5	d	1					

Mo	del	Units	lid4	$b^{\bullet} \mathrm{i} \mathrm{d}_4$	$\mathbf{b}^{i}\mathbf{d}_{i}$	(b\d_) ¹³	$m^\prime\rho b^2$	Ka
MI	Q.	1	1	3	4 1/2	193		0.41
M2	ð,	1	1.9_2	3	5 1/2	390	-	0.37
М3	•	2	η_{\pm}	7 1/3	8	1448	4156	0.29
M4	0	2	$1^{-1}/_{2}$	7 1/2	10	3162	1005106010	0.27
M5	0	3	1	12	13 ¹ / ₂	9040	-	0.24
M6	0	3	1 1/2	12	$14^{-1}/_{2}$	11609		0.23

	GENERAL	NOTES	0	$=\frac{2y_{max}}{b}$	- ; K _R ;	is for ve	rtical bendia	e	PARAPET all P		MODEL RI Reduced over	CE. rhang	M3
mode	EP Ref.	δs	2	$\frac{Im\delta_5}{\rho b^4}$	$\frac{V}{f_{T}d_{4}}$		easured edicted	$\left(\theta^{e}\right)_{e}$	$\frac{2K_By_{max}}{10^{-3}b}$	$\left(\frac{\rho b^4}{Im}\right) \frac{10^3}{4\delta s}$	$\left(\frac{d_4}{b}\right)^{33}$		easured redicted
=	1.1.1.2.2.2.2.2.2.4.2	100.001				rule ¹	proposed2	- J8	A 31325-98			rule ³	proposed
st	XP 2a	0.01	0	0.029	8.95	0.97	1.08	.4680	2.51	6.02		.42	.51
st	XP 2a	0.02	0	0.057	8.95	0.97	1.08	.3530	1.90	3.01		.63	.76
lst	XP 2a d	0.03	0	0.086	8.69	0.94	1.05	.2650	1.42	2.01		.71	.86
Ist	XP 7	0.04	0	10000	25000	11110-000	100000	1.000	1.24.40	1.50			
lst	XP 🖻	0.05	0							1.20			
lst	XP	0.06	0			1000				1.00			
lst	YP 2a	0.01	0	0.029	8.95	0.27	1.08	.4400	2.36	6.02		.39	.27
lst	YP m	0.02	0										
0.00	YP 🤤	0.03	0										
	YP T	0.04	0										
	11 .	0.05	0										
-	YP	0.06	0						-			-	-
	YP	0.01	0				2						
	YP 3	0.02	0										
	YP PO-TPA	0.03	0										
1st	YP T	0.04	0										
	YP	0.05	0										
	YP	0.06	0	0.029	0.77	0.07	1.05	.5510	2.96	6.02		.49	.24
	YP 2a	0.01	0	0.029	8.63	0.93	1.05	.2010	4.90	0.02		1.00	
	YP si	0.02	0										1000
	YP STO = THOM	0.03	0										-
	YP 3	0.04	0										
	YP YP	0.05	0										
1st 1st	UP	0.06	0		-	-	2			-		-	
1st	UP	0.02	0									1.1	
1st	UP 2	0.02	0				Stable						
lst	UP III	0.04	0										
1st	UP	0.05	0										
1st	UP	0.06	0				1						
1st	UP	0.01	0				1					-	1.1.1.1.1.1
1st	THE	0.02	0			100							1.00
1st	UP Dela	0.03	0	1.1			Stable						
1st	UP H	0.04	0				1						1.00
	UP	0.05	0										
	U'P	0.06	0)						

² c.f. 8.25

⁴ without 3.0 facto ⁴ with c factor

Test Results v. Current Rules - Vertical Bending - Table M3E3

EP ref. = Edge / Parapet reference										
Edge	K/d ₄	Parapet	h/d4	suffix	ó					
х	0.05	P1	0.3	a	1/4					
	U.N.O	P2	0,7	b	1/2					
Y	0.20	P3	1.1	c	3/4					
	U.N.O	P4	1.5	d	1					

Model	Units	Fd4	b^{α}/d_{α}	h/d ₆	$(b/d_4)^{1.5}$	$m \prime \rho b^2$	K _R
M1 👳	1	1	3	4 1/2	9.5		0.41
M2 0	1	$1^{-1}/_{2}$	3	5 1/2	12.9		0.37
M3 •	2	\$1/2	7 1/2	8	22.6	774	0.31
M4 0	2	$1^{-1}/_{2}$	7 1/2	10	31.6		0.27
M5 0	3	1	12	13 ¹ / ₂	49.6		0.24
M6 0	3	1 1/2	12	14 1/2	55,2	·	0.23

MODEL REF.	3.50
Reduced overhang	M3
	Reduced overhang

mode	EP Ref.	δ	2	$\frac{m\delta_S}{\rho b^2}$	V f,d,	-	easured redicted	ymax d.	Kaymax 10 ⁻³ d ₄	$\left(\frac{\rho b^2}{m}\right) \frac{10^3}{45s} \left(\frac{d^4}{b}\right)^{1.3}$		easured edicted
8						rule ¹	proposed ²	C. 7	145155		rule ³	proposed
2nd 2nd	XP 2b XP 2b XP 2b	0.01 0.02 0.03	0 0 0	0.342 0.685 1.027	16.32 16.32 15.58	1.76 1.76 1.68	1.97	.0954 .0785 .0657	29.3 24.1 20.2	32.3 16.1 10.8	.91 1.50 1.87	.67 1.11 1.39
2md 2md	XP 2b Z XP 2b	0.04 0.05	0	1.370 1.712	15.23 15.17	1.65 1.64		.0576 .0492	17.7	8.1 6.5	2.19	1.62
2md	XP 2b YP YP YP YP YP YP YP	0.06 0.01 0.02 0.03 0.04 0.05 0.06	0	2.054	14.78	1.60	1.79	.0438	13.5	5.4	2.50	1.85
Construction of the	YP YP YP YP YP YP	0.01 0.02 0.03 0.04 0.05 0.06					2					
2nd 2nd 2nd 2nd	YP 2b YP 2b YP 2b YP 2b YP 2b YP 2b YP 2b	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	0.342 0.685 1.027 1.370 1.712 2.054	15.92 15.41 14.89 14.84 14.38 14.33	1.72 1.66 1.61 1.60 1.55 1.55	1.93	.0890 .0731 .0615 .0545 .0461 .0412	27.6 22.5 18.9 16.8 14.2 12.7	32.3 16.1 10.8 8.1 6.5 5.4	.85 1.40 1.75 2.07 2.18 2.35	.33 .55 .69 .81 .85 .92
-110	UP 2a UP 2a UP 2a UP 2a UP 2a UP 2a	0.01 0.02 0.03 0.04 0.05 0.06		2000	1433	1.00				117		
	UP' 2a UP' 2a UP' 2a UP' 2a UP' 2a UP' 2a	0.01										

² c.f. 8.25

4 with c factor

Test Results v. Current Rules - Divergent TI - Table M3E3

EP	ref. =	Edge / I	Parapo	t refere	nce
Edge	K/d ₄	Parapet	h'd4	suffix	φ.
х	0.05	P1	0.3	a	1/4
	U.N.O	P2	0.7	ъ	1/2
Y	0.20	P3	1.1	c	3/4
	U.N.O	P4	1.5	d	1

Mo	del	Units	1-d ₁	$b^{\ast}d_4$	b d ₄	(b/d)1.5	(b/d) ¹³
MI	0	1	1	3	4 1/2	9.5	193
M2	0	1	1 1/2	3	5 1/2	12.9	390
М3	•	2	η_{\pm}	7 1/2	8	22.6	1448
M4	0	2	$1\frac{\eta_2}{2}$	7 ¹ / ₂	10	31.6	3162
M5	0	3	1	12	13 1/2	49.6	9040
M6	0	3	1 1/2	12	14 1/2	55.2	11609

GENERAL	NOTES		ergent To	orsional Ins	tability		PARAPETS additional	25	MODEL R			
EP Ref.	δs	2	$\frac{1m\delta_{S}}{\rho b^{4}}$	$\frac{V_{\mu tt}}{f_{T}b}$	predicted			pr	asured edicted	3.3	-	
XP	0.01	0				-		(5.0)	(1.5)	-	-	
	0.02	0	0.057	2.91	5 (or 1.5)	33		.58	1.94	.88	1 - 1	
XP 26 10 XP 26 2 XP 26 2 XP 26 2	0.04	0	0.114	3.51	5 (or 1.5)	3.3		.70	2.34	1.06	18.2	
XP 2b	0.06	0	0.171	3.78	5 (or 1.5)	3.3		.76	2.52	1.15		
YP	0.01	0										
YP m	0.02	0										
YP 🚆	0.03	0					1 1				1.2	
YP P	0.04	0					1 1					
3.4	0.05	0										
YP YP	0.06	0	-			-				-	-	
100	0.01	0										
YP -*PA	0.02	0										
YP =	0.04	0					1 1					
YP 3	0.05	0					1 1					
YP	0.06	0										
YP	0.01	0	-		12. 11. 410				0/10/10/10			
	0.02	0	0.057	3.16	5 (or 1.5)	3.3	1 1	.63	2.11	.96	1.00	
YP 2a vi YP 0	0.03	0								1.1.1	1.1.1	
YP 2a	0.04	0	0.114	3.57	5 (or 1.5)	3.3		.71	2.38	1.08		
YP =	0.05	0								1.1.1		
YP 2a	0.06	0	0.171	3.78	5 (or 1.5)	3.3		.76	2.52	1.15		
UP	0.01	0										
UP	0.02	0										
UP IV	0.03	0										
UP 🚆	0.04	0										
UP -	0.05	0										
UP	0.06	0				-	-		_	-	-	
U'P	0.01	0										
UP 5	0.02	0										
UP Dala	0.03	0									1.8	
U'P E	0.04	0										
U'P	0.05	0				1					1	
U'P	0.06	0					-				-	

Test Results v.	Current	Rules -	Torsion -	M3E3
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EP	ef. = E	dge / P	arape	t refere	ence
Edge	K/d4	Parapet	h/d ₄	suffix	ģ
х	0.05	Pl	0.3	a	4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
	U.N.O	P4	1.5	d	1

Mod	fel	Units	I'd ₄	6*/d4	bid,	(b/d ₄) ^{1.5}	m/pb [‡]	KR
M3	0	1	1	3	4 1/2	193		0.41
M2	0	1	$1\frac{1}{2}$	3	5 1/2	390		0.37
М3	•	2	1/2	7 1/2	8	1448	4127	0.31
M4	0	2	1 1/2	7 1/2	10	3162		0.27
M5	0	3	1	12	13 1/2	9040		0.24
M6	0	3	1 1/2	12	14 1/2	11609		0.23

	GENERAL	NOTES	0	= 2y_man b	- ; K _a a	s for ve	rtical bendin	e	PARAPET addition			<u>L REF.</u> d Overhan	g	M3
node	EP Ref.	δs	2	$\frac{Im\delta_3}{\rho b^2}$	$\frac{V}{f_{1}d_{4}}$		casured	$\left(\theta^{a} \right)_{s}$	$\frac{2K_Ry_{max}}{10^{2}h}$	(pb ⁴) Im	$\frac{10^{1}}{458}$	$\left(\frac{d_4}{b}\right)^{1/2}$	pr	easured redicted
-					000	rule1	proposed ²	the second se			3 3	- L	rule1	propose
st	XP 2b	0.01	0	0.029	9.07	0.98	1.10	.9630	5.17	6.0	6		.85	.63
st	XP 2b	0.02	0	0.057	8.84	0.96	1.07	.6160	3.31	3.0			1.09	.81
st	XP 2b	0.03	0	0.086	8.87	0.96	1.08	.4500	2.42	2.0			1.20	.89
st	XP 2b	0.04	0	0.114	8.72	0.94	1.06	.2790	1.50	1.5	1		.99	.73
st	XP 🖻	0.05	0							1.2				10.0
st	XP	0.06	0	1000		5				1.0	1	1		
	YP	0.01	0										1.1	-
	YP m	0.02	0											10.0
	YP 0	0.03	0	1										1.1
	YP 3	0.04	0											
		0.05	0	6 1										
-1	YP	0.06	0										_	_
	YP	0.01	0	3										100
	YP -	0.02	0											
	17 TY 17	0.03	0											1.00
	YP 🚽	0.04	0											
	YP 🗳	0.05	0											
	YP	0.06	0			1								
lst	YP 2b	0.01	0	0.029	9.16	0.99	1.11	1.151	6.18	6.0	16		1.02	.40
lst	VP2b w	0.02	0	0.057	9.01	0.97	1.09	0.737	3.96	3.0	3		1.31	.51
lat	YP 2b of YP 2b of	0.03	0	0.086	8.87	0.96	1.08	0.462	2.48	2.0	02		1.23	.48
lst		0.04	0	1.00										
lst	YP at	0.05	0										1.1.1	
lst	YP	0.06	0	1.000		1						1.5		
	UP	0.01	0											
	UP	0.02	0											
	UP Defig	0.03	0											
	UP 📲	0.04	0											1.000
	UP	0.05	0											
	UP	0.06	0			1				-				
	UP	0.01	0											
	1110	0.02	0											
	UP 10 22 23	0.03	0							1				
	UP 🖁	0.04	0											
	UP	0.05	0											
	UP	0.06	0			-			1.00					

¹ c.f. 9.25 ² c.f. 8.25 3 without 3.0 factor

Futher Test Results v. Current Rules - Vertical Bending - Table M3F1

Edge	K/d.	Parapet	h/d4	suffix	d.
X	0.05		0.3	a	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
	U.N.O	P4	1.5	d	1

Mo	del	Units	ld,	b^{ϕ}/d_{ϕ}	bid,	$\left(b'd_4\right)^{1,2}$	m/pb ²	Ka
MI	0	1	1	3	4 1/2	9.5	-	0.41
342	0	1	$1 \frac{1}{2}$	3	5 1/2	12.9		0.37
313	•	2	17.4	7 ¹ / ₂	7 1/2	20.5	816 or (709)	0.32
M4	0	2	$1^{-1}/_{2}$	7 ¹ / ₂	10	31.6		0.27
М5	0	3	1	12	$13^{-1}/_{2}$	49.6		0.24
M6	0	3	1 1/2	12	14 1/2	55.2		0.23

GENERAL NOTES	0.87	PARAPETS	MODEL REF.	3.50
reduction for rough flow ,	$K_{\rm R} = \frac{1}{\sqrt{7 \sqrt{2}}} = 0.318$	P2a, P2b	further reduced overhang.	M3

mode	EP Ref.	Edge d	δs	4	$\frac{m\delta_S}{\rho b^2}$	V Ed.		easured redicted	(Ymax d,	Kaymax 10 ⁻³ d,	$\left[\frac{pb^2}{m}\right] \frac{10^3}{45s} \left[\frac{d_4}{b}\right]^{1.5}$		easured vdicted
=		2	12		1	100	rule ¹	proposed ²	- 18			rule3	proposed
2nd	YP 2a		0.01	0	0.398	16.83	1.82	2.04	.0566	18.0	30.6	.59	.41
2md	YP 2a	0.3	0.02	0	0.795	16.25	1.76		.0426	13.5	15.3	.89	.62
2nd	YP 2a	° I	0.03	0	1.193	15.45	1.67		.0319	10.1	10.2	.99	.69
2nd	YP 2a	lod.	0.04	0	1,590	15.45	1.67		.0265	8,4	7.7	1,09	.76
2md	YP 2a	3	0.05	0	1.988	14.87	1.61		.0227	7.2	6.1	1.18	.83
2nd	YP 2a		0.06	0	2.385	14.36	1.55	1.74	.0171	5.4	5.1	1.07	.75
1	YP		0.01					· · · · · · · · · · · · · · · · · · ·		1			1
11	YP		0.02					1 0					
19	YP		0.03										
	YP .		0.04		1 1								
	YP		0.05		1.1								
	YP		0.06										1000
2nd	YP 2a		0.01	0	0.398	17.52	1.89	2.12	.0652	20.7	30.6	.68	.34
2nd	YP 2a	0.5	0.62	0	0,795	16.65	1.80		.0511	16.2	15.3	1.06	.53
2nd	YP 2a	ĩ	0.03	0	1.193	16.07	1.74		.0413	13.1	10.2	1.29	.64
2nd	YP 2a	b'd.	0.04	0	1.590	15.79	0.17		.0349	11.1	7.7	1.44	.72
2nd	YP 2a	-	0.05	0	1.988	15.51	1.68	1.1	.0299	9.5	6.1	1.56	.77
2nd	YP 2a		0.06	0	2.385	15.51	1.68	1.88	.0249	7.9	5.1	1.55	.77
2nd	YP 2b		0.01	0	0,398	16.14	1.74	1.95	.0898	28.6	30.6	.93	.36
2nd	YP 2b	w.	0.02	0	0.795	15.62	1.69	0.2105-0	.0706	22.4	15.3	1.47	.58
2nd	YP 2b	0	0.03	0	1.193	14.76	1.60		.0582	18.5	10.2	1.81	.71
2nd	YP 2b	liva.	0.04	0	1.590	14.53	1.57		.0503	16.0	7.7	2.08	.82
2nd	YP 2b	-	0.05	0	1.988	14.24	1.54		.0451	14.3	6.1	2.35	.92
2nd	YP 2b		0.06	0	2.385	13.96	1.51	1.69	.0398	12.7	5.1	2.48	.97
1	P		0.01							con success a			
	P		0.02										
	P		0.03										
	P		0.04										
	P		0.05										
-	P		0.06		L	1			-			1.1	
2md	U'P 2a		0.01	0	0.394	15.76	1.84	2.06	.0774	25.7	35.2	.73	.36
2nd	UPP 2a		0.02	0	0.788	15.23	1.78		.0657	21.8	17.6	1.24	.61
2md	UPP 2a	0.10	0.03	0	1.182	14.71	1.72		.0554	18.4	11.7	1.57	.78
2mi	U'P 2a	Edge	0.04	0	1.576	14.45	1.69		.0492	16.3	8.8	1.86	.92
2md	UPP 2a	-	0.05	0	1.969	15.24	1.66		,0450	14.9	7,0	2.13	1.05
2nd	U'P 2a		0.06	0	2.363	14.24	1.66	1.86	.0407	13.5	5,9	2.29	1.13

¹ c.f. 9.25 ² c.f. 8.25 3 without 3.0 factor

4 with c factor

Further Test Results v. Current Rules - Divergent TI - Table M3F1

EP	ref. = H	Edge / P	arape	t refere	nce
Edge	K/d ₄	Parapet	h/d ₄	suffix	ф
Х	0.05	P1	0.3	a	174
	U.N.O	P2	0.7	ь	1/2
Y	0.20	P3	1.1	с	3/4
	U.N.O	P4	1.5	d	1

Model	Units	۱d,	$b^{\mu}d_{4}$	h'd _i	(b/d) ¹³	(b/d) ^{3.5}
мі 🕤	1	1	3	4 1/2	9.5	193
M2	1	$1 \frac{1}{2}$	3	5 1/2	12.9	390
мз •	2	174	7 1/2	7 1/2	20.5	1155
M4 o	2	$1^{-1}/_{2}$	7 1/2	10	31.6	3162
M5 0	3	1	12	13 1/2	49.6	9040
M6 0	3	1 %	12	14 1/2	55.2	11609

GENERAL NOTES	PARAPETS	MODEL REF.	3.50
Divergent Torsional Instability	P2a, 2b	further Reduced overhang	M3
	The second se	2000 co. 2000 (2200 P.	

				$\underline{Im\delta_{S}}$	$\underline{v_{i\pi}}$			measured		
EP Ret	۲.	$\delta_{\rm S}$	2	pb4	frb	"predicted	or <u>3.3</u>	predicted	3.3	
	-		-					(5.0) (N/A)		
XP XP 2b		0.01	0	.0815	3.70	5.0	3.3	.74	1.12	
XP 20	= 0.3	0.02	0	.0815	3.70	5.0	3.5			
XP 2b	ī,	0.05	0	.1630	3.97	5.0	3.3	.79	1.20	
XP	Fold,	0.05	0	.1000		0.0				
XP 2b		0.06	0	.2445	4.22	5.0	3.3	.84	1.28	
YP	-	0.01	0							
YP		0.02	0							
YP		0.03	0							
YP	1	0.04	0							
YP	. 1	0.05	0						1 1	
YP	_	0.06	0						-	_
YP		0.01	0							
YP 2a	- 0.5	0.02	0	.0815	3.56	5.0	3.3	.71	1.08	
YP	ĩ	0.03	0							
YP 2a	Pod.	0.04	0	.1630	3.81	5.0	3.3	.76	1.15	
YP		0.05	0	.2445	4.04	5.0	3.3	.81	1.22	
YP 2a YP	-	0.00	0	.2443	4.04	3.0	3.5	-01	1	_
		0.02	0	.0838	3.22	5.0	3.3	.64	.98	
YP	0.5	0.03	0							
	lid4=	0.04	0	.1675	3.77	5.0	3.3	.75	1.14	
YP	2	0.05	0	1353	2043.1			1228	1.000	
YP 2b		0.06	0	.2513	4.10	5.0	3.3	.82	1.24	
UP		0.01	0							
UP		0.02	0							
UP		0.03	0							
UP		0.04	0							
UP		0.05	0							
UP	_	0.06	0			5 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 -				-
YP		0.01	0	10.20	14.42					
YP 2a	5	0.02	0	.0774	3.04	5.0	3.3	.61	.92	
YP	Edge U'	0.03	0		10			.71	1.07	
YP 2a	E	0.04	0	.1547	3.53	5.0	3.3	.74	1.07	
YP		0.05	0	.2321	3.78	5.0	3.3	.76	1.14	
YP 2a	_	0.06	0	-2321	3.78	5.0	3.3	+10	4.24	

Futher Test Results v. Current Rules - Torsion - Table M3F1

EP r	ref. = E	dge / Pa	arape	t refere	nce
Edge	K/d ₄	Parapet	h/d ₄	suffix	φ
х	0.05	PI	0.3	a	1/4
	U.N.O	P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
	U.N.O	P4	1.5	d	1

Model	Units	i'd.	b^{ϕ}/d_{a}	b/d ₄	$(h/d_{4})^{1.5}$	$m/\rho b^2$	Ka
MI O	1	1	3	4 1/2	193		0.41
M2 O	1	1%	3	5 1/2	390		0.37
M3 •	2	٩4	7 1/2	7 1/3	1155	4608 ar 3289	0.32
M4 🖸	2	$1 \ ^{t}\!\!/_{2}$	7 1/2	10	3162	Page 2	0.27
M5 0	3	1	12	13 1/2	9040		0.24
M6 0	3	$1^{1}\!\!/_{2}$	12	14 1/1	11609		0.23

-NOTE :

	GENER	ALY	NOTES		$\theta = \frac{2y_m}{b}$	т — ; К,	as for	vertical bend	ing	PARAPET P2a, I	0.000	MODEL REF.	werhang.	M3
mode	EP Ref.	Edge d	$\delta_{\rm S}$	2	$\frac{Im\delta_{8}}{\rho b^{4}}$	$\frac{V}{f_{I}d_{a}}$	p	easured redicted	(0°)	$\frac{2K_{ik}y_{max}}{10^{-3}b}$	(pb ⁴)	$\left \frac{10^3}{45s}\left(\frac{d_4}{b}\right)^{1.5}\right $		edicted
		-						proposed ⁷					rule ³	proposed
	YP 2a YP 2a YP YP YP YP	$bM_s = 0.3$	0.01 0.02 0.03 0.04 0.05 0.06	0 0 0 0 0	0.040	8.55 8.19	0.92	1.04 0.99	0.451 0.207	2.52 1.16	5.4 2.7 1.8 1.3 1.0 0.9	1 1 6 9	.46 .43	.32 .30
ALL STATES AND	ҮР ҮР ҮР ҮР ҮР ҮР		0.01 0.02 0.03 0.04 0.05 0.06		1 24 02 1									
	YP 2a		0.01	0	0.040	8.55	0.92	1.04	0.508	2.84	5.4	3	.52	.26
Number of the second se	YP 2a YP 2a YP YP YP	$k/d_s = 0.5$	0.02 0.03 0.04 0.05 0.06	0 0 0 0	0.080 0.120	8.55 9.24	0.92 1.00	1.04 1.12	0.198 0.061	1.11 0.34	2.7	-	.41 .19	.20 .09
-	YP 2b	-	0.01	0	0.040	8.78	0.95	1.06	1.190	6.64	5.4	1	1.22	.48
	YP 2b	6	0.02	0	0.080	8.59	0.93	1.04	0.730	4.08	2.7		1.50	.59
	YP 2b YP YP YP	$l_{\rm b}/d_{\rm s} = 0.5$	0.03 0.04 0.05 0.06	0 0 0	0.120	8.60	0.93	1.04	0.299	1.67	1.8		.92	.36
	YP		0.01											
	¥Р ¥Р ¥Р ¥Р ¥Р		0.02 0.03 0.04 0.05 0.06											
	YP 2a		0.01	0	0.039	7.87	0.92	1.01	.442	2.56	7.6	0	.34	.17
	YP 2a YP YP YP	Edge U'	0.02 0.03 0.04 0.05	0 0 0	0.077	8.28	0.97	1,06	.216	1.25	3.8	5 J.	.33	.16
	YP		0.05	0					1.					

¹ c.f. 9.25 ² c.f. 8.25

³ without 3.0 factor ⁴ with c factor

Test Results v. Current Rules - Divergent TI - Table M3X & Y

EP	ref. =	Edge / P	arape	t referen	ice
Edge	K/d4	Parapet	h/d ₄	suffix	φ
х	0.05	P1	0.3	a	1/4
		P2	0,7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Model	Units	I'd _e	b^{μ}/d_{4}	h/d ₄	$(b/d)^{\ddagger}$	m/pb ²
M1 .	1	1	3	4 1/2	20	
M2 0	1	1.9_{2}	3	5 1/2	30	
M3 🕤	2	1	7 1/2	9	81	
M4 o	2	1 %	7 1/+	10	100	
M5 o	3	1	12	$13^{-1}/_{2}$	182	1.00
M6 0	3	1 1/2	12	14 1/2	210	

GENERA	L NOT	ES	Initia	lly C _t →	1.0			N & Y: 2a,b;	3a,b	MODEL R	EF.	M3
EP Ref.	δs	2	$\frac{Im\delta_{3}}{\rho b^{4}}$	$\frac{V_{f}}{f_{T}d}$		Τ	or			asured edicted	3.3	
	-								(5.0)	(1.3)		1.1
XP 2a	0.02	0	.0569	4.90	5 (or 1.3)		3.3		.98	3.8	1.48	
YP 2a	0.02	0	.0569	4.71	5 (or 1.3)		3.3		.94	3.6	1.42	1
XP 2a	0.04	0	.1138	5.22	5 (or 1.3)		3.3		1.04	4.0	1.58	1.11
YP 2a	0.04	0	.1138	4.92	5 (or 1.3)	_	3.3		.98	3.8	1.48	-
XP 2a	0.06	0	.1707	5.35	5 (or 1.3)		3.3		1.07	4.1	1.62	
YP 2a	0.06	0	.1707	5.10	5 (or 1.3)	_	3.3		1.02	3.9	1.55	
XP 2b	0.02	0	.0569	2.90	5 (or 1.3)		3.3		_58	2.2	.88	
YP 2b	0.02	0	.0569	3.13	5 (or 1.3)		3.3		.63	2.4	.95	
XP 2b	0.04	0	.1138	3.13	5 (or 1.3)		3.3		.63	2.4	.95	
YP 2b	0.04	0	.1138	3.48	5 (or 1.3)		3.3		.70	2.7	1.06	
XP 2b	0.06	0	.1707	3.70	5 (or 1.3)		3.3		.74	2.8	1.12	
YP 2b	0.06	0	.1707	3.69	5 (or 1.3)		3.3		.74	2.8	1.12	
XP 3a	0.02	0	.0569	4.55	5 (or 1.3)		3.3		.91	3.5	1.38	
XP 3a	0.02	0	.0569	5.08	5 (or 1.3)	-	3.3		1.02	3.9	1.55	
XP 3a	0.04	0	.1138	4.72	5 (or 1.3)		3.3		.94	3.6	1.42	
XP 3a	0.04	0	.1138	5.34	5 (or 1.3)		3.3		1.07	4.1	1.62	1.1
XP 3a	0.06	Û	.1707	5.01	5 (or 1.3)		3,3		1.00	3.8	1.52	
XP 3a	0.06	0	.1707	5.52	5 (or 1.3)	- 1	3.3		1.10	4.2	1.67	
XP 3b	0.02	0	.0569	2.22	5 (or 1.3)		3.3		.44	1.7	.67	
YP 3b	0.02	0	.0569	2.85	5 (or 1.3)	- 1	3.3	1 1	.57	2.2	.86	
XP 3b	0.04	0	.1138	2.80	5 (or 1.3)		3.3		.56	2.1	.85	1.11
YP 3b	0.04	0	.1138	3.37	5 (or 1.3)		3.3		.67	2.6	1.02	an the l
XP 3b	0.06	0	.1707	3.01	5 (or 1.3)		3.3		.60	2.3	.91	1.0
YP 3b	0.06	0	.1707	3.89	5 (or 1.3)		3.3	1	.78	3.0	1.18	
XP	0.02	-				-						
YP	0.02					_						
XP	0.04	-				-						
YP	0.04	1										
XP	0.06	+	-				-					
YP	0.06							1 1				
XP 2b	0.02	0	.0569	3.47	5 (or 1.3)	-	3.3		.69	2.7	1.05	
YP	0.02	0		0.000	- (- · · · · · · /	lo			18366	22.0	1 200	
XP 2b	0.04	0	.1138	3.76	5 (or 1.3)		3.3	1 1	.75	2.9	1.14	Turbulen
YP	0.04	0			2 (or 10)	=			100	-576		Flow
XP 2b	0.04	0	.1707	4.12	5 (or 1.3)	Turbulent Flow	3.3		.82	3.2	1.24	
YP 26	0.06	0	1.10/		o (0, 1-3)	ιĒ.	diat					

Test Results v. Current Rules - Torsion - Table M3X & Y

EP	ref. =)	Edge / P	arape	t refere	nce
Edge	K/d4	Parapet	h/d ₄	suffix	ф
х	0.05	P1	0.3	a	1/4
· .	2.000	P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
		P4	1.5	d	1

Model	Units	1/d _a	b*/d4	b/d_4	$(b/d_i)^{1.5}$	m'pb ²	Ka
MI O	1	1	3	4 1/2	193		0.41
M2 0	1	$1^{-1}/_{2}$	3	5 1/2	390		0.37
мв •	2	1	7 1/2	9	2187	6233	0.29
M4 🗆	2	$1^{-1}/_{2}$	$7^{-1}/_{2}$	10	3162	-	0.27
№5 о	3	1	12	$13^{-1}\!/_{\pm}$	9040		0.24
M6 0	3	1 1/2	12	14 1/2	11609	_	0.23

GENERAL NOTES	PARAPETS	MODEL REF.	3.00
$\theta = \frac{2y_{max}}{h}$; K _R as for vertical bending	X & Y : P2, 3, 4		M3

EP Ref.	δs	2	$\frac{Im\delta_{S}}{\rho b^{4}}$	V frd.		edicted	(0°)	$\frac{2K_Ry_{max}}{10^{-3}h}$	$\left(\frac{pb^4}{lm}\right)\frac{10^3}{4\delta s}\left(\frac{d_4}{b}\right)^{3.3}$		easured edicted
	0.5	-			rule	proposed ²	61	100.0	() ()	rule ³	proposed
YP 2a	.0138	0	.0393	8.45	0.91	1.02	.460	2.33	2.91	.80	.71
YP 2a	0.02	0	.0569	8.18	0.88	0.02021	.280	1.42	2.00	.71	.63
YP 2a	0.03	0	.0854	7.96	0.86	0.96	.183	0.93	1.34	.69	.61
YP	0.04										
YP	0.05										
XP 2a	.0106	0	All S	table			All	Stable			
YP 2b	.0138	0	Sta	ble			the second se	table			
YP	0.02	11			NB A	so stable in			NB All M3X		
YP	0.03				Turb	ulent Flow			Results STABLE		
YP	0.04				A (M	(3Xp2b)			2000 - 00 CONCORE VI		
YP	0.05				1						
XP 2b	.0106	0	All S	table			All	Stable			
YP 3a	.0138	0	.0393	8.14	0.88	0.96	.353	1.79	2.91	.61	.43
YP	0.02										
YP	0.03										
YP	0.04										
YP	0.05	200			-					-	-
XP 3a	.0106	0	All S	table			All	Stable			
YP 3b	.0138	0	Sta	ble			S	table		12.000	
YP	0.02						1 11				
YP	0.03		1								
YP	0.04										
YP	0.05						2 0				
XP 3b	.0106	0	All S	table			All	Stable	Second Second Second	100	
YP 2d	.0138	0	Sta	ble			s	table			
YP	0.02		1 1				1				
YP	0.03										
YP	0.04										
YP	0.05	1						desautro.			
XP 24	.0106	0	All S	table			AB	Stable			
YP 3d	.0138	0	.0393	9.15	0.99	1.11	.596	3.02	2.91	1.04	.27
YP	0.02	0	.0569	8.65	0.94	A 1994 A 1995	.357	1.81	2.00	.90	.23
YP	0.03	0	.0854	8.42	0.91	1.02	.250	1.27	1.34	.94	.24
YP	0.04				1000	03000		10.6564	· · · · · · · · · · · · · · · · · · ·	1	- 22
YP 4d	.0138	0	.0393	7.64	0.83	0.93	.338	1.71	2.91	.59	.16
YP	0.02		·								-

¹ c.f. 9.25 ² c.f. 8.25 3 without 3.0 factor

4 with c factor

Test Results v. Current Rules - Vertical Bending - Table M3X

EP r	ef. = I	Edge / P	arape	t refere	nce
Edge	K/da	Parapet	h/d,	suffix	φ.
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Model	Units	lid4	$b^{\mathbf{x}/d_4}$	b/d4	$(b/d_4)^{1.5}$	$m^\prime\rho b^7$	Ka
MI 0	1	1	3	4 1/2	9.5		0.41
M2 0	1	$1^{-1} \tilde{\gamma}_2$	3	5 1/2	12.9		0.37
M3 .	2	1	7 1/2	9	27.0	763	0.29
M4 0	2	1 1/2	7 1/2	10	31.6		0.27
M5 0	3	1	12	13 1/2	49.6		0.24
M6 0	3	1 1/2	12	14 1/2	55.2		0.23

GI	ENER	K _R		9 ; <u>Note</u>	changes	1st to 2	nd mode		NP 2a-d	-	MODEL RI	CE.	M3
E	P Ref.	δ	4	$\frac{m\delta_S}{\rho b^2}$	$\frac{V}{f_b d_4}$	-	easured redicted	$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_{B}y_{max}}{10^{-3}d_{4}}$	$\left(\frac{\rho b^2}{m}\right)\frac{10^3}{4\bar{a}s}$	$\left(\frac{d^4}{b}\right)^{1/2}$		edicted
						rule	proposed ²					rule ³	propose
XI	P 2a	0.01	0	0.283	15.94	1.72	1.93	.0636	18.4	32.8	.17	.56	.68
	P 2a	0.02	0	0.565	15.92	1.72		.0494	14.3	16.4		.87	1.05
x	P2a	0.03	0	0.848	15.17	1.64		.0424	12.3	10.9		1.13	1.37
x	P 2a	0.04	0	1.130	15.19	1.64		.0362	10.5	8.2		1.28	1.55
XI	P 2a	0.05	0	1,413	15.23	1.65	1 400 1	.0284	8.2	6.6		1.25	1.51
XI	P 2a	0.06	0	1.696	15.19	1.64	1.84	.0226	6.6	5.5		1.19	1.44
XI	P 26	0.01	0	0.283	15.28	1.65	1.85	.0654	19.0	32.8		.58	.43
XI	P 26	0.02	0	0.565	14.52	1.57		.0526	15.3	16.4		.93	.69
	P 26	0.03	0	0.848	14.52	1.57	Note Also:-	.0448	13.0	10.9		1.19	.88
	P 2b	0.04	0	1.130	14.54	1.57	Stable in	.0378	11.0	8.2		1.34	.99
X	P 2b	0.05	0	1.413	13.77	1.49	Turbulent	.0308	8.9	6.6		1.35	1.00
XI	P 2b	0.06	0	1.696	13.72	1.48	Flow	.0258	7.5	5.5		1.36	1.01
	P3a	0.01	0	0.283	15.31	1.65	1.85	.0650	18.8	32.8		.57	.51
X	P 3a	0.02	0	0.565	15.29	1.65		.0572	14.8	16.4		.91	.81
X	P 3a P 3a P 3a P 3a	0.03	0	0.848	14.52	1.57		.0422	12.2	10.9		1.12	.99
X	P 3a	0.04	0	1.130	14.52	1.57		,0370	10.7	8.2		1.31	1.16
5 XI	P 3a	0.05	0	1.413	14.52	1.57		.0306	8.9	6.6		1.35	1.20
X	P 3a	0.06	0	1.696	14.12	1.53	1.72	.0252	7.3	5.5	_	1.33	1.18
X	P 3b	0.01	0	0.283	15.26	1.65	1.85	,0764	22.2	32.8		.68	.35
X	P 3b	0.02	0	0.565	14.47	1.56		.0604	17.5	16.4		1.07	.55
XIXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	P 3b	0.03	0	0.848	14.44	1.56		.0532	15.4	10.9		1.42	.73
	P 3b	0.04	0	1.130	14.44	1.56		.0452	13.1	8.2		1.60	.82
X	P 3b	0.05	0	1.413	14.44	1.56		.0366	10.6	6.6		1.61	,83
X	P 3b	0.06	0	1.696	14.48	1.56	1.75	.0300	8.7	5.5		1.58	.81
a X	P 2d	0.01	0	0.283	22.36	2.42	2.71	.1172	34.0	32.8		1.04	.43
5 X	P 2d	0.02	0	0.565	21.45	2.32		.0960	27,8	16.4		1.70	.71
9 1	P 2d	0.03	0	0.848	20.72	2.24		.0856	24.8	10.9		2.28	.95
	P 2d	0.04	0	1.130	19.95	2.16		.0750	21.7	8.2		2.65	1.10
-	P 2d	0.05	0	1.413	19.99	2.16		.0712	20.6	6.6		3.13	1.30
-	P 2d	0.06	0	1.696	19.23	2.08	2.33	.0618	-	5.5		3.26	1.36
x	100	0.01											
X		0.02											
X		0.03											
X	201	0.04				J							1
X		0.05											
X	P	0.06				1 c.f.						-	3.0 fact

² c.f. 8.25

3 without 3.0 factor 4 with c factor

Test Results v. Current Rules - Vertical Bending - Table M3Y

EP r	ref. =]	Edge / P	arape	t refere	nce
Edge	K/d4	Parapet	h/d.	suffix	φ
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
		P4	1.5	d	1

Me	del	Units	ŀd4	b*/d _i	b/d4	(b/d_s) ^{1.5}	m/pb [‡]	Ka
ΜΙ	0	1	1	3	4 1/2	9.5		0.41
342	0	1	$1^{(1)} 2$	3	5 1/2	12.9		0.37
M3	•	2	1	7 1/2	9	27.0	763	0.29
M4	0	2	$1^{-1}/_2$	7 1/2	10	31.6	-	0.27
M5	0	3	1	12	13 1/2	49.6		0.24
M6	0	3	13/2	12	14 1/2	55.2		0.23

- NOTE :

	DEVER	AL NO		19 ; <u>Note</u>	changes	1st to 2	nd mode		PARAPET YP 2au	∆ od, 3a tod	MODEL R	<u>EF.</u>	M3
	EP Ref.	δ	2	$\frac{m\delta_S}{\rho b^2}$	$\frac{V}{f_{s}d_{4}}$	p	easured redicted	$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_{R}y_{max}}{10^{3}d_{a}}$	$\left(\frac{\rho b^2}{m}\right) \frac{10^3}{4\delta s}$	$\left(\frac{d^4}{b}\right)^{1.3}$	-	easured edicted
4						rule	proposed ²		1	2.5	522	rule ³	propose
	YP 2a	0.01	0	0.283	14.13	1.53	1.71	.0366	10.6	32.8		.32	.28
	YP 2a	0.02	0	0.565	13.73	1.48		.0228	6.6	16.4		.40	.36
	YP 2a	0.03	0	0.848	12.92	1.40		.0176	5.1	10.9		.47	.42
	YP 2a	0.04	0	1,130	12.53	1.35		.0094	2.7	8.2		.33	.29
There are a second	YP 2a YP	0.05	0	1.413	12.15	1.31	1.47	.0038	1,1	6.6		.17	.15
	YP 2b	0.01	0	0.283	14.72	1.59	1.78	.0744	21.6	32.8		.66	.40
	YP 2b	0.02	0	0.565	14.26	1.54		.0520	15.1	16.4		.92	.56
	YP 2b	0.03	0	0.848	13.89	1.50		.0356	10.3	10.9		.95	.58
1	YP 2b	0.04	0	1.130	13.50	1.46		.0184	5.3	8.2		.65	.39
	YP 2b	0.05	0	1.413	13.23	1.43	1.60	.0034	1.0	6.6		.15	.09
	YP	0.06	0	200 cash									0.832
	YP 3a	0.01	0	0.283	13.66	1.55	1.74	.0932	12.1	32.8	_	.37	.26
1	YP 3a	0.02	0	0.565	13.20	1.55	1220623	.0654	6.6	16.4		.40	.28
	YP 3a	0.03	0	0.848	12.42	1.46		.0492	3.2	10.9		.30	.21
ł	YP 3a	0.04	0	1.130	12.46	1.46	1.64	.0040	0.6	8.2		.08	.06
	YP	0.05	0	120022			Press of a			1.000011		151.55	
	YP	0.06	0					I					
	YP 3b	0.01	0	0.283	14.31	1.55	1.74	.0932	27.0	32.8		.82	.36
	YP 3b	0.02	0	0.565	14.31	1.55		.0654	19.0	16.4		1.16	.52
A DOLLAR	YP 3b	0.03	0	0.848	13.54	1.46		.0492	14.3	10.9		1.31	.58
1	YP 3b	0.04	0	1.130	13.54	1.46	1.64	.0040	1.2	8.2		.14	.06
1	YP	0.05	0										
	YP	0.06	0	011									
	YP 2d	0.01	0	0.283	22.54	2.44	2.74	.1160	33.6	32.8		1.03	.38
A DOLLAR A	YP 2d	0.02	0	0.565	21.42	2.22	100000	.0990	28.7	16,4		1.75	.65
1	YP 2d	0.03	0	0.848	20.72	2.24		.0868	25.2	10.9		2.31	.86
1	YP 2d	0.04	0	1.130	19.90	2.15		.0742	21.5	8.2		2.62	.97
I	YP 2d	0.05	0	1.413	19.88	2.15		.0696	20.2	6.6		3.06	1.13
· .	YP 2d	0.06	0	1.696	19.13	2.07	2.32	.0602	17.5	5.5		3.17	1.17
I	YP 3d	0.01	0	0.283	19.80	2.14	2.39	.1500	43.5	32.8		1.33	.34
	YP 3d	0.02	0	0.565	18.98	2.05		.1304	37.8	16.4		2.31	.59
	YP 3d	0.03	0	0.848	18.31	1.98		.1170	33.9	10.9		3.11	.80
	YP 3d	0.04	0	1.130	18.30	1.98		.1090	31.6	8.2		3.85	.99
	YP 3d	0.05	0	1.413	17.46	1.89		.1026	29.8	6.6		4.51	1.16
	YP 3d	0.06	0	1.696	17.47	1.89	2.12	.0958	27.8	5.5		5.05	1.29

² c.f. 8.25

⁵ Note : for M3:YP4d see table M5Y

4 with c factor

Test Results v. Current Rules - Divergent TI - Table M5X & Y

Edge	K/d4	Parapet	h/d ₄	suffix	φ.
х	0.05	P1	0.3	a	1/4
	-	P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
36	(*****) 	P4	1.5	d	1

Mo	del	Units	l/d _a	b*/d ₄	b'd ₄	(b/d)2	m/pb ²
М1	0	1	1	3	4 1/2	20	
М2	0	1	1 1/2	3	5 1/2	30	
МЗ	0	2	1	7 1/2	9	81	
M4	0	2	1 1/2	7 1/2	10	100	
М5	•	3	1	12	13 1/2	182	
Мб	0	3	$1^{-1}/_{2}$	12	14 1/2	210	

GENERAL NOTES	PARAPETS	MODEL REF.	3.5.5
Initially $C_{i} \rightarrow 1.0$	X & Y : 2b, 3a, 3b	1	M5

			$Im\delta_{3}$	V _i			meas			
EP Ref.	$\delta_{\rm S}$	4	ρb"	frd		्र	predi (5.0)	cted (0.9)	<u>3.3</u>	
(P	0.01						(3.0)	(0.9)		
YP	0.02									
YP	0.03									
YP	0.04									
YP	0.05			- 1						
YP	0.06		· · · · · ·	_						
XP 2b	0.02	0	.0174	3.28	5 (or 0.9)	3.3	.66	3.6	1.00	
YP 2b	0.02	0	.0170	3.54	5 (or 0.9)	3.3	.71	3.9	1.08	
XP 2b	0.04	0	.0347	3.32	5 (or 0.9)	3.3	.66	3.6	1.00	1.1
YP 2b	0.04	0	.0347	3.61	5 (or 0.9)	3.3	.72	4.0	1.09	
XP 2b	0.06	0	.0521	3.42	5 (or 0.9)	3.3	.68	3.8	1.03	
YP 2b	0.06	0	.0521	3.68	5 (or 0.9)	3.3	.74	4.1	1.12	1
XP 3a	0.02	0	.0174	3.82	5 (or 0.9)	3.3	.76	4.2	1.15	
YP 3a	0.02	0	.0170	4.30	5 (or 0.9)	3.3	.86	4.8	1.30	
XP 3a	0.04	0	.0347	3.92	5 (or 0.9)	3.3	.78	4.4	1.18	
YP 3a	0.04	0	.0347	4.38	5 (or 0.9)	3.3	.88	4.9	1.33	
XP 3a	0.06	0	.0521	3.96	5 (or 0.9)	3.3	.79	4.4	1.20	
YP 3a	0.06	0	.0521	4.43	5 (or 0.9)	3.3	.89	4.9	1.35	
XP 3b	0.02	0	.0174	2.86	5 (or 0.9)	3.3	.57	3.2	.86	
YP 3b	0.02	0	.0170	3.18	5 (or 0.9)	3.3	.64	3.5	.97	
XP 3b	0.04	0	.0347	2.95	5 (or 0.9)	3.3	.59	3.3	.89	
YP 3b	0.04	0	.0347	3.27	5 (or 0.9)	3.3	.65	3.6	.98	-
XP 3b	0.06	0	.0521	3.03	5 (or 0.9)	3.3	.61	3.4	.92	
YP 3b	0.06	0	.0521	3.43	5 (or 0.9)	3.3	.69	3.8	1.05	
YP	0.01									
YP	0.02									
YP	0.03					1 1				
YP	0.04									
YP	0.05									
YP	0.06									
YP	0.01	1					1.			
YP	0.02									
YP	0.03									
YP	0.04									
YP	0.05									
YP	0.06									

Test Results v. Current Rules - Divergent TI - Table M5X & Y

EP	ref. =	Edge / P	arape	t referen	ice
Edge	K/d _e	Parapet	h/d4	suffix	¢
Х	0.05	PI	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Model	Units	Vd.	b* d4	b/d ₄	$(b/d)^2$	m/pb2
M1 0	1	1	3	4 1/2	20	
M2 0	1	1 1/2	3	5 1/2	30	
M3 🔾	2	1	7 1/2	9	81	
M4 o	2	1 1/2	7 1/2	10	100	
M5 .	3	1	12	13 1/2	182	
M6 0	3	$1^{-1}/_{2}$	12	14 %	210	

GENERA	L NOT	ES	Initia	ally $C_{\mu} \rightarrow$	1.0		<u>PARAPETS</u> N & Y : 2b, 3a	, 3b	MODEL R	EF.	M5
EP Ref.	δs	2	$\frac{Im\delta_{S}}{\rho b^{4}}$	Vs frd		er		-	edicted (0.9)	3.3	
YP	0.01		_			-		(and)	(our)	-	
YP	0.02									1	
YP	0.03						1 1				
YP	0.04						1 1				
YP	0.05						1 1				
YP	0.06			-			1.				
XP 2b	0.02	0	.0174	3.28	5 (or 0.9)	3.3		.66	3.6	1.00	
YP 2b	0.02	0	.0170	3.54	5 (or 0.9)	3.3		.71	3.9	1.68	3
XP 2b	0.04	0	.0347	3.32	5 (or 0.9)	3.3	1 1	.66	3.6	1.00	
YP 2b	0.04	0	.0347	3.61	5 (or 0.9)	3.3		.72	4.0	1.09	
XP 2b	0.06	0	.0521	3.42	5 (or 0.9)	3.3		.68	3.8	1.03	
YP 2b	0.06	0	.0521	3.68	5 (or 0.9)	3.3		.74	4.1	1.12	-
XP 3a	0.02	0	.0174	3.82	5 (or 0.9)	3.3		.76	4.2	1.15	
YP 3a	0.02	0	.0170	4.30	5 (or 0.9)	3.3		,86	4.8	1.30	-
XP 3a	0.04	0	.0347	3.92	5 (or 0.9)	3.3	1 1	.78	4.4	1.18	
YP 3a	0.04	0	.0347	4.38	5 (or 0.9)	3.3		.88	4.9	1.33	
XP 3a	0.06	0	.0521	3.96	5 (or 0.9)	3.3		.79	4.4	1.20	
YP 3a	0.06	0	.0521	4.43	5 (or 0.9)	3.3		.89	4.9	1.35	
XP 3b	0.02	0	.0174	2.86	5 (or 0.9)	3.3		.57	3.2	.86	
YP 3b	0.02	0	,0170	3.18	5 (or 0.9)	3.3		.64	3.5	.97	-
XP 3b	0.04	0	.0347	2.95	5 (or 0.9)	3.3		.59	3,3	.89	
YP 3b	0.04	0	.0347	3.27	5 (or 0.9)	3.3		.65	3.6	.98	
XP 3b	0.06	0	.0521	3.03	5 (or 0.9)	3.3		.61	3.4	.92	
YP 3b	0.06	0	.0521	3.43	5 (or 0,9)	3.3		.69	3.8	1.05	-
YP	0.01	1 1				1				1	
YP	0.02						1 1			1	
YP YP	0.04										
YP	0.04									1	
YP	0.06									1	
YP	0.00	-				-				-	
YP	0.02					1					
YP	0.03					1				1	
YP	0.04										
YP	0.05					1				1	
YP	0.06									1	

Test Results v. Current Rules - Torsion - Table M5X & Y

_		nagera	an ape	t refere	nee
Edge	K/d ₄	Parapet	h/d ₄	suffix	φ
х	0.05	PI	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	с	3/4
		P4	1.5	d	1

Mo	del	Units	₿'d _e	b^{α/d_4}	b/d4	$(b'd_{i})^{13}$	$m^\prime\rho b^2$	Ka
М1	0	1	1	3	4 1/2	193		0.41
MZ	ò	1	$1^{-1}/_{2}$	3	5 1/2	390		0.37
М3	0	2	1	7 1/2	9	2187		0.29
314	0	2	$1 \stackrel{1}{\rightarrow}_2$	7 1/2	10	3162		0.27
M5		3	1	12	13 1/2	9040	7865	0.24
M6	0	3	1 %	12	14 %	11609		0.23

- NOTE :

GENER.			$\theta = \frac{2\gamma}{2}$	b ; K	R as for	vertical hen	ding	PARAPET X & Y :	52.54 (1.155) (1.155) (1.155) (1.155)		M5
EP Ref.	$\delta_{\rm S}$	2	$\frac{Im\delta_5}{\rho b^4}$	$\frac{V}{f_{1}d_{4}}$	pro	asured edicted	(θ°)	$\frac{2K_Ry_{max}}{10^3b}$	$\left(\frac{\rho b^4}{lm}\right)\frac{10^4}{4\delta s}\left(\frac{d_4}{b}\right)^{13}$	pro	asured rdicted
	50.000				rule1	proposed ¹	-8			rule ³	propose
YP 25 YP YP YP YP	0.01 0.02 0.03 0.04 0.05	0	Sta	ble			8	table			
YP	0.06		_								-
XP 2b	.0176	0	.0153	14.12	1.18	1.41	.354	1.48	1.81	.82	.61
XP 2b	0.02	0	.0174	14.11	1.18		,337	1.41	1.59	.89	.66
XP 2b	0.03	0	.0261	13.38	1.12		.251	1.05	1.06	.99	.73
XP 2b	0.04	0	.0347	13.38	1.12		.184	0.77	0.80	.96	.71
XP 2b	0.05	0	.0434	13.40 12.76	1.12	1.28	.119	0.50	0.64 0.53	.78	.58
XP 2b XP 3a	.0176	0	.0521	13.86	1.16	1.28	.289	1.21	1.81	.54	.60
	0.02	0	.0153	13.86	1.16	1.39	.268	1.12	1.59	.71	.63
XP 3a XP 3a	0.02	0	.0261	13.77	1.16		.208	0.87	1.06	.82	.73
XP 3a XP 3a	0.03	0	.0261	13.78	1.15		.153	0.64	0.80	.80	.71
XP 3a	0.04	0	.0434	13.78	1.15		.099	0.41	0.64	.65	.58
XP 3a	0.05	0	.0434	13.79	1.15	1.38	.049	0.21	0.53	.39	.35
XP 3b	.0176	0	.0153	13.83	1.16	1.38	.469	1.96	1.81	1.08	.55
XP 3b	0.02	0	.0174	13.77	1.15	1.00	.441	1.85	1.59	1.16	.59
XP 3b	0.03	0	.0261	13.79	1.15		340	1.42	1.06	1.34	.69
XP 3b	0.04	0	,0347	13.47	1.12		.280	1.17	0.80	1.47	.75
XP 3h	0.05	0	.0434	13.18	1.10		.217	0.91	0.64	1.42	.73
XP 3b	0.06	0	.0521	13.18	1.10	1.32	.151	0.63	0.53	1.19	.61
YP 3a	.0176		.0153	7.44	0.62	0.74	.142	0.59	1.81	.33	.23
YP 3a	0.02		.0174	7.47	0.62	0.972232	.127	0.53	1.59	.33	.23
YP 3a	0.03		.0261	7.49	0.62	l	.088	0.37	1.06	.35	.25
YP 3a	0.04		.0347	7.44	0.62	0.74	.063	0.26	0.80	.33	.23
YP	0.05				12.000		1000015		2005		1000
YP	0.06		All S	table			All	Stable			
YP 3b	.0176		.0153	13.56	1.13	1.36	.037	0.15	1.81	.09	.05
YP	0.02										
YP	0.03										
YP	0.04										
YP	0.05										
YP	0.06										

² c.f. 10.0

⁴ without 3.0 factor
⁴ with c factor

Test Results v. Current Rules - Vertical Bending - Table M5X

EP	ref. =	Edge / P	arape	t refere	nce
Edge	K/d4	Parapet	h/d_{ϵ}	suffix	φ
Х	0.05	P1	0.3	a	1/4
		P2	0.7	Ъ	1/2
Y	0.20	P3	1.1	c	3/4
.11		P4	1.5	d	1

Mo	del	Units	I'd,	h*/d,	h'd,	(b/d_i) ^{1.5}	m'pb ²	Ka
MI	0	1	I	3	4 1/4	9.5		1
M2	0	1	$ {}^{\pm} \rangle_2$	3	5 1/2	12.9		
M3	0	2	1	$7^{-1}/_{\pm}$	9	27.0		
M4	0	2	1 1/2	7 1/2	10	31.6		
M5	•	3	1	12	13 ¹ / ₁	49.6	802	0.24
M6	0	3	1 1/2	12	14 1/2	55.2		

GENER/	AL NOT	ES		K _B = 0.24				PARAPETS NP 2b, 3		MODEL R	<u>EF.</u>	M5
EP Ref.	$\delta_{\rm s}$	2	$\frac{m\delta_S}{\rho b^2}$	$\frac{V}{f_{s}d_{4}}$	pre	dicted	$\left(\frac{y_{max}}{d_4}\right)_{s}$	$\frac{K_{\rm E}y_{max}}{10^{-3}d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10}{45}$	$\frac{d^4}{b}\left(\frac{d^4}{b}\right)^{1.5}$	pro	edicted
	0.01		-		rule	proposed ²		-		1993	rule ³	proposed
XP XP	0.01											
XP	0.02											
XP	0.03											
XP	0.05								1			
XP	0.06											
XP 2b	.0151	0	0.244	10.51	0.88	1.05	.0210	5.0	20.6	_	.24	.18
XP 2b	0.02	0	0.323	10.57	0.88		.0186	4.5	15.6		.29	.21
XP 2b	0.03	0	0.485	10.59	0.88		.0150	3.6	10,4		.35	.26
XP 2b	0.04	0	0.647	10.59	0.88		.0126	3.0	7.8		.39	.29
XP 2b	0.05	0	0.808	10.61	0.88		.0106	2.5	6.2		.41	.30
XP 2b	0.06	0	0.970	10.60	0.88	1.06	.0090	2.2	5.2		.42	.31
XP 3a	.0151	0	0.244	11.00	0.92	1.10	.0158	3.8	20.6		.18	.16
XP 3a	0.02	0	0.323	10.98	0.92		.0140	3.4	15.6		.21	.19
XP 3a	0.03	0	0.485	11.02	0.92		.0110	2.6	10.4		.25	.22
XP 3n	0.04	0	0.647	10.55	0.92		.0092	2.2	7.8		.28	.25
XP 3a	0.05	0	0.808	11.01	0.92		.0080	1.9	6.2		.31	.28
XP 3a	0.06	0	0.970	11.05	0.92	1.10	.0068	1.6	5.2		.31	.28
XP 3b	.0151	0	0.244	10.51	0.88	1.05	.0204	4.9	20.6		.24	.13
XP 3b	0.02	0	0.323	10.51	0.88		.0170	4.1	15.6		.26	.13
XP 3b	0.03	0	0.485	10.51	0.88		.0112	2.7	10.4		.26	.13
XP 3b	0.04	0	0.647	10.49	0.88	1.05	.0056	1.3	7.8		.17	.09
XP	0.05	0										
XP	0.06	0			1.1.1.1.1.1							
XP	0.01			1								
XP	0.02			1.1								1
XP	0.03											
XP	0.04			1 1								
XP	0.05				1.10							
XP	0.06										-	
XP	0.01											
XP	0.02											
XP	0.03											
XP	0.04											
XP	0.05											
XP	0.06			¹ c.f. 1							1	3.0 facto

⁴ with c factor

Test Results v. Current Rules - Vertical Bending - Table M5Y

EP r	ef. = I	Edge / P	arape	t refere	ence
Edge	K/d ₄	Parapet	h/d4	suffix	ф
х	0.05	P1	0.3	a	1/4
		P2	0.7	b	1/2
Y	0.20	P3	1.1	c	3/4
		P4	1.5	d	1

Model	Units	Vd.	h^{\ast}/d_{4}	b/d,	$(b/d_4)^{1.9}$	m/pb ²	KR
M1 0	1	1	3	4 1/2	9.5		
M2 0	1	$1 \ ^1/_2$	3	5 1/2	12.9		
M3 O	2	1	7 1/2	9	27.0		
M4 o	2	$1 \frac{1}{2}$	7 1/2	10	31.6		
M5 •	3	1	12	13 1/2	49.6	802	0.24
M6 O	3	1 1/2	12	14 1/2	55.2		

- NOTE :

GENERAL	NOTES		K	= 0.24				PARAPET NP 26.3		<u>:EF.</u>	M5
EP Ref.	δs	2	$\frac{m\delta_k}{pb^2}$	$\frac{V}{f_{b}d_{4}}$		rasured edicted	$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_Ry_{ams}}{10^3d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10^3}{4\delta s} \left(\frac{d^4}{b}\right)^{1.5}$		rasured redicted
					rule ¹	proposed ²				rule ³	proposed
ҮР ҮР ҮР ҮР ҮР	0.01 0.02 0.03 0.04 0.05 0.06										
YP 2b YP 2b YP YP YP YP	.0151 .0200 .0300 .0400 .0500 .0600	0	0.244 0.323	10.03 10.03	0.84 0.84	1.00 1.00	.0126 .0084	3.0 2.0	20.6 15.6	.15 .13	.09 .08
YP 3a YP 3a YP 3a YP YP YP	.0151 .0200 .0300 .0400 .0500 .0600	0000	0.244 0.323 0.485	9.22 9.60 9.57	0.77 0.80 0.80	0.92	.0136 .0126 .0060	3.3 3.0 1.4	20.6 15.6 10.4	.16 .19 .14	.11 .14 .10
YP 35 YP YP YP YP YP	.0100 .0200 .0300 .0400 .0500 .0500		Str	hle							

¹ c.f. 12.0 ² c.f. 10.0

Additional M3Y Test - See Table M3Y

M3YP 4d	0,01	0	.283				N.A		32.8	N.A	N.A
M3YP 4d	0.02	0	.565				N.A		16.4	N.A	N_A
YP	0.03	0	4	- See	notes or	table M3	Ý —		-	N.A	N_A.
M3YP 4d	0.04	0	1.130	18.08	1.95	2.19	.1962	56.9	8.2	6.94	1.36
M3YP 4d	0.05			18.12			.1756	50.9	6.6	7.71	1.51
M3YP 4d	0.06	1000	 Provide the 	17.49	1000000	2.12	.1590	46.1	5.5	8.38	1.64

³ without 3.0 factor

4 with c factor

BR ref. = Barrier / Railing reference											
Barrier	b/d4	Railing	h/d ₄	suffix	φ.						
BO	No Barrier										
B1	0.15	R1	0.2	a	0.15						
B2	0.34	R2	0.5	b	0.21						
B3	0.37	R3	0.7	c	0.50						
B4	0.50			d	1.00						

Model	Units	ld,	b^{ϕ}/d_{ϕ}	h/d ₄	(b/d) ^{1.5}	KR
8B1		1.	4	6	14.7	0.355
8B _{1.5}		1 1/2	4	7	18.5	0.329
3A		1	4	б	14.7	0.355

Box Girder Test Results v. Current Rules - Vertical Bending - Table 3A

GENERAL	NOTES						<u>e Beam</u> = 0.05	MODEL REE. Box Girder		3A
BR ref.	δs	4	$\frac{m\delta_S}{\rho b^2}$	V fd,		Ymax d4	$\frac{K_R y_{max}}{10^{-3} d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10}{4\delta_s} \left(\frac{d^4}{b}\right)^{1.5}$	1.1.7.1	asured dicted
						1.1.1			rule	proposed
BO R2a	0.014		0.47	9.5		.270	9.59	36.00	.27	.72*
BO R2a	0.058		1.96	9.3		.010	3.55	8.68	.41	1.09*
BO R2a	0.115		3.89	9.6		.005	1.78	4.37	.41	1.09*
B4c R2a	0.014		0.47			.050	17.80	36.00	.49	.44
B4c R2a	0.058		1.96	11.0		.035	12.40	8.68	1.43	1.27
B4c R2a	0.115		3.89	10.4		.014	4.97	4.37	1.14	1.01
B4c R2a	0.173		5.83	10.0		.008	2.84	2.92	.97	.86
B4c R2a	0.014		0.47	1000		.005	1.78	36.00	.05	.04
B4c R2a	0.014		0.47		2nd Peak	.050	17.80	36.00	.49	.44
B4c R2a	0.058		1.96	10.0		.016	5.68	8.68	.65	.58
B4c R2a	0.115		3.89	9.6		.007	2.49	4.37	.57	.51
B4c R2a	0.173		5.83	9.5		.005	1.78	2.92	.61	.54
B2b R2a	0.014		0.47	10.4		.021	7.46	36.00	.21	.41
B2b R2a	0.058		1.96	10.3		.012	4.26	8.68	.49	.95
B2b R2a	0.115		3.89	10.1		.007	2.49	4.37	.57	1.01
B2b R2a	0.173		5.83	10.0		.005	1.78	2.92	.61	1.19
B2b R2c	0.014		0.47	10.1		.031	11.00	36.00	.31	.27
B2b R2c	0.058		1.96	10.1		.026	9.23	8.68	1.06	.93
B2b R2c	0.115		3.89	10.3		.013	4.62	4.37	1.06	.93
B2b R2c	0.173		5.83	10.3		.008	2.84	2.92		
B1b R1c	0.014		0.47	9.5		.024	8.52	36.00	.24	.44
B1b R1c	0.058		1.96	9.2		.014	4.97	8.68	.57	1.05
B1b R1c	0.115		3.89	8.9		.007	2.49	4.37	.57	1.05
B1b R1c	0.173		5.83	8.9		,005	1.78	2.92	.61	1.12
B3d R2c	0.014		0.47	10.5		.050	17.80	36.00	.49	.24
B3d R2c	0.058		1.96	10.5		.025	8.88	8.68	1.02	.51
B3d R2c	0.115		3.89	10.6		.010	3.55	4.37	.81	.40
B3d R2c	0.173		5.83	10.1		.007	2.49	2.92	.85	.42
B1d R1c	0.014		0.47	9.4		.025	8.88	36.00	.25	.28
B1d R1c	0.058	1	1.96	9.2		.014	4.97	8.68	.57	.63
Rle	0.115		3.89	9.0		.007	2.49	4.37	.57	.63
Ric	0.173		5.83	9.1		.005	1.78	2.92	.61	0.68
P P P P										

Box Girder Test Results v. Current Rules - Vertical Bending - Table 8B

BR ref. = Barrier / Railing reference											
Barrier	b/d4	Railing	h/d_4	suffix	φ.						
BO	No Barrier										
BI	0.15	R1	0.2	а	0.15						
B2	0.34	R2	0.5	b	0.21						
B3	0.37	R3	0.7	c	0.50						
B4	0.50			d	1.00						

Model	Units	l'd4	$b^{\mathfrak{p}/d_{2}}$	b/d4	(b/d)1.5	Ka
8B _t		1	4	6	14.7	0.355
8B _{1.3}		1 1/2	4	7	18.5	0.329
3A		1	4	6	14.7	0.355

GENERAL	NOTES	8	$K_{R} = 0.$	329			e Beam = 0.05	MODEL REF. Box Girder	8	B _{1.5}
BR ref.	δs	2	$\frac{m\delta_S}{\rho \ b^2}$	V fd4		$\frac{y_{max}}{d_4}$	$\frac{K_R y_{max}}{10^3d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10}{4\delta_s} \left(\frac{d^4}{b}\right)^{12}$	10000	edicted
						- 12			rule	proposed
BO R2a	0.014		0.346	7.3	10.2.1.1.1	.007	2.30	39.00	.06	.16*
BO R2a	0.014		0.346	13.3	2nd Peak	.018	5.92	39.00	.15	.40*
BO R2a	0.058		1.432	11.4	2nd Peak	.007	2.30	9.43	.24	.64*
BO R2a	0.014		0.346	9.6		.020	6.58	39.00	.17	.45*
BO R2a	0.058		1.432	9.7		.007	2.30	9.43	.24	.64*
BO R2a	0.014	F	0.346	9.6		.031	10.19	39.00	.26	.69*
BO R2a	0.058		1.432	9.2		,009	2.96	9.43	.31	.83*
BO R3c	0.014		0.346	10.2		.034	11.18	39.00	.29	+18
BO R3c	0.058		1.432	11.7		.015	4.93	9.43	.52	.19
BO R2d	0.014		0.346	5.5		.006	1.97	39.00	.05	.03
BO R2d	0.014		0.346	14.7	2nd Peak	.044	14.47	39.00	.37	.22
BO R2d	0.058		1.432	13.1		.026	8.55	9.43	.90	.55
BO R2d	0.115		2.840	12.5		.012	5.26	4.75	1.11	.67
BO R2d	0.173		4.270	12.0		.013	4.27	3.16	1.35	.82
P										
P									10.00	
P										
P							1000	1111/		
P		1					19 19			
P										
P	1.2									
Р										
Р									_	
Р										
P						1				
P										
P	1									
P									1	
P						-				
P						1	(C)			
P									11	
P									1	
Р									1	
Р										
р										

* Note c = 0.375 - very low - see table 3

Box Girder Test Results v. Current Rules - Vertical Bending - Table 8B

BR r	ef. = B	arrier /	Railir	ng refer	ence	
Barrier	b/d4	Railing	h/d_{ϵ}	suffix	ф	
BO	No Barrier					
B1	0.15	RI	0.2	a	0,15	
B2	0.34	R2	0.5	b	0.21	
B3	0.37	R3	0.7	c	0.50	
B4	0.5			d	1.00	

Model	Units	1/de	b*/d4	b/d _e	(b/d)1.5	Ka
8B ₁		1	4	6	14.7	0.355
8B _{1.3}		1 1/2	4	7	18.5	0.329
3A		1	4	6	14.7	0.355

GENERAL	NOTES						- 0.05	MODEL REF. Box Girder		8 B ₁
BR ref.	δs	2	$\frac{m\delta_8}{\rho \ b^2}$	V fd4		$\left(\frac{y_{max}}{d_4}\right)$	$\frac{K_R y_{max}}{10^3d_4}$	$\left(\frac{\rho b^2}{m}\right) \frac{10}{4\delta_s} \left(\frac{d^4}{b}\right)^{13}$		easured edicted
	1.3.		100			63			rule	proposed
	0.014		0.472	7.3		.037	13.10	36.00	.36	
Bare	0.014		1.960	13.3		.015	5.33	8.68	.61	Bare Bo
Box	0.058		3.890	11.4		.008	2.84	4.37	.65	rule
	0.014		5,830	9.6		.005	1.78	2.92	.61	N.A
BO R0	0.014		0.472	7.9		.018	6.39	36.00	.18	- 11
BO R2a	0.014		0.472	6.2		.005	1.78	36.00	.05	.13*
BO R2a	0.014		0.472	12.4	2nd Peak	.030	10.66	36.00	,30	.80*
BO R2a	0.058		1.960	11.5	2nd Peak	.016	5.68	8.68	.65	1.73*
BO R2a	0.115		3,890	10.7	2nd Peak	.008	2.84	4.37	.65	1.73*
BO R2a	0.014		0.472	8.9	2nd Peak	.010	3.55	36.00	.10	.27*
BO R2a	0.014		0.472	8.4		.026	9.23	36.00	.26	.69*
BO R2a	0.058		1.960	8.9		.007	2.49	8.68	.29	.77*
BO R3c	0.014		0,472	11.6		.012	4.26	36.00	.12	.07
BO R3d	0.014		0.472	11.7		.054	19.18	36.00	.53	.17
BO R3d	0.058		1.960	10.7		,034	12.08	8.68	1.39	.44
BO R3d	0.115		3.890	9.9		.021	7,46	4.37	1.71	.54
BO R3d	0.173		5.830	9.7		.014	4.97	2.92	1.70	.54
B4c R2a	0.014		0.472	6.4		.012	4.26	36.00	.12	.13
B4c R2a	0.014		0.472	11.9	2nd Peak	.026	9.23	36.00	.26	.28
B4c R2a	0.058		1.960	10.2	2nd Peak	.013	4.62	8.68	.53	.57
B4c R2a	0.115		3.890	9.9	2nd Peak	.007	2.49	4.37	.57	.62
B4c R2a B4c R	0.173		5.830	9.9	2nd Peak	.005	1.78	2.92	.61	.67
B4c R2a	0.014		0.472	6.5	Vinen in	.008	2.84	36.00	.08	.09
B4c R2a	0.014		0.472	10.5	2nd Peak	.017	6.04	36.00	.17	.18
B4c R2a B4c R2a P P P	0.058		1.960	10.1		.008	2.84	8.68	.33	.36
P P P P P										

* Note c = 0.375 - very low - see table 3

Abstract

Highways Agency Standard BD 49: *Design rules for aerodynamic effects on bridges* was first published in 1993 and contains simplified methods for determining the susceptibility of bridge decks to aerodynamic instability. The standard was devised based on information available at that time. However, it contained a number of caveats for some deck configurations and geometry, particularly in relation to plate girder bridges. Following later studies, experience of application and additional wind tunnel data, some of the clauses of BD 49 were revised and the document was re-published in 2001. The background to this update is reported in TRL Report TRL528.

This report describes a programme of wind tunnel tests that was carried out on plate girder bridges to provide information for this revision. Over 60 different configurations were tests, using different parapets and leading edge details. The report describes the models, the wind tunnel characteristics and the conclusions from the tests. The wind tunnel data are presented in a form that can be used directly by engineers to assess the susceptibility different bridge forms to aerodynamic instability in cases were the simplified rules in BD 49 indicate the possibility of aerodynamic effects.

Related publications

TRL528	Background to the development of BD 49/01: Design rules for aerodynamic effects on bridges
	by B W Smith, T A Wyatt, D K McKenzie, M V Hubband and A F Daly. 2002 (price £25, code E)
CR256	A re-appraisal of certain aspects of the design rules for bridge aerodynamics by Flint and Neill Partnership. 1992 (price $\pounds 25$, code E)

CR36 Partial safety factors for bridge aerodynamics rules and requirements for wind tunnel testing by Flint and Neill Partnership. 1986 (price £35, code H)

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